Section H. Appendices

H.1 Detailed Budget

H.1.1 Budget Summaries

Foldout H-1 shows the entire LISA budget in the required format (Excel tab Table 2). Tables H-1 through H-3 are summary breakouts of this information by project phase.

Table H-1: Technology Development Cost Funding Profile (FY Costs in Real Year Dollars (\$M). Totals in RY and FY03 Dollars)

							SUBT	OTAL
			Adv Te	ch Dev			Adv Te	ch Dev
Cost Element	PY	FY03	FY04	FY05	FY06	FY07	RY\$	FY03\$
Start of ATD to Launch + 30 days								
(ATD, formulation and implementation)								
1.1 Management	0.6	0.3	0.4	2.0	2.9	0.1	6.3	6.0
1.9 Technology Development (excl. 1.9.4)	2.0	0.9	4.0	8.8	11.4	4.6	31.7	29.6
1.9.4 Systems Verification Subtotal before Reserves	1.2 3.8	1.2 2.4	1.4 5.8	3.7 14.5	3.3 17.6	3.3 8.0	14.1 52.1	13.2 48.7
Reserves	0.0	0.5	0.8	3.7	12.6	11.0	28.7	26.1
110001700	0.0	0.0	0.0	0.7	12.0	11.0	20.7	20.1
Total ATD, formulation, and implementation	3.8	3.0	6.6	18.2	30.2	19.0	80.8	74.8
through launch plus 30 days								
Total LISA (NASA) cost	3.8	3.0	6.6	18.2	30.2	19.0	80.8	74.8
FULL COST OF:								
NASA FTEs		0.8	1.3	2.7	4.2	2.3	11.2	8.9
OSS Required Funding for LISA	3.8	2.2	5.3	15.5	26.1	16.7	69.6	65.9
CONTRIBUTIONS OUTSIDE OF UPN 785-30								
US (NON-LISA) CONTRIBUTIONS								
Technology Flight Validation (NMP/ST-7)	3.0	14.4	20.0	11.7	3.6	2.8	55.7	53.8
Total US Cost (NASA)	3.0	14.4	20.0	11.7	3.6	2.8	55.7	53.8
		'					'	
ESA CONTRIBUTIONS								
1.1 Management	0.2	0.1	0.2	0.7	1.1	0.0	2.3	2.2
1.9 Technology Development	4.3	2.6	4.0	8.8	11.4	0.0	31.1	29.6
LISA Test Package	5.3	18.2	21.8	13.2	3.6	0.0	62.1	60.5
Technology Flight Validation (SMART-2)	7.0	9.0	15.5	17.9	13.2	0.0	62.5	60.0
Subtotal before Reserves	16.8	29.9	41.4	40.6	29.3	0.0	158.0	152.3
Reserves	0.0	9.5	13.2	13.9	11.1	0.0	47.7	45.5
Total ESA Cost	16.8	39.3	54.6	54.5	40.3	0.1	205.7	197.8
Combined Contributions	19.8	53.8	74.7	66.3	44.0	2.9	261.3	251.6
Total Mission Cost (All Sources)	23.6	56.8	81.3	84.4	74.2	21.9	342.1	
Total Wission Cost (All Sources)	23.5	8.80	81.3	84.4	74.2	21.9	342.1	326.4

Table H-2: Formulation Cost Funding Profile (FY costs in Real year Dollars (\$M). Totals in RY and Fixed Year '03 Dollars).

							SUBTO	OTAL							SUBT	OTAL
							Adv T	ech			ATD	plus			ATD	plus
			Adv Ted	h Dev			De	ν			Formu	lation			Formu	lation
Cost Element	PY	FY03	FY04	FY05	FY06	FY07	RY\$	FY03\$	PY	FY03	FY04	FY05	FY06	FY07	RY\$	FY03\$
Start of ATD to Launch + 30 days																
(ATD, formulation and implementation)																
1.1 Management	0.6	0.3	0.4	2.0	2.9	0.1	6.3	6.0	1.0	0.6	1.6	4.3 3.6	6.7	6.1	20.4	18.9
System Engineering/Integration Contractor Mission Assurance	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.7	0.4	0.7	7.5 1.6	7.7	21.2 5.0	4.6
1.4 Science (excluding 1.4.4)	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.9	0.5	1.2	2.1	2.7	3.3	10.8	10.0
1.4.4 Education & Public Outreach	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.2	0.3	0.5	0.6	0.9	2.5	2.3
1.5 Payload System	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.7	0.8	1.3	2.7	7.7	9.8	23.1	21.1
1.6 Flight System				0.0					0.0							
1.6.2 ATLO (SE&I Contractor) 1.7 Mission Operations System	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0 2.2	2.3	3.8 2.7	6.2 8.6	5.5 8.0
1.9 Technology Development (excl. 1.9.4)	2.0	0.0	4.0	8.8	11.4	4.6	31.7	29.6	2.0	0.9	4.0	8.8	11.4	4.6	31.7	29.6
1.9.4 Systems Verification	1.2	1.2	1.4	3.7	3.3	3.3	14.1	13.2	1.2	1.2	1.4	3.7	3.3	3.3	14.1	13.2
1.10 Mission Software	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	1.0	2.8	3.6	10.2	12.9	30.5	27.9
Subtotal before Reserves	3.8	2.4	5.8	14.5	17.6	8.0	52.1	48.7	6.5	6.4	15.4	32.2	56.1	57.4	174.0	160.8
Reserves	0.0	0.5	0.8	3.7	12.6	11.0	28.7	26.1	0.0	0.5	0.8	3.7	12.6	30.9	48.6	43.7
Total ATD, formulation, and implementation	3.8	3.0	6.6	18.2	30.2	19.0	80.8	74.8	6.5	7.0	16.2	35.9	68.7	88.3	222.6	204.0
through launch plus 30 days	3.0	3.0	0.0	10.2	30.2	13.0	00.0	74.0	0.5	7.0	10.2	35.3	00.7	00.0	222.0	204.0
Total LISA (NASA) cost	3.8	3.0	6.6	18.2	30.2	19.0	80.8	74.8	6.5	7.0	16.2	35.9	68.7	88.3	222.6	204.0
FULL COST OF:																
NASA FTEs		0.8	1.3	2.7	4.2	2.3	11.2	8.9		1.7	3.3	5.3	9.4	10.4	30.2	27.7
OSS Required Funding for LISA	3.8	2.2	5.3	15.5	26.1	16.7	69.6	65.9	6.5	5.3	12.9	30.5	59.3	77.9	192.4	176.3
CONTRIBUTIONS OUTSIDE OF UPN 785-30																
US (NON-LISA) CONTRIBUTIONS																
Technology Flight Validation (NMP/ST-7)	3.0	14.4	20.0	11.7	3.6	2.8	55.7	53.8	3.0	14.4	20.0	11.7	3.6	2.8	55.7	53.6
Total US Cost (NASA)	3.0	14.4	20.0	11.7	3.6	2.8	55.7	53.8	3.0	14.4	20.0	11.7	3.6	2.8	55.7	53.8
, ,																
ESA CONTRIBUTIONS																
1.1 Management	0.2	0.1	0.2	0.7	1.1	0.0	2.3	2.2	0.4	0.2	0.6	1.6	2.5	2.3	7.5	6.9
1.2 System Engineering/Integration Contractor	0.0	0.0		0.0	0.0	0.0	0.0	0.0	0.0	0.2	0.4	1.0	2.0	2.1	5.7	5.2
1.3 Mission Assurance	0.0	0.0		0.0	0.0	0.0	0.0	0.0	0.0	0.1	0.3	0.5	1.0	1.4	3.3	3.0
1.4 Science	0.0	0.0		0.0	0.0	0.0	0.0	0.0	0.9	0.5	1.2	2.1	2.7	3.3	10.8	10.0
1.4.4 Education & Public Outreach	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.1	0.1	0.2	0.3	0.4	1.0	1.0
1.5 Payload System	0.0	0.0		0.0	0.0	0.0	0.0	0.0	0.7	0.8	1.3	2.7	7.7	9.8	23.1	21.1
1.6 Flight System	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.7	0.0	1.3	2.1	1.1	3.0	23.1	21.1
			0.0	0.0	0.0	0.0	0.0		0.0		0.0	0.0		20.0	50.4	50.0
1.6.1 Spacecraft (including FEEPs)	0.0	0.0		0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	21.1	38.0	59.1	52.9
1.7 Mission Operations System	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.1	0.1	0.1	0.3	0.3	0.4	1.2	1.1
1.9 Technology Development	4.3	2.6		8.8	11.4	0.0	31.1	29.6	4.3	2.6	4.0	8.8	11.4	0.0	31.1	29.6
LISA Test Package	5.3	18.2	21.8	13.2	3.6	0.0	62.1	60.5	5.3	18.2	21.8	13.2	3.6	0.0	62.1	60.5
Technology Flight Validation (SMART-2)	7.0	9.0		17.9	13.2	0.0	62.5	60.0	7.0	9.0	15.5	17.9	13.2	0.0	62.5	60.0
Subtotal before Reserves	16.8	29.9	41.4	40.6	29.3	0.0	158.0	152.3	18.7	31.7	45.3	48.2	65.8	57.6	267.3	251.2
Reserves	0.0	9.5		13.9	11.1	0.0	47.7	45.5	0.0	9.5	13.2	13.9	22.7	18.0	77.4	62.4
Total ESA Cost	16.8	39.3	54.6	54.5	40.3	0.1	205.7	197.8	18.7	41.2	58.5	62.1	88.5	75.6	344.6	313.7
Combined Contributions	19.8	53.8	74.7	66.3	44.0	2.9	261.3	251.6	21.7	55.6	78.5	73.8	92.1	78.4	400.3	367.5
Total Mission Cost (All Sources)	23.6	56.8	81.3	84.4	74.2	21.9	342.1	326.4	28.2	62.6	94.8	109.7	160.8	166.7	622.9	571.5

Table H-3: Implementation Cost Funding Profile (FY Costs in Real Year Dollars (\$M).

Totals in RY and Fixed Year '03 Dollars)

	1											SUBT	OTAL	тот	AL
					Imp	lemental	tion					Impleme	ntation	LIFE C	YCLE
Cost Element	FY08	FY09	FY10	FY11	FY12	FY13	FY14	FY15	FY16	FY17	FY18	RY\$	FY03\$	RY\$	FY03\$
Start of ATD to Launch + 30 days															
(ATD, formulation and implementation)															
1.1 Management	8.8	10.3	13.3	12.4		0.0	0.0	0.0	0.0	0.0	0.0	44.8	36.6	65.3	5
1.2 System Engineering/Integration Contractor	8.8	10.4	13.4	12.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	45.2	36.9	66.3	5
1.3 Mission Assurance	2.5	2.9	3.5	3.3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	12.3	10.0	17.3	1
1.4 Science (excluding 1.4.4)	5.0	5.9 1.3	7.6	7.1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	25.6	20.9 4.7	36.4 8.2	3
1.4.4 Education & Public Outreach 1.5 Payload System	1.1	1.3	1.7 19.5	1.6 18.2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	5.7 65.7	53.7	88.8	7
1.6 Flight System	12.8	15.2	19.5	10.2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	65.7	53.7	00.0	
1.6.2 ATLO (SE&I Contractor)	6.0	7.1	9.1	8.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	30.7	25.1	36.9	9
1.7 Mission Operations System	2.7	3.1	4.0	3.7	0.0	0.0	0.0	0.0	0.0	0.0	0.0	13.6	11.1	22.1	1
1.8 Launch Systems	25.9	30.5	39.3	36.6	0.0	0.0	0.0	0.0	0.0	0.0	0.0	132.4	108.1	132.4	10
1.9.4 Systems Verification	3.4	3.5	3.6	3.7	0.0	0.0	0.0	0.0	0.0	0.0	0.0	14.1	11.5	28.2	2
1.10 Mission Software	15.8	18.8	23.3	21.4	0.0	0.0	0.0	0.0	0.0	0.0	0.0	79.3	64.8	109.9	9
Subtotal before Reserves	93.0	109.0	138.5	128.9	0.0	0.0	0.0	0.0	0.0	0.0	0.0	469.4	383.4	643.4	54
Reserves	27.8	36.1	45.7	51.9	0.0	0.0	0.0	0.0	0.0	0.0	0.0	161.4	131.4	210.0	17
Total ATD, formulation, and implementation	120.8	145.1	184.2	180.8	0.0	0.0	0.0	0.0	0.0	0.0	0.0	630.8	514.8	853.4	71
through launch plus 30 days	120.0	145.1	104.2	100.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	000.0	314.0	030.4	′'
Launch + 30 Days to End of Mission												Enter each c			ļ
1.1 Management	0.0	0.0	0.0	0.0	1.2	0.7	0.7	0.7	0.8	0.8	0.2	5.0	3.5	5.0	
1.2 System Engineering/Integration Contractor	0.0	0.0	0.0	0.0	2.0	1.3	0.9	0.9	0.9	0.9	1.0	7.9	5.6	7.9	
1.4 Science	0.0	0.0	0.0	0.0	13.2	13.6	12.7	13.1	10.8	11.1	11.5	86.0	60.0	86.0	6
1.4.4 Education & Public Outreach	0.0	0.0	0.0	0.0	1.6	1.7	0.9	0.9	0.9	0.8	0.8	7.5	5.3	7.5	
1.5 Payload System	0.0	0.0	0.0	0.0	2.5 8.2	9.3	0.0 9.6	0.0 9.5	0.0 8.4	0.0 8.6	0.0	2.5 62.4	1.9 43.4	2.5 62.4	
Mission Operations System Systems Verification	0.0	0.0	0.0	0.0	3.6	0.0	0.0	0.0	0.0	0.0	0.0	3.6	2.7	3.6	4
1.10 Mission Software	0.0	0.0	0.0	0.0	3.3	1.4	1.4	1.5	1.5	1.6	1.6	12.2	8.6	12.2	
Subtotal before Reserves	0.0	0.0	0.0	0.0	35.7	27.9	26.2	26.5	23.3	23.8	23.9	187.2	131.0	187.2	13
Reserves	0.0	0.0	0.0	0.0	8.4	4.2	4.1	4.1	3.9	4.0	3.9	32.6	22.9	32.6	2
Total from launch plus 30 days to end	0.0	0.0	0.0	0.0	44.0	32.1	30.3	30.7	27.1	27.8	27.8	219.8	154.0	219.8	15
Total LISA (NASA) cost	120.8	145.1	184.2	180.8	44.0	32.1	30.3	30.7	27.1	27.8	27.8	850.6	668.8	1,073.2	87
FULL COST OF: NASA FTEs	12.5	15.1	19.5	18.7	4.1	2.3	2.1	2.2	2.2	2.3	2.3	83.3	66.0	113.5	9
OSS Required Funding for LISA	108.2	130.0	164.6	162.0	39.9	29.8	28.2	28.5	24.9	25.5	25.5	767.3	602.8	959.7	77
OSS Required Furiding for LISA	100.2	130.0	164.6	162.0	39.9	29.0	20.2	20.5	24.9	25.5	25.5	101.3	602.6	959.7	
CONTRIBUTIONS OUTSIDE OF UPN 785-30															
ESA CONTRIBUTIONS															
1.1 Management	3.2	3.8	4.9	4.5	0.4	0.3	0.3	0.3	0.3	0.3	0.1	18.2	14.7	25.7	2
1.2 System Engineering/Integration Contractor	2.4	2.8	3.6	3.3	0.5	0.3	0.2	0.2	0.2	0.3	0.3	14.2	11.3	19.8	1
1.3 Mission Assurance	1.7	1.9	2.3	2.2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	8.1	6.6	11.4	
1.4 Science	5.0	5.9	7.6	7.1	13.2	13.6	12.7	13.1	10.8	11.1	11.5	111.6	80.9	122.4	g
	0.5	0.5	0.7	0.6	0.7	0.7	0.3	0.4	0.4		0.3	5.4	4.0	6.4	-
1.4.4 Education & Public Outreach															
1.5 Payload System	12.9	15.2	19.5	18.2	2.5	0.0	0.0	0.0	0.0	0.0	0.0	68.3	55.6	91.3	7
1.6 Flight System															
	82.8	109.8	43.0	32.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	267.7	222.3	326.7	27
1.6.1 Spacecraft (including FEEPs)			0.6	0.5	1.2	1.3	1.4	1.4	1.2	1.2	1.3	11.0	7.9	12.2	
Spacecraft (including FEEPs) Mission Operations System	0.4	0.5	0.0	0.0											
	0.4	140.3	82.2	68.5	18.5	16.2	14.9	15.3	12.9	13.2	13.4	504.3	403.4	771.6	65
1.7 Mission Operations System Subtotal before Reserves	108.8	140.3	82.2	68.5	18.5	16.2									
Mission Operations System Subtotal before Reserves Reserves	108.8	140.3 43.1	82.2 26.0	68.5 26.5	18.5 2.0	16.2 0.6	0.6	0.6	0.5	0.6	0.5	131.1	107.3	208.5	65 16
1.7 Mission Operations System Subtotal before Reserves	108.8	140.3	82.2	68.5	18.5	16.2									

Laser Interferometer Space Antenna
Section H: Appendices

Excel Tab Table 2

TOTAL MISSION COST FUNDING PROFILE

FY Costs in Real Year Dollars (\$M). Totals in RY and Fixed Year '03 Dollars

							SUBTO	TAL							SUBT	OTAL												SUBTOT	AL	TOTA	L
			Adv Tech	n Dev			Adv Tech	Dev		AT	D plus For	mulation			ATD Formu						Imple	ementatio	7					Implement	ation	LIFE CYC	CLE
Cost Element	PY	FY03	FY04	FY05	FY06	FY07	RY\$	FY03\$	PY	FY03	FY04	FY05	FY06	FY07			FY08	FY09	FY10	FY11	FY12	FY13	FY14	FY15	FY16	FY17	FY18	RY\$	FY03\$	RY\$	FY03\$
Start of ATD to Launch + 30 days																															
(ATD, formulation and implementation)	0.0	0.0	0.4	0.0	0.0	0.1	0.0	0.0	1.0	0.0	4.0	4.0		nter each c		nt	0.0	10.0	10.0	10.4	0.0	0.0	0.0	0.0	0.0	0.0	0.0	44.0	00.0	05.0	
1.1 Management 1.2 System Engineering/Integration Contractor	0.6	0.3	0.4	0.0	0.0	0.1	6.3 0.0	0.0	0.1	0.6	1.6	3.6	6.7 7.5	7.7	20.4	18.9	8.8	10.3	13.3	12.4 12.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	44.8 45.2	36.6 36.9	65.3 66.3	55 56
1.3 Mission Assurance	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.1	0.4	0.7	1.6	2.2	5.0	4.6	2.5	2.9	3.5	3.3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	12.3	10.0	17.3	14
1.4 Science (excluding 1.4.4)	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.9	0.5	1.2	2.1	2.7	3.3	10.8	10.0	5.0	5.9	7.6	7.1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	25.6	20.9	36.4	30
1.4.4 Education & Public Outreach	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.2	0.3	0.5 2.7	7.7	9.8	2.5	2.3	1.1	1.3	1.7	1.6	0.0	0.0	0.0	0.0	0.0	0.0	0.0	5.7	4.7 53.7	8.2	7. 74
1.5 Payload System 1.6 Flight System	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.7	0.6	1.3	2.1	7.7	9.6	23.1	21.1	12.9	15.2	19.5	18.2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	65.7	53.7	88.8	74.
1.6.2 ATLO (SE&I Contractor)	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	2.3	3.8	6.2	5.5	6.0	7.1	9.1	8.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	30.7	25.1	36.9	30
1.7 Mission Operations System	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.6	0.3	0.8	2.2	1.9	2.7	8.6	8.0	2.7	3.1	4.0	3.7	0.0	0.0	0.0	0.0	0.0	0.0	0.0	13.6	11.1	22.1	19
1.8 Launch Systems 1.9 Technology Development (excl. 1.9.4)	0.0 2.0	0.0	0.0 4.0	0.0 8.8	0.0 11.4	0.0 4.6	0.0 31.7	29.6	2.0	0.0	0.0 4.0	0.0 8.8	0.0 11.4	0.0 4.6	0.0 31.7	29.6	25.9	30.5 0.0	39.3	36.6	0.0	0.0	0.0	0.0	0.0	0.0	0.0	132.4	108.1	132.4 31.7	108 29
1.9.4 Systems Verification	1.2	1.2	1.4	3.7	3.3	3.3	14.1	13.2	1.2	1.2	1.4	3.7	3.3	3.3	14.1	13.2	3.4	3.5	3.6	3.7	0.0	0.0	0.0	0.0	0.0	0.0	0.0	14.1	11.5	28.2	24
1.10 Mission Software	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	1.0	2.8	3.6	10.2	12.9	30.5	27.9	15.8	18.8	23.3	21.4	0.0	0.0	0.0	0.0	0.0	0.0	0.0	79.3	64.8	109.9	92
Subtotal before Reserves Reserves	3.8 0.0	2.4 0.5	5.8 0.8	14.5 3.7	17.6	8.0 11.0	52.1 28.7	48.7 26.1	6.5	6.4 0.5	15.4 0.8	32.2 3.7	56.1 12.6	57.4 30.9	174.0 48.6	160.3 43.7	93.0 27.8	109.0 36.1	138.5 45.7	128.9 51.9	0.0	0.0	0.0	0.0	0.0	0.0	0.0	469.4 161.4	383.4 131.4	643.4 210.0	543 . 175
neserves	0.0	0.5	0.6	3.7	12.0	11.0	20.7	20.1	0.0	0.5	0.6	3.7	12.0	30.9	46.0	43.7	27.0	30.1	45.7	51.9	0.0	0.0	0.0	0.0	0.0	0.0	0.0	101.4	131.4	210.0	175
Total ATD, formulation, and implementation through launch plus 30 days	3.8	3.0	6.6	18.2	30.2	19.0	80.8	74.8	6.5	7.0	16.2	35.9	68.7	88.3	222.6	204.0	120.8	145.1	184.2	180.8	0.0	0.0	0.0	0.0	0.0	0.0	0.0	630.8	514.8	853.4	718.
Launch + 30 Days to End of Mission	Enter each co						6 -1					1											-	c =1				nter each cost e		1	
1.1 Management 1.2 System Engineering/Integration Contractor	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	2.0	1.3	0.7	0.7	0.8	0.8	1.0	7.9	3.5 5.6	5.0 7.9	3.
1.4 Science	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	13.2	13.6	12.7	13.1	10.8	11.1	11.5	86.0	60.0	86.0	60
1.4.4 Education & Public Outreach	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	1.6	1.7	0.9	0.9	0.9	0.8	0.8	7.5	5.3	7.5	5
1.5 Payload System	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	2.5	0.0	0.0	9.5	0.0	0.0	0.0	2.5	1.9	2.5	1.
1.7 Mission Operations System 1.9.4 Systems Verification	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	3.6	9.3	9.6	0.0	0.0	8.6 0.0	8.9 0.0	62.4 3.6	43.4 2.7	62.4	43
1.10 Mission Software	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	3.3	1.4	1.4	1.5	1.5	1.6	1.6	12.2	8.6	12.2	8
Subtotal before Reserves	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	35.7	27.9	26.2	26.5	23.3	23.8	23.9	187.2	131.0	187.2	131
Reserves Total from launch plus 30 days to end	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0 0.0	0.0	0.0	0.0	0.0	0.0	8.4 44.0	4.2 32.1	4.1 30.3	30.7	3.9 27.1	4.0 27.8	3.9 27.8	32.6 219.8	22.9 154.0	32.6 219.8	22 1 54
Total LISA (NASA) cost	3.8	3.0	6.6	18.2	30.2	19.0	80.8	74.8	6.5	7.0	16.2	35.9	68.7	88.3	222.6	204.0	120.8	145.1	184.2	180.8	44.0	32.1	30.3	30.7	27.1	27.8	27.8	850.6	668.8	1,073.2	872
10121 21071 (101071) 0001	0.0	0.0					55.5		0.0	- 1.0				55.5									00.0					555.5		1,010.2	V
FULL COST OF:																															
NASA FTES		0.8	1.3	2.7	4.2	2.3	11.2	8.9		1.7	3.3	5.3	9.4	10.4	30.2	27.7	12.5	15.1	19.5	18.7	4.1	2.3	2.1	2.2	2.2	2.3	2.3	83.3	66.0	113.5	93.
OSS Required Funding for LISA	3.8	2.2	5.3	15.5	26.1	16.7	69.6	65.9	6.5	5.3	12.9	30.5	59.3	77.9	192.4	176.3	108.2	130.0	164.6	162.0	39.9	29.8	28.2	28.5	24.9	25.5	25.5	767.3	602.8	959.7	779
CONTRIBUTIONS OUTSIDE OF UPN 785-30																															
US (NON-LISA) CONTRIBUTIONS																															
Technology Flight Validation (NMP/ST-7)	3.0	14.4	20.0	11.7	3.6	2.8	55.7	53.8	3.0	14.4	20.0	11.7	3.6	2.8	55.7	53.8	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	55.7	53
Total US Cost (NASA)	3.0	14.4	20.0	11.7	3.6	2.8	55.7	53.8	3.0	14.4	20.0	11.7	3.6	2.8	55.7	53.8	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	55.7	53.
ESA CONTRIBUTIONS																															
1.1 Management	0.2	0.1	0.2	0.7	1.1	0.0	2.3	2.2	0.4	0.2	0.6	1.6	2.5	2.3	7.5	6.9	3.2	3.8	4.9	4.5	0.4	0.3	0.3	0.3	0.3	0.3	0.1	18.2	14.7	25.7	21.
1.2 System Engineering/Integration Contractor	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.2	0.4	1.0	2.0	2.1	5.7	5.2	2.4	2.8	3.6	3.3	0.5	0.3	0.2	0.2	0.2	0.3	0.3	14.2	11.3	19.8	16
1.3 Mission Assurance	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.1	0.4	0.5	1.0	1.4	3.3	3.0	1.7	1.9	2.3	2.2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	8.1	6.6	11.4	9.
1.4 Science	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.9	0.5	1.2	2.1	2.7	3.3	10.8		5.0	5.9	7.6	7.1	13.2	13.6	12.7	13.1	10.8	11.1	11.5	111.6	80.9	122.4	90
1.4.4 Education & Public Outreach	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.1	0.1	0.2	0.3	0.4	1.0		0.5	0.5	0.7	0.6	0.7	0.7	0.3	0.4	0.4	0.3	0.3	5.4	4.0	6.4	5
1.5 Payload System	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.7	0.8	1.3	2.7	7.7	9.8			12.9	15.2	19.5		2.5	0.0	0.0	0.0	0.0	0.0	0.0	68.3	55.6	91.3	76
1.6 Flight System																															
1.6.1 Spacecraft (including FEEPs)	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	21.1	38.0	59.1	52.9	82.8	109.8	43.0	32.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	267.7	222.3	326.7	275
1.7 Mission Operations System	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.1	0.1	0.1	0.3	0.3	0.4		-	0.4	0.5	0.6	0.5	1.2	1.3	1.4	1.4	1.2	1.2	1.3	11.0	7.9	12.2	9.
1.9 Technology Development	4.3	2.6	4.0	8.8	11.4	0.0	31.1	29.6	4.3	2.6	4.0	8.8	11.4	0.0			0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	31.1	29
LISA Test Package	5.3	18.2	21.8	13.2	3.6	0.0	62.1	60.5	5.3	18.2	21.8	13.2	3.6	0.0	62.1	60.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	62.1	60
Technology Flight Validation (SMART-2)	7.0	9.0	15.5	17.9	13.2	0.0	62.5	60.0	7.0	9.0	15.5	17.9	13.2	0.0	62.5	60.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	62.5	60
Subtotal before Reserves	16.8	29.9	41.4	40.6	29.3	0.0	158.0	152.3	18.7	31.7	45.3	48.2	65.8	57.6	267.3	251.2	108.8	140.3	82.2	68.5	18.5	16.2	14.9	15.3	12.9	13.2	13.4	504.3	403.4	771.6	654
Reserves	0.0	9.5	13.2	13.9	11.1	0.0	47.7	45.5	0.0	9.5	13.2	13.9	22.7	18.0			30.1	43.1	26.0	26.5	2.0	0.6	0.6	0.6	0.5	0.6	0.5	131.1	107.3	208.5	169
Total ESA Cost	16.8	39.3	54.6	54.5	40.3	0.1	205.7	197.8	18.7	41.2	58.5	62.1	88.5	75.6	344.6	313.7	139.0	183.4	108.2	95.0	20.5	16.8	15.5	15.9	13.5	13.8	13.9	635.4	510.7	980.0	824
Combined Contributions	19.8	53.8	74.7	66.3	44.0	2.9	261.3	251.6	21.7	55.6	78.5	73.8	92.1	78.4	400.3	367.5	139.0	183.4	108.2	95.0	20.5	16.8	15.5	15.9	13.5	13.8	13.9	635.4	510.7	1,035.7	878
Total Mission Cost (All Sources)	23.6	56.8	81.3	84.4	74.2		342.1	326.4	28.2	62.6	94.8	109.7	160.8	166.7		571.5	259.7	328.5	292.3	275.8	64.5	48.8	45.8	46.6	40.6	41.6	41.8	1,486.0	1,179.4	2,108.9	1,751

H.1.2 LISA Funding Profile (Real Year \$M)

The JPL Team X and GSFC MICM cost models were used to arrive at constant year FY03 dollars. These amounts were time phased using standard project funding profiles developed by the GSFC Resources Analysis Office (RAO) who uses a historical database of large GSFC missions. The time phased constant year dollars were then inflated using the index provided by Code S.

Table H-4 shows the LISA Combined Budget. Section H.1.2 includes six pages documenting the process used to spread the budget dollars across the WBS Level 2 elements and fiscal years of the LISA mission. The first three pages show the dollar values spread across the WBS Level 2 elements and fiscal years. The next three pages show the values (percentages) utilized in the calculations to derive the dollar values

For example: under Technology Development, Management element, FY03; the formula is the total budget for Management (see LISA Total Mission section, Page 3) x the percentage of Management dollars estimated for Technology Development in FY03 x the value from the inflation matrix for FY03.

i.e. \$70.3M * 0.0050 * 1.00 = \$0.3M

Table H-4: LISA Combined Budget (\$M)

	LIS	A CON	IBINED	BUDG	ET				
TRIP Report	CY	RY							
NASA Contribution:	FY-03 \$M	\$M	Tech Dev	Pre-A	Phase A	Phase B	Phase C/D	Phase E	Total
Management	59.0	70.3	6.3	1.2	3.1	9.8	44.8	5.0	70.3
System Engineering/Integration Contractor	61.8	74.2		1.2	4.8	15.2	45.2	7.9	74.2
Mission Assurance	14.6	17.3		0.4	0.9	3.7	12.3		17.3
Science	90.9	122.4		1.9	2.8	6.1	25.6	86.0	122.4
Education & Public Outreach	12.3	15.7		0.3	0.7	1.5	5.7	7.5	15.7
Payload System	76.7	91.3		1.9	3.6	17.5	65.7	2.5	91.3
Flight System	0.0	0.0							0.0
ATLO (SE&I Contractor)	30.6	36.9				6.2	30.7		36.9
Mission Operations System	62.4	84.6		1.1	2.8	4.6	13.6	62.4	84.6
Launch Systems	108.1	132.4					132.4		132.4
Technology Development	29.6	31.7	31.7						31.7
Systems Verification	27.5	31.8	14.1				14.1	3.6	31.8
Technology Flight Validation (ST-7)	53.8	55.7	55.7						55.7
Mission Software	101.3	122.1		2.6	4.8	23.1	79.3	12.2	122.1
SUBTOTAL WITHOUT CONTINGENCY:	728.5	886.2	107.8	10.5	23.5	87.8	469.4	187.2	886.2
Contingency (M\$)		242.6	28.7	0.0	0.0	19.9	161.4	32.6	242.6
Contingency (%)		27.4	26.6	0.0	0.0	22.7	34.4	17.4	
NASA TOTAL:	728.5	1,128.8	136.5	10.5	23.5	107.8	630.8	219.8	1,128.8
NOTE: ST-7/DRS sponsored by the NMP.									
ESA Contribution:			See NOTE b	elow:					-
(Includes member state contributions)									
Management	21.6	25.7	2.3	0.4	1.1	3.6	16.4	1.8	25.7
Mission System Engineering	16.5	19.8		0.3	1.3	4.1	12.1	2.1	19.8
Mission Assurance	9.6	11.4		0.2	0.6	2.5	8.1		11.4
Science	90.9	122.4		1.9	2.8	6.1	25.6	86.0	122.4
Education & Public Outreach	5.0	6.4		0.1	0.3	0.6	2.3	3.0	6.4
Member State Payload Contributions	76.7	91.3		1.9	3.6	17.5	65.7	2.5	91.3
Flight System	0.0	0.0							0.0
Spacecraft (including FEEPs)	275.2	326.7				59.1	267.7		326.7
Mission Operations System	9.0	12.2		0.2	0.4	0.7	2.0	9.0	12.2
Launch Systems	0.0	0.0							0.0
Technology Development	29.6	31.1	31.1						31.1
LISA Test Package	60.5	62.1	62.1						62.1
Technology Flight Validation (SMART-2)	60.0	62.5	62.5						62.5
Spacecraft Software	0.0	0.0							0.0
SUBTOTAL WITHOUT CONTINGENCY:	654.6	771.6	158.0	5.0	10.1	94.1	399.8	104.5	771.6
Contingency (M\$)		208.5	47.7	0.0	0.0	29.6	125.7	5.4	208.5
Contingency (%)		27.0	30.2	0.0	0.0	31.5	31.5	5.1	
ESA TOTAL:	654.6	980.0	205.7	5.0	10.1	123.7	525.6	109.9	980.0
TOTAL FOR LISA MISSION:		2,108.9	342.1	15.6	33.7	231.5	1,156.3	329.6	2,108.9
			Consequently	y, European	goods and se	rvices contri	derably across buted to the L itional NASA	JSA Mission	

H-5

LISA Funding Profile - TRIP Real Year (\$M) and Spreading Matrix Percentages (%)

PROGRAM ELEMENTS	Dollars To Spread	II Row II										Launch									
(Includes 3 S/C, plus chem. propulsion modules) TECHNOLOGY DEVELOPMENT	(FY03\$M)	Totals II	PY	FY03	FY04	FY05	FY06	FY07	FY08	FY09	FY10	FY11	FY12	FY13	FY14	FY15	FY16	FY17	FY18	FY19	TOTAL
NASA Contribution:		"																			
Management		6.3	0.6	0.3	0.4	2.0	2.9	0.1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	6.3
Technology Development Systems Verification		31.7 14.1	2.0 1.2	0.9 1.2	4.0 1.4	8.8 3.7	11.4 3.3	4.6 3.3	0.0 0.0	31.7 14.1											
NMP/ST-7 Distrubance Reduction System		55.7 II	3.0	14.4	20.0	11.7	3.6	2.8	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	55.7
SUBTOTAL WITHOUT CONTINGENCY:		107.8	6.8	16.9	25.8	26.2	21.3	10.8	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	107.8
Contingency NASA SUBTOTAL:		28.7 136.5	0.0 6.8	0.5 17.4	0.8 26.7	3.7 29.9	12.6 33.9	11.0 21.8	0.0 0.0	28.7 136.5											
ESA Contribution:		130.5	0.0	17.4	20.7	29.9	33.9	21.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	130.5
Management		2.3	0.2	0.1	0.2	0.7	1.1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	2.3
Technology Development		31.1 62.1	4.3 5.3	2.6 18.2	4.0 21.8	8.8 13.2	11.4 3.6	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0	0.0 0.0	0.0 0.0	31.1 62.1
LISA Test Package SMART-2 Mission		62.1 II	5.3 7.0	9.0	21.6 15.5	17.9	13.2	0.0	0.0	0.0	0.0	0.0	0.0 0.0	0.0	0.0	0.0	0.0	0.0 0.0	0.0	0.0	62.1
SUBTOTAL WITHOUT CONTINGENCY:		158.0 II	16.8	29.9	41.4	40.6	29.3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	158.0
Contingency		47.7	0.0	9.5	13.2	13.9	11.1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	47.7
ESA SUBTOTAL: TECHNOLOGY DEVELOPMENT TOTAL:		205.7 342.1	16.8 23.6	39.3 56.8	54.6 81.3	54.5 84.4	40.3 74.2	0.1 21.9	0.0 0.0	205.7 342.1											
TECHNOLOGY DEVELOPMENT TOTAL.		J42.1 II	20.0	30.0	01.5	04.4	74.2	21.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	042.1
PRE-PHASE A DEFINITION		II																			
NASA Contribution:																					
Management System Engineering/Integration Contractor		1.2 1.2	0.5 0.1	0.3 0.7	0.4 0.4	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	1.2 1.2
Mission Assurance		0.4	0.1	0.7	0.4	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.4
Science		1.9	0.9	0.5	0.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	1.9
Education & Public Outreach Payload System		0.3 1.9	0.0 0.7	0.2 0.8	0.1 0.4	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.3 1.9
Mission Operations System		1.1	0.7	0.8	0.4	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	1.1
Mission Software		2.6 II	0.0	1.0	1.6	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	2.6
SUBTOTAL WITHOUT CONTINGENCY:		10.5	2.7	4.0	3.8	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.5
Contingency NASA SUBTOTAL:		0.0 10.5	0.0 2.7	0.0 4.0	0.0 3.8	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 10.5
ESA Contribution		II																			
Management		0.4	0.2	0.1	0.2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.4
System Engineering Mission Assurance		0.3 II 0.2 II	0.0 0.0	0.2 0.1	0.1 0.1	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.3 0.2
Science		1.9	0.9	0.5	0.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	1.9
Education & Public Outreach		0.1	0.0	0.1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.1
Payload System Mission Operations System		1.9 0.2	0.7 0.1	0.8 0.1	0.4 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	1.9 0.2
Spacecraft Software		0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
SUBTOTAL WITHOUT CONTINGENCY:		5.0	1.9	1.8	1.3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	5.0
Contingency ESA SUBTOTAL:		0.0 II 5.0 II	0.0 1.9	0.0 1.8	0.0 1.3	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 5.0
PRE-PHASE A TOTAL:		15.6 II	4.6	5.8	5.1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	15.6
		II																			
PHASE A DEFINITION		II																			
NASA Contribution: Management		Ⅱ 3.1 Ⅱ	0.0	0.0	0.7	2.3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	3.1
System Engineering/Integration Contractor		4.8	0.0	0.0	1.1	3.6	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	4.8
Mission Assurance		0.9	0.0	0.0	0.2	0.7	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.9
Science Education & Public Outreach		2.8 II 0.7 II	0.0 0.0	0.0 0.0	0.7 0.2	2.1 0.5	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	2.8 0.7
Payload System		3.6 II	0.0	0.0	0.2	0.5 2.7	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	3.6
Mission Operations System		2.8	0.0	0.0	0.6	2.2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	2.8
Mission Software		4.8	0.0	0.0	1.3	3.6	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	4.8
SUBTOTAL WITHOUT CONTINGENCY: Contingency		23.5 0.0	0.0 0.0	0.0 0.0	5.8 0.0	17.7 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	23.5 0.0
NASA SUBTOTAL:		23.5	0.0	0.0	5.8	17.7	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	23.5
ESA Contribution			0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	
Management System Engineering		1.1 1.3	0.0 0.0	0.0 0.0	0.3 0.3	0.8 1.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	1.1 1.3
Mission Assurance		0.6	0.0	0.0	0.3	0.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.6
Science S. P. His Q. Land		2.8	0.0	0.0	0.7	2.1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	2.8
Education & Public Outreach Payload System		0.3 II 3.6 II	0.0 0.0	0.0 0.0	0.1 0.9	0.2 2.7	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.3 3.6
Mission Operations System		0.4	0.0	0.0	0.9	0.3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.4
Spacecraft Software		0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
SUBTOTAL WITHOUT CONTINGENCY:		10.1	0.0	0.0	2.6	7.6	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	10.1
Contingency ESA SUBTOTAL:		0.0 10.1	0.0 0.0	0.0 0.0	0.0 2.6	0.0 7.6	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 10.1
PHASE A TOTAL:		33.7	0.0	0.0	8.4	25.3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	33.7
		II																			
PHASE B DEFINITION:																					
NASA Contribution:		II																			

LISA Funding Profile - TRIP Real Year (\$M) and Spreading Matrix Percentages (%)

								-													
PROCEAN ELEMENTS	Dollars											Land									
PROGRAM ELEMENTS (Includes 3 S/C, plus chem. propulsion modules)	To Spread (FY03\$M)	Row II Totals II	PY	FY03	FY04	FY05	FY06	FY07	FY08	FY09	FY10	Launch FY11	FY12	FY13	FY14	FY15	FY16	FY17	FY18	FY19	TOTAL
Management	, ,	9.8	0.0	0.0	0.0	0.0	3.8	6.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	9.8
System Engineering/Integration Contractor Mission Assurance		15.2 3.7	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	7.5 1.6	7.7 2.2	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	15.2 3.7
Science		6.1	0.0	0.0	0.0	0.0	2.7	3.3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	6.1
Education & Public Outreach		1.5	0.0	0.0	0.0	0.0	0.6	0.9	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	1.5
Payload System		17.5	0.0	0.0	0.0	0.0	7.7	9.8	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	17.5
Flight System: ATLO Mission Operations System		6.2 4.6	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	2.3 1.9	3.8 2.7	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	6.2 4.6
Mission Software		23.1	0.0	0.0	0.0	0.0	10.2	12.9	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	23.1
SUBTOTAL WITHOUT CONTINGENCY:		87.8 II	0.0	0.0	0.0	0.0	38.4	49.4	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	87.8
Contingency NASA SUBTOTAL:		19.9 107.8	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 38.4	19.9 69.3	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	19.9 107.8
ESA Contribution:		107.011	0.0	0.0	0.0	0.0	00.1	00.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	107.0
Management		3.6	0.0	0.0	0.0	0.0	1.4	2.2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	3.6
System Engineering Mission Assurance		4.1 2.5	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	2.0 1.0	2.1 1.4	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	4.1 2.5
Science		6.1	0.0	0.0	0.0	0.0	2.7	3.3	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	6.1
Education & Public Outreach		0.6	0.0	0.0	0.0	0.0	0.3	0.4	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.6
Payload System		17.5	0.0	0.0	0.0	0.0	7.7	9.8	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	17.5
Spacecraft Mission Operations System		59.1 0.7	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	21.1 0.3	38.0 0.4	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	59.1 0.7
Spacecraft Software		0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
SUBTOTAL WITHOUT CONTINGENCY:		94.1	0.0	0.0	0.0	0.0	36.6	57.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	94.1
Contingency ESA SUBTOTAL:		29.6 123.7	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	11.6 48.2	18.0 75.6	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	29.6 123.7
PHASE B TOTAL:		231.5	0.0	0.0	0.0	0.0	86.6	144.9	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	231.5
PHASE C/D		II II																			
NASA Contribution:		"																			
Management		44.8	0.0	0.0	0.0	0.0	0.0	0.0	8.8	10.3	13.3	12.4	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	44.8
System Engineering/Integration Contractor		45.2 II	0.0	0.0	0.0	0.0	0.0	0.0	8.8	10.4	13.4	12.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	45.2
Mission Assurance Science		12.3 25.6	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	2.5 5.0	2.9 5.9	3.5 7.6	3.3 7.1	0.0 0.0	0.0 0.0	12.3 25.6						
Education & Public Outreach		5.7	0.0	0.0	0.0	0.0	0.0	0.0	1.1	1.3	1.7	1.6	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	5.7
Payload System		65.7 II	0.0	0.0	0.0	0.0	0.0	0.0	12.9	15.2	19.5	18.2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	65.7
Flight System: ATLO Mission Design & Operations		30.7 13.6	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	6.0 2.7	7.1 3.1	9.1 4.0	8.5 3.7	0.0 0.0	0.0 0.0	30.7 13.6						
Launch Systems		132.4	0.0	0.0	0.0	0.0	0.0	0.0	25.9	30.5	39.3	36.6	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	132.4
Systems Verification		14.1	0.0	0.0	0.0	0.0	0.0	0.0	3.4	3.5	3.6	3.7	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	14.1
Mission Software		79.3	0.0	0.0	0.0	0.0	0.0	0.0	15.8	18.8	23.3	21.4	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	79.3
SUBTOTAL WITHOUT CONTINGENCY: Contingency		469.4 161.4	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	93.0 27.8	109.0 36.1	138.5 45.7	128.9 51.9	0.0 0.0	0.0 0.0	469.4 161.4						
NASA SUBTOTAL:		630.8 II	0.0	0.0	0.0	0.0	0.0	0.0	120.8	145.1	184.2	180.8	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	630.8
ESA Contribution:		10.4 !!	2.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	4.0	4.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	40.4
Management System Engineering		16.4 12.1	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	3.2 2.4	3.8 2.8	4.9 3.6	4.5 3.3	0.0 0.0	0.0 0.0	16.4 12.1						
Mission Assurance		8.1	0.0	0.0	0.0	0.0	0.0	0.0	1.7	1.9	2.3	2.2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	8.1
Science		25.6	0.0	0.0	0.0	0.0	0.0	0.0	5.0	5.9	7.6	7.1	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	25.6
Education & Public Outreach Payload System		2.3 II 65.7 II	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.5 12.9	0.5 15.2	0.7 19.5	0.6 18.2	0.0 0.0	0.0 0.0	2.3 65.7						
Spacecraft		267.7 II	0.0	0.0	0.0	0.0	0.0	0.0	82.8	109.8	43.0	32.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	267.7
Mission Operations System		2.0	0.0	0.0	0.0	0.0	0.0	0.0	0.4	0.5	0.6	0.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	2.0
Launch Systems LISA Test Package		0.0 II 0.0 II	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0
Spacecraft Software		0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
SUBTOTAL WITHOUT CONTINGENCY:		399.8	0.0	0.0	0.0	0.0	0.0	0.0	108.8	140.3	82.2	68.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	399.8
Contingency		125.7 525.6	0.0	0.0	0.0	0.0	0.0	0.0	30.1	43.1	26.0	26.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	125.7
ESA SUBTOTAL: PHASE C/D TOTAL:		1156.3	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	139.0 259.7	183.4 328.5	108.2 292.3	95.0 275.8	0.0 0.0	0.0 0.0	525.6 1156.3						
		II																			
PHASE E: NASA Contribution:		II II																			
Management		5.0 II	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	1.2	0.7	0.7	0.7	0.8	0.8	0.2	0.0	5.0
System Engineering/Integration Contractor		7.9	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	2.0	1.3	0.9	0.9	0.9	0.9	1.0	0.0	7.9
Science		86.0 II	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	13.2	13.6	12.7	13.1	10.8	11.1	11.5	0.0	86.0
Education & Public Outreach Payload System		7.5 2.5	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	1.6 2.5	1.7 0.0	0.9 0.0	0.9 0.0	0.9 0.0	0.8 0.0	0.8 0.0	0.0 0.0	7.5 2.5
Mission Operations System		62.4	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	2.5 8.2	9.3	9.6	9.5	8.4	8.6	8.9	0.0	2.5 62.4
Systems Verification		3.6	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	3.6	0.0	0.0	0.0	0.0	0.0	0.0	0.0	3.6
Mission Software		12.2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	3.3	1.4	1.4	1.5	1.5	1.6	1.6	0.0	12.2
SUBTOTAL WITHOUT CONTINGENCY: Contingency		187.2 32.6	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	35.7 8.4	27.9 4.2	26.2 4.1	26.5 4.1	23.3 3.9	23.8 4.0	23.9 3.9	0.0 0.0	187.2 32.6
NASA SUBTOTAL:		219.8	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	44.0	32.1	30.3	30.7	27.1	27.8	27.8	0.0	219.8
ESA Contribution:		II																			
Management		1.8	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.4	0.3	0.3	0.3	0.3	0.3	0.1	0.0	1.8

LISA Funding Profile - TRIP Real Year (\$M) and Spreading Matrix Percentages (%)

						icai i cai	ψivi) alla	opicadii	ig watin	ı ercenta <u></u>	ges (/o)										
	Dollars	II																			
PROGRAM ELEMENTS	To Spread	Row II										Launch									
(Includes 3 S/C, plus chem. propulsion modules)	(FY03\$M)	Totals II	PY	FY03	FY04	FY05	FY06	FY07	FY08	FY09	FY10	FY11	FY12	FY13	FY14	FY15	FY16	FY17	FY18		TOTAL
System Engineering Science		2.1 86.0	0.0 0.0	0.0 0.0	0.0 0.0	0.5 13.2	0.3 13.6	0.2 12.7	0.2 13.1	0.2 10.8	0.3 11.1	0.3 11.5	0.0 0.0	2.1 86.0							
Education & Public Outreach		3.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.7	0.7	0.3	0.4	0.4	0.3	0.3	0.0	3.0
Payload System		2.5 II	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	2.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	2.5
Mission Operations System		9.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	1.2	1.3	1.4	1.4	1.2	1.2	1.3	0.0	9.0
LISA Test Package Spacecraft Software		0.0 II 0.0 II	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0							
SUBTOTAL WITHOUT CONTINGENCY:		104.5 II	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	18.5	16.2	14.9	15.3	12.9	13.2	13.4	0.0	104.5
Contingency		5.4 II	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	2.0	0.6	0.6	0.6	0.5	0.6	0.5	0.0	5.4
ESA SUBTOTAL: PHASE E TOTAL:		109.9 329.6	0.0 0.0	0.0 0.0	0.0 0.0	20.5 64.5	16.8 48.8	15.5 45.8	15.9 46.6	13.5 40.6	13.8 41.6	13.9 41.8	0.0 0.0	109.9 329.6							
LISA TOTAL MISSION:		Phases																			
NASA Contributiuon:																					
Management	59.0	70.3	1.0	0.6	1.6	4.3	6.7	6.1	8.8	10.3	13.3	12.4	1.2	0.7	0.7	0.7	0.8	0.8	0.2	0.0	70.3
System Engineering/Integration Contractor Mission Assurance	61.8 14.6	74.2 17.3	0.1 0.0	0.7 0.1	1.5 0.4	3.6 0.7	7.5 1.6	7.7 2.2	8.8 2.5	10.4 2.9	13.4 3.5	12.5 3.3	2.0 0.0	1.3 0.0	0.9 0.0	0.9 0.0	0.9 0.0	0.9 0.0	1.0 0.0	0.0 0.0	74.2 17.3
Science	90.9	122.4	0.9	0.5	1.2	2.1	2.7	3.3	5.0	5.9	7.6	7.1	13.2	13.6	12.7	13.1	10.8	11.1	11.5	0.0	122.4
Education & Public Outreach	12.3	15.7	0.0	0.2	0.3	0.5	0.6	0.9	1.1	1.3	1.7	1.6	1.6	1.7	0.9	0.9	0.9	0.8	0.8	0.0	15.7
Payload System	76.7	91.3	0.7	0.8	1.3	2.7	7.7	9.8	12.9	15.2	19.5	18.2	2.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	91.3
Flight System ATLO (SE&I Contractor)	0.0 30.6	0.0 Ⅱ 36.9 Ⅱ	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 2.3	0.0 3.8	0.0 6.0	0.0 7.1	0.0 9.1	0.0 8.5	0.0 0.0	0.0 36.9							
Mission Operations System	62.4	84.6	0.6	0.3	0.8	2.2	1.9	2.7	2.7	3.1	4.0	3.7	8.2	9.3	9.6	9.5	8.4	8.6	8.9	0.0	84.6
Launch Systems	108.1	132.4	0.0	0.0	0.0	0.0	0.0	0.0	25.9	30.5	39.3	36.6	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	132.4
Technology Development Systems Verification	29.6 27.5	31.7 Ⅱ 31.8 Ⅱ	2.0	0.9 1.2	4.0 1.4	8.8 3.7	11.4	4.6	0.0 3.4	0.0 3.5	0.0 3.6	0.0 3.7	0.0 3.6	0.0 0.0	0.0 0.0	0.0 0.0	0.0	0.0 0.0	0.0 0.0	0.0	31.7 31.8
Technology Flight Validation	53.8	55.7	1.2 3.0	1.2 14.4	20.0	3.7 11.7	3.3 3.6	3.3 2.8	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0 0.0	0.0	0.0	0.0 0.0	55.7
Mission Software	101.3	122.1	0.0	1.0	2.8	3.6	10.2	12.9	15.8	18.8	23.3	21.4	3.3	1.4	1.4	1.5	1.5	1.6	1.6	0.0	122.1
SUBTOTAL WITHOUT CONTINGENCY:	728.5	886.2	9.6	20.9	35.4	43.9	59.7	60.2	93.0	109.0	138.5	128.9	35.7	27.9	26.2	26.5	23.3	23.8	23.9	0.0	886.2
Contingency NASA TOTAL:		242.6 1128.8	0.0 9.6	0.5 21.4	0.8 36.2	3.7 47.6	12.6 72.3	30.9 91.1	27.8 120.8	36.1 145.1	45.7 184.2	51.9 180.8	8.4 44.0	4.2 32.1	4.1 30.3	4.1 30.7	3.9 27.1	4.0 27.8	3.9 27.8	0.0 0.0	242.6 1128.8
European Contributions:		II																			
Management	21.6	25.7	0.4	0.2	0.6	1.6	2.5	2.3	3.2	3.8	4.9	4.5	0.4	0.3	0.3	0.3	0.3	0.3	0.1	0.0	25.7
Mission System Engineering Mission Assurance	16.5 9.6	19.8 ∥ 11.4 ∥	0.0 0.0	0.2 0.1	0.4 0.3	1.0 0.5	2.0 1.0	2.1 1.4	2.4 1.7	2.8 1.9	3.6 2.3	3.3 2.2	0.5 0.0	0.3 0.0	0.2 0.0	0.2 0.0	0.2 0.0	0.3 0.0	0.3 0.0	0.0 0.0	19.8 11.4
Science	90.9	122.4	0.9	0.5	1.2	2.1	2.7	3.3	5.0	5.9	7.6	7.1	13.2	13.6	12.7	13.1	10.8	11.1	11.5	0.0	122.4
Education & Public Outreach	5.0	6.4	0.0	0.1	0.1	0.2	0.3	0.4	0.5	0.5	0.7	0.6	0.7	0.7	0.3	0.4	0.4	0.3	0.3	0.0	6.4
Member State Payload Contributions	76.7	91.3	0.7	0.8	1.3	2.7	7.7	9.8	12.9	15.2	19.5	18.2	2.5	0.0	0.0	0.0	0.0	0.0	0.0	0.0	91.3
Flight System Spacecraft (including FEEPs)	0.0 275.2	0.0 Ⅱ 326.7 Ⅱ	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 21.1	0.0 38.0	0.0 82.8	0.0 109.8	0.0 43.0	0.0 32.0	0.0 0.0	0.0 326.7							
Mission Operations System	9.0	12.2	0.1	0.1	0.1	0.3	0.3	0.4	0.4	0.5	0.6	0.5	1.2	1.3	1.4	1.4	1.2	1.2	1.3	0.0	12.2
Launch Systems	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
Technology Development LISA Test Package	29.6 60.5	31.1 ∥ 62.1 ∥	4.3 5.3	2.6 18.2	4.0 21.8	8.8 13.2	11.4 3.6	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	31.1 62.1
Technology Flight Validation (SMART-2)	60.0	62.5	7.0	9.0	15.5	17.9	13.2	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	62.5
Spacecraft Software	0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0	0.0
SUBTOTAL WITHOUT CONTINGENCY:	654.6	771.6 Ⅱ 208.5 Ⅱ	18.7 0.0	31.7 9.5	45.3 13.2	48.2	65.8 22.7	57.6 18.0	108.8	140.3 43.1	82.2 26.0	68.5 26.5	18.5 2.0	16.2 0.6	14.9 0.6	15.3 0.6	12.9 0.5	13.2 0.6	13.4 0.5	0.0 0.0	771.6 208.5
Contingency ESA TOTAL:		980.0	18.7	9.5 41.2	58.5	13.9 62.1	88.5	75.6	30.1 139.0	183.4	108.2	26.5 95.0	20.5	16.8	15.5	15.9	13.5	13.8	13.9	0.0	980.0
TOTAL FOR LISA MISSION:		2108.9	28.2	62.6	94.8	109.7	160.8	166.7	259.7	328.5	292.3	275.8	64.5	48.8	45.8	46.6	40.6	41.6	41.8	0.0	2108.9
INFLATION MATRIX:		ii	0.000	1 000	1 001	1.000	1.000	1 100	1.105	1 001	1.000	1 077	1.010	1.057	1 000	1 440	1 407	1.500	1 501	1 000	
Constant FY-03\$		11	0.969 1.000	1.000 1.000	1.031 1.000	1.063 1.000	1.096 1.000	1.130 1.000	1.165 1.000	1.201 1.000	1.238 1.000	1.277 1.000	1.316 1.000	1.357 1.000	1.399 1.000	1.442 1.000	1.487 1.000	1.533 1.000	1.581 1.000	1.630 1.000	
Real Year \$		11	0.969	1.000	1.031	1.063	1.006	1.130	1.165	1.201	1.238	1.277	1.316	1.357	1.399	1.442	1.487	1.533	1.581	1.630	
CONTINGENCY MATRIX:		II II																			
NASA:																					
Management	0.3	ii.																			
System Engineering/Integration Contractor	0.4	II																			
Mission Assurance Science	0.1 0.0																				
Education & Public Outreach	0.0																				
Payload System	0.5	ii																			
Flight System	0.0	11																			
ATLO Mission Operations System	0.4 0.3																				
Launch Vehicle	0.3																				
Technology Development	0.5	ii																			
Systems Verification	0.5																				
Technology Flight Validation Mission Software	0.5 0.5	II II																			
Mission Soπware	0.5	11																			

0.3

Management

ESA:

LISA Funding Profile - TRIP Real Year (\$M) and Spreading Matrix Percentages (%)

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	Dollars	II																			
PROGRAM ELEMENTS	To Spread	Row II										Launch									
(Includes 3 S/C, plus chem. propulsion modules)	(FY03\$M)	Totals II	PY	FY03	FY04	FY05	FY06	FY07	FY08	FY09	FY10	FY11	FY12	FY13	FY14	FY15	FY16	FY17	FY18	FY19	TOTAL
System Engineering Mission Assurance	0.4 0.1																				
Science	0.1	ii																			
Education & Public Outreach	0	ii																			
Member State Payload Contributions	0.5	II																			
Flight System	0.0	II																			
Spacecraft (including FEEPs)	0.3	II																			
Mission Operations System	0.3	II II																			
Launch Vehicle	0.0 0.5	II II																			
Technology Development LISA Test Package	0.3	" "																			
Technology Flight Validation (SMART-2)	0.3	ii ii																			
Spacecraft Software	0.5	II																			
CDDE A DING MATRIX.		II																			
SPREADING MATRIX:		II																			
TECHNOLOGY DEVELOPMENT		II																			
NASA Contribution:			0.0400	0.0050	0.0070	0.0000	0.0450	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.4040
Management Technology Development		0.1010 1.0000	0.0100 0.0700	0.0050 0.0300	0.0070 0.1300	0.0320 0.2800	0.0450 0.3520	0.0020 0.1380	0.0000 0.0000	0.0000	0.1010 1.0000										
Systems Verification		0.4800 II	0.0700	0.0300	0.1500	0.2500	0.3320	0.1050	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.4800
NMP/ST-7 Distrubance Reduction System		1.0000	0.0581	0.2684	0.3607	0.2051	0.0615	0.0462	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	1.0000
ESA Contribution:		II																			
Management		0.1010	0.0100	0.0050	0.0070	0.0320	0.0450	0.0020	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1010
Technology Development		1.0000	0.1500	0.0880	0.1300	0.2800	0.3520	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	1.0000
LISA Test Package		1.0000	0.0900	0.3000	0.3500	0.2051	0.0549	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	1.0000
SMART-2 Mission		1.0000 II	0.1200	0.1500	0.2500	0.2800	0.2000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	1.0000
PRE-PHASE A DEFINITION		 II																			
NASA Contribution:		ii																			
Management		0.0200 II	0.0080	0.0050	0.0070	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0200
System Engineering/Integration Contractor		0.0190	0.0010	0.0120	0.0060	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0190
Mission Assurance		0.0250	0.0000	0.0100	0.0150	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0250
Science		0.0207	0.0100	0.0057	0.0050	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0207
Education & Public Outreach Payload System		0.0234 0.0250	0.0000 0.0100	0.0159 0.0100	0.0075 0.0050	0.0000 0.0000	0.0000	0.0000 0.0000	0.0000	0.0000 0.0000	0.0000 0.0000	0.0000	0.0234 0.0250								
Mission Operations System		0.0230 II 0.0186 II	0.0100	0.0100	0.0030	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000 0.0000	0.0000	0.0000	0.0000	0.0230
Mission Software		0.0250	0.0000	0.0100	0.0150	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0250
ESA Contribution		II																			
Management		0.0200 II	0.0080	0.0050	0.0070	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0200
System Engineering		0.0190	0.0010	0.0120	0.0060	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0190
Mission Assurance		0.0250	0.0000	0.0100	0.0150	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0250
Science Education & Public Outreach		0.0207 0.0234	0.0100 0.0000	0.0057 0.0159	0.0050 0.0075	0.0000 0.0000	0.0000	0.0000 0.0000	0.0000	0.0207 0.0234											
Payload System		0.0254 II	0.0100	0.0100	0.0073	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0254
Mission Operations System		0.0186	0.0100	0.0056	0.0030	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0186
Spacecraft Software		0.0250 II	0.0000	0.0100	0.0150	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0250
DUAGE A DEFINITION		II 																			
PHASE A DEFINITION		II II																			
NASA Contribution: Management		0.0490 II	0.0000	0.0000	0.0120	0.0370	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0490
System Engineering/Integration Contractor		0.0734	0.0000	0.0000	0.0120	0.0554	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0734
Mission Assurance		0.0570	0.0000	0.0000	0.0120	0.0450	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0570
Science		0.0295 II	0.0000	0.0000	0.0080	0.0215	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0295
Education & Public Outreach		0.0545	0.0000	0.0000	0.0140	0.0405	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0545
Payload System		0.0450	0.0000	0.0000	0.0120	0.0330	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0450
Mission Operations System Mission Software		0.0430	0.0000	0.0000 0.0000	0.0100	0.0330 0.0330	0.0000 0.0000	0.0000	0.0000 0.0000	0.0000	0.0000 0.0000	0.0000	0.0000	0.0000 0.0000	0.0000	0.0000 0.0000	0.0000	0.0000	0.0000 0.0000	0.0000	0.0430 0.0450
ESA Contribution		0.0450 ll	0.0000	0.0000	0.0120	0.0330	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0450
Management		0.0490 II	0.0000	0.0000	0.0120	0.0370	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0490
System Engineering		0.0734	0.0000	0.0000	0.0180	0.0554	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0734
Mission Assurance		0.0570 II	0.0000	0.0000	0.0120	0.0450	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0570
Science		0.0295	0.0000	0.0000	0.0080	0.0215	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0295
Education & Public Outreach		0.0545	0.0000	0.0000	0.0140	0.0405	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0545
Payload System		0.0450	0.0000	0.0000	0.0120	0.0330	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0450
Mission Operations System Spacecraft Software		0.0430 0.0450	0.0000 0.0000	0.0000 0.0000	0.0100 0.0120	0.0330 0.0330	0.0000 0.0000	0.0000	0.0430 0.0450												
•		ii																			
PHASE B DEFINITION:		II																			
NASA Contribution:		II																			
Management Control of the Control of		0.1495	0.0000	0.0000	0.0000	0.0000	0.0593	0.0902	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1495
System Engineering/Integration Contractor Mission Assurance		0.2210 0.2300	0.0000 0.0000	0.0000	0.0000 0.0000	0.0000 0.0000	0.1105 0.0988	0.1105	0.0000 0.0000	0.0000	0.2210 0.2300										
Science		0.2300 II 0.0600 II	0.0000	0.0000	0.0000	0.0000	0.0988	0.1312 0.0324	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.2300
Education & Public Outreach		0.1125	0.0000	0.0000	0.0000	0.0000	0.0460	0.0665	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1125
Payload System		0.2050 II	0.0000	0.0000	0.0000	0.0000	0.0920	0.1130	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.2050

LISA Funding Profile - TRIP
Real Year (\$M) and Spreading Matrix Percentages (%)

							(, , , , ,		5		3 (,										
	Dollars	The state of the s																			
PROGRAM ELEMENTS	To Spread	Row II										Launch									
(Includes 3 S/C, plus chem. propulsion modules)	(FY03\$M)	Totals II	PY	FY03	FY04	FY05	FY06	FY07	FY08	FY09	FY10	FY11	FY12	FY13	FY14	FY15	FY16	FY17	FY18	FY19	TOTAL
Flight System: ATLO		0.1800	0.0000	0.0000	0.0000	0.0000	0.0700	0.1100	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1800
Mission Operations System		0.0660 II	0.0000	0.0000	0.0000	0.0000	0.0276	0.0384	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0660
Mission Software		0.2050 II	0.0000	0.0000	0.0000	0.0000	0.0920	0.1130	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.2050
ESA Contribution: Management		0.1495	0.0000	0.0000	0.0000	0.0000	0.0593	0.0902	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1495
System Engineering		0.1495	0.0000	0.0000	0.0000	0.0000	0.0593	0.0902	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1495
Mission Assurance		0.2300 II	0.0000	0.0000	0.0000	0.0000	0.0988	0.1312	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.2300
Science		0.0600 II	0.0000	0.0000	0.0000	0.0000	0.0276	0.0324	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0600
Education & Public Outreach		0.1125	0.0000	0.0000	0.0000	0.0000	0.0460	0.0665	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1125
Payload System		0.2050 II	0.0000	0.0000	0.0000	0.0000	0.0920	0.1130	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.2050
Spacecraft		0.1921	0.0000	0.0000	0.0000	0.0000	0.0700	0.1221	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1921
Mission Operations System		0.0660 II	0.0000	0.0000	0.0000	0.0000	0.0276	0.0384	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0660
Spacecraft Software		0.2050	0.0000	0.0000	0.0000	0.0000	0.0920	0.1130	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.2050
PHASE C/D																					
NASA Contribution: Management		0.6205 II	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1278	0.1458	0.1824	0.1645	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.6205
System Engineering/Integration Contractor		0.5966 II	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1276	0.1436	0.1624	0.1543	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.5966
Mission Assurance		0.6880 II	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1223	0.1402	0.1754	0.1381	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.6880
Science		0.2298	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0473	0.0540	0.0676	0.0609	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.2298
Education & Public Outreach		0.3796 II	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0782	0.0892	0.1116	0.1006	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.3796
Payload System		0.7000 II	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1442	0.1645	0.2058	0.1855	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.7000
Flight System: ATLO		0.8200 II	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1689	0.1927	0.2411	0.2173	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.8200
Mission Operations System		0.1774	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0365	0.0417	0.0522	0.0470	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1774
Launch Systems		1.0000 II	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.2060	0.2350	0.2940	0.2650	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	1.0000
Technology Development		0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
Systems Verification		0.4200	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1050	0.1050	0.1050	0.1050	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.4200
Mission Software ESA Contribution:		0.6400	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1342	0.1545	0.1858	0.1655	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.6400
Management		0.6205 II	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1278	0.1458	0.1824	0.1645	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.6205
System Engineering		0.5966 II	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1279	0.1402	0.1024	0.1581	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.5966
Mission Assurance		0.6880	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1492	0.1645	0.1958	0.1785	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.6880
Science		0.2298 II	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0473	0.0540	0.0676	0.0609	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.2298
Education & Public Outreach		0.3796 II	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0782	0.0892	0.1116	0.1006	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.3796
Payload System		0.7000 II	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1442	0.1645	0.2058	0.1855	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.7000
Spacecraft		0.8079 II	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.2584	0.3323	0.1262	0.0910	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.8079
Mission Operations System		0.1774	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0365	0.0417	0.0522	0.0470	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1774
Launch Systems		0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
Technology Development		0.0000 II 0.0000 II	0.0000	0.0000	0.0000 0.0000	0.0000 0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000 0.0000	0.0000 0.0000	0.0000 0.0000	0.0000	0.0000 0.0000	0.0000	0.0000 0.0000	0.0000	0.0000	0.0000 0.0000
LISA Test Package Spacecraft Software		0.6400 II	0.0000 0.0000	0.0000	0.0000	0.0000	0.0000	0.0000 0.0000	0.0000 0.1342	0.0000 0.1545	0.0000	0.0000	0.0000	0.0000	0.0000 0.0000	0.0000	0.0000 0.0000	0.0000	0.0000	0.0000 0.0000	0.6400
opusosian conmune		0.0100 11	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1012	0.1010	0.1000	0.1000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0100
PHASE E:		ii																			
NASA Contribution:		ï																			
Management		0.0600 II	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0150	0.0086	0.0086	0.0086	0.0086	0.0086	0.0020	0.0000	0.0600
System Engineering/Integration Contractor		0.0900 II	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0250	0.0150	0.0100	0.0100	0.0100	0.0100	0.0100	0.0000	0.0900
Science		0.6600 II	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1100	0.1100	0.1000	0.1000	0.0800	0.0800	0.0800	0.0000	0.6600
Education & Public Outreach		0.4300 II	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1000	0.1000	0.0500	0.0500	0.0500	0.0400	0.0400	0.0000	0.4300
Payload System		0.0250	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0250	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0250
Mission Operations System		0.6950	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1000	0.1100	0.1100	0.1050	0.0000	0.0900	0.0900	0.0000	0.6950
Systems Verification Mission Software		0.1000 0.0850	0.0000 0.0000	0.0000	0.0000 0.0000	0.0000 0.0000	0.0000 0.0000	0.0000 0.0000	0.0000 0.0000	0.0000 0.0000	0.0000	0.0000 0.0000	0.1000 0.0250	0.0000 0.0100	0.0000 0.0100	0.0000 0.0100	0.0000 0.0100	0.0000 0.0100	0.0000 0.0100	0.0000 0.0000	0.1000 0.0850
ESA Contribution:		0.0030 11	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0230	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0000	0.0050
Management		0.0600 II	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0150	0.0086	0.0086	0.0086	0.0086	0.0086	0.0020	0.0000	0.0600
System Engineering		0.0900 II	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0250	0.0150	0.0100	0.0100	0.0100	0.0100	0.0100	0.0000	0.0900
Science		0.6600 II	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1100	0.1100	0.1000	0.1000	0.0800	0.0800	0.0800	0.0000	0.6600
Education & Public Outreach		0.4300 II	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1000	0.1000	0.0500	0.0500	0.0500	0.0400	0.0400	0.0000	0.4300
Payload System		0.0250 II	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0250	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0250
Mission Operations System		0.6950 II	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.1000	0.1100	0.1100	0.1050	0.0900	0.0900	0.0900	0.0000	0.6950
LISA Test Package		0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
Spacecraft Software		0.0850	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0250	0.0100	0.0100	0.0100	0.0100	0.0100	0.0100	0.0000	0.0850
		11 J1																			
		" 																			
Total Program		NASA II	ESA																		
Management		1.0000 II	1.0000																		
System Engineering/Integration Contractor		1.0000 II	1.0000																		
Mission Assurance		1.0000 II	1.0000																		
Science		1.0000 II	1.0000																		

Flight System

ATLO

Education & Public Outreach Payload System

Mission Design & Operations Launch System 1.0000 1.0000 1.0000

0.0000

1.0000

0.0000

1.0000 Spacecraft

1.0000 || 1.0000 || 0.0000 ||

1.0000 ||

1.0000 ||

1.0000 II

Laser Interferometer Space Antenna

LISA Funding Profile - TRIP Real Year (\$M) and Spreading Matrix Percentages (%)

	Dollars	II.																			
PROGRAM ELEMENTS	To Spread	Row II										Launch									
(Includes 3 S/C, plus chem. propulsion modules)	(FY03\$M)	Totals II	PY	FY03	FY04	FY05	FY06	FY07	FY08	FY09	FY10	FY11	FY12	FY13	FY14	FY15	FY16	FY17	FY18	FY19	TOTAL
Technology Development		1.0000	1.0000)																	
Systems Verification		1.0000	1.0000	LISA Test P	ackage																
Technology Flight Validation		1.0000 II	1.0000	SMART-II M	lission																
Mission Software		1.0000 II	1.0000) S/C Softwar	e																
Contingency		II																			
		II																			

H.1.3 Independent Cost Estimate

The independent cost estimate prepared by the JPL Systems Management Office is shown in Table H-5.

The assumptions are based on the schedules identified in Section H.2:

- Phase A activities will begin in February 2004 lasting for ~ 18 months.
- Phase B activities will begin in August 2005 lasting for ~ 2 years.
- Phases C/D will begin in October 2007 lasting ~ 4 years, with launch planned for August 2011.
- Detailed design is ~ 12 months
- Manufacturing is ~ 18 months
- Integration and testing is ~ 9 months
- ATLO is \sim 9 months
- The costing focused on the first 3 activities in Phase C/D for the instruments and spacecrafts using a combination of models and analogy. Only internal JPL models and databases were used at this time. External sources (Aerospace, SAIC, MCR, etc) will be used when the project requires an update to the cost estimate prior to entering Phase B and Phase C/D.
- Phases A and B costs were determined as a percentage of the Phases C/D costs, as was ATLO in Phases C/D. The Phase B costs and reserve (50%) are higher than normal missions because of the technology development activities that need to be completed prior to entering Phases C/D.
- Phases C/D reserves are ~ 30%, no reserve is carried on the ELV.
- Phase E is estimated at 6 years total, 1 year checkout/commissioning and 5 years operations; the reserve is 15%.
- The cost has a ±15% uncertainty resulting in a cost range of \$680M to \$922M before reserves. Reserves of \$207M were calculated as specified above on the mid-point of the cost range

and added to the cost range to give a total mission cost range of \$888M to \$1128M in FY'03\$ for Phases A thru E. The midpoint with reserves is \$1008M.

	Costs in FY03 \$	M with reserves
	JPL ICE	LISA Estimate
Phase A/B	146.1	222.6
Phase C/D	733.0	630.8
Phase E	129.0	219.8
Total	1,008.1	1,073.2

LISA Project Response:

The LISA Project estimate compares favorably with the JPL ICE at the total level. This agreement was expected since the JPL Team X Model and the GSFC MICM models were the basis for the LISA budget. The differences in the formulation and implementation phase estimates are due to the LISA Project's emphasis on early planning and technology development as compared to the average project represented by the JPL Team X Model. Since the LISA estimate is well within the uncertainty range of the ICE, no reconciliation of the two estimates is necessary.

Table H-5: LISA Independent Cost Estimate

Schedule Dura	ations			
Phase A/B	3.5 Yrs			18 mo. Phase A, 24 mo. Phase B
Phase C/D	4 Yrs			
Phase E	6 Yrs			1 yr checkout, 5 years ops
Launch	08/2011			
Costs in FY03 \$M				
Phase A/B	\$97.4		Without reserves	
Phase A	12.2			
Phase B	85.2			
Reserves		48.7		
Phase C/D	591.8		Without reserves	
Mgmt	9.5		2% of Dev – L/V-Res-Mgmt	
Sys Eng	15.8		5% of S/C + Inst.	
MA	18.7		4% of Dev – L/V-Res	
Science	4.0		\$200k * duration * FTE	5 FTEs for Phase C/D timeframe
Spacecraft	165.0		Models / Analogy	
Instrument	150.0		Model	
Software	10.0		Starlight number	
ALTO	113.8		Analogy	35% of the S/C, Inst, & S/W
ELV	105.0		Lookup Table for Team X	
Reserves		141.2	30% of Dev – L/V – Res	30% of C/D costs – ELV & Res
Phase E	112.2		Without reserves	
Mgmt	10.2		10% of OpsSci + MODA	
Science	12.0		\$200K * duration * FTE	10 FTEs for Phase E timeframe
MO&DA	90.0		Model	
Reserves		16.8	15% of Ops – Ops Res	15%
Totals	801.3		Without Reserves	

Total Cost Range	\$887.9 – \$1,128.3		
Total Reserves	\$206.7		Calculated on mid-point of cost range
Cost Range	\$681.1 - \$921.5	Without Reserves	Represents an uncertainty of ± -15%

H.1.4 NASA Workforce

Table H-6 shows the NASA workforce estimate. The LISA staffing plan shows the full time equivalent (FTE) civil servant staffing that relate to the LISA funding profile, using the current loaded labor rates, inflated by the Code S inflation index.

Section H.1.3 includes two pages documenting the process used to estimate FTEs and spreading them across the WBS Level 2 elements and fiscal years of the LISA mission. These pages show: the dollar values from the budget contained in Appendix H.1.2; the number of FTEs spread across the WBS Level 2 elements and fiscal years; and the values utilized in the formulas for calculating the full cost of the FTEs.

Table H-6: NASA Workforce

		FOR	MULA	TION						II.	VIPLE	VIENT.	ATION					TOTAL
	<u>PY</u>	FY03	<u>FY04</u>	<u>FY05</u>	<u>FY06</u>	<u>FY07</u>	<u>FY08</u>	<u>FY09</u>	<u>FY10</u>	<u>FY11</u>	<u>FY12</u>	<u>FY13</u>	<u>FY14</u>	<u>FY15</u>	<u>FY16</u>	<u>FY17</u>	<u>FY18</u>	
FTE	4.9	8.5	16.0	25.0	43.0	46.0	53.8	62.9	78.8	73.4	15.6	8.3	7.6	7.6	7.3	7.4	7.2	473.4
RY\$		1.7	3.3	5.3	9.4	10.4	12.5	15.1	19.5	18.7	4.1	2.3	2.1	2.2	2.2	2.3	2.3	113.5
CY\$		1.7	3.2	5.0	8.6	9.2	10.8	12.6	15.8	14.7	3.1	1.7	1.5	1.5	1.5	1.5	1.4	93.7

H.1.5 NASA Facilities

Table H-7 shows the cost of NASA facilities and equipment used for LISA. These are located at the GSFC and will be used during the Integration & Test activity. The facilities and equipment are located in:

- Building 7, the Payload Test Facility
- Building 10, the Environmental Test Laboratory
- Building 29, the Spacecraft Systems Development & Integration Facility

Table H-7: LISA Observatory & Constellation NASA Facility & Equipment Requirements (\$K)

Test	Facility	Days	Cost / month	FY08	FY09	FY10	FY11	Total Cost
Cleanroom Acquisition	SSDIF B29	761	43.3	201.8	624.0	643.3	442.4	1911.4
EMI/EMC	EMI Facility/B7	50	1.9		22.8	94.1		116.9
TV Instrumentation	SES Chamber B10		223.8		268.8			268.8
TV Fixturing	SES Chamber B10		335.7		403.2			403.2
TV Dry Run	SES Chamber B10		55.95		67.2			67.2
Vibration Test	Vibration facility B10	60	7.4		88.9	458.1		546.9
Thermal Vac Test	SES Chamber B10	180	300.0		720.6	1485.6		2206.2
Observatory Shock Test	SES Chamber B10	20	5		30.0	92.9		122.9
Total I&T Costs				201.8	2225.5	2773.9	442.4	5,643.5

H.1.6 Master Equipment List

LISA Master Equipment List, Detailed Mass and Power Budget spreadsheet follows:

	CBE + Contin.		431.86	1.30	1.30	00.00	14.30	00.00	6.50	00.7	32.50	32.50	110.50	31.20	1.30	78.00	00.00	0 00	800		63.18	1	20.54	00.0	00.00	1.04	20.80		0.00	00.51	0.00	00.00	0.00	3	0.00	0.00
	Growth Cont. (W)		99.66	0:30	0.30	0.00	3.30	00.00	1.50	06.1	7.50	7.50	25.50	7.20	0.30	18.00	0.00	0000	0.00		14.58	•	4.74	00.0	00.00	0.24	4.80		0.00	8.5	0.00	00.00	0000	0.0	0.00	00.0
_	Power Growth Cont. (%)		30%	30%	30%	30%	30%	30%	30%	3029	30%	30%	30%	30%	30%	30%	30%	30%	20%		30%	,000	30%	30%	30%	30%	30%		30%	8,00	#DIV/0	30%	30%	808	#DIV/0	30%
Power	CBE Total		332.20	1.00	1.00	0.00	11.00	00:00	5:00	0.00	25.00	25.00	85.00	24.00	1.00	00:09	00.00	00.00	0.0		48.60	0	15.80	00.0	00'0	0.80	16.00	;	0000	000	0.00	00.00	00.0	86.0	0.00	000
	Each		Total	Total	1.00	0.00	Total	00:00	5.00	0.0	Total	25.00	Total	12.00	1.00	30.00	0.00	0.00	00:0		Total	-	2.86	0.00	0.00	0.80	2.67		0.00	20.0	Total	00:00	00.00	8	Total	000
	CBE + Contin.	3562.31	401.66	79.86	89.70	8.97	37.50	8:38	13.00	10:12	20.67	20.67	25.48	9.10	1,30	3.64	00'0	7.80	2.60		24.08	i.	1.82	0.21	0.26	0.23	13.81		1.30	00.5	14.30	5.20	7.80	00:-	27.30	02.00
	Mass Growth Cont. (kg)	384.45	92.69	22.77	20.70	2.07	8.65	1.93	3.00	37.6	4.77	4.77	5.88	2.10	0.30	0.84	00:00	1.80	09.0	10000	5.56	000	0.42	0.05	90.0	0.05	3.19		0.30	20.0	3.30	1.20	1.80	00.0	6.30	000
Mass	Mass Growth Cont. (%)	12%	30%	30%	30%	30%	30%	30%	30%	30%	30%	30%	30%	30%	30%	30%	30%	30%	30%		30%	7000	30%	30%	30%	30%	30%		%08	8,00	30%	30%	30%	30.20	30%	7008
	CBE Total	3093.00	308.97	75.90	00.69	06'9	28.85	6.45	10.00	12.40	15.90	15.90	19.60	7.00	1.00	2.80	00:00	00.00	200		18.52	000	1.40	0.16	0.20	0,18	10.62		8.8	200	11.00	4.00	0.90	3	21.00	00 10
	Each		Total	Total	00.69	06.9	Total	6.45	10.00	12.40	Total	15.90	Total	3.50	1.00	1.40	0.75	3.00	2.00	STREET, STREET	Total	1	0.70	0.08	0.20	0.18	1.77		0.50	3	Total	4.00	9.00	3	Total	00 10
	Off,				-	-		-		-		-		2	-	8	0	0 4	-			•	* 0	8	-	-	φ		N C	V		-				
	Component	-ISA Flight System (3 Wet Spacecraft and Launch Vehicle Adapter)			Structure & Mechanism	Secondary Structure		Solar Array	Battery	Fower controller Distribution offic (PCDO)		CPS		Transponders	RFDU	SSPA	TWT	HG Antenna	Cabling			Star Camera Assembly	SC Electronics	Sun Sensor	Att Anomaly Detector	Magnetometer	FEEP EI. Clusters	HGA Drive	HD Mechanism			MLI Blankets	Radiators	SILIE		Cables & Hames
	Sub-system	ystem (3 Wet Spacecraft.	Science Module (Spacecraft +Payload)	Structures & Mechanisms			Power				Command & Data Handling		Telecom								Attitude Control										Thermal Control				Cabling	
1	Mission Element	LISA Flight S	Science Mod																																	

Carowth Confin. Conf							Mass				Power	Ļ		
Accounty	Mission	Sub-evetam	Component	Ë			Mass Groudh	Mass	CBE			_	Power	, and
According to the processor According to t	Element	Sub-system	nie iodino	Qt		CBE Total		Gont. (kg)	Contin.	Each	CBE Total	Gont (%)		Contin.
Assembly Training	Payload				Total	118.20		35.46	153.66	Total	161.60	30%		210.08
Company		Optical Assembly			Total	66.30	30%	19.89	86.19	Total	124.60	30%	37.38	161.98
Control Burnel			Telescope	7	6.50	13.00	30%	3.90	16.90	0.00	00.00	30%	00:00	00.00
Control Cont			Optical Bench	N	5.60	11.20	30%	3.36	14.56	4.50	9.00	30%	2.70	11.70
LASSER Electrical Locations LASS			OA Structure	cı.	2.00	10.00	30%	3.00	13.00	0.00	0.00	30%	0.00	0.00
Location Further Electronics 2 200 400 30% 120 570			LASER Head	4	2.00	8.00	30%	2.40	10.40	0.00	0.00	30%	00:00	0.00
Figure F			LASER Electronics	2	2.00	4.00	30%	1.20	5.20	37.00	74.00	30%	22.20	96.20
			Fiber Positioner	8	0.30	09:0	30%	0.18	0.78	0.00	00'00	30%	00:00	00.00
Figure 10 Figu			Interferometer Electronics											
Figure F			IE Front Unit	0	1.50	3.00	30%	0.90	3.90	5.20	10.40	30%	3.12	13.52
Underlanger Company			IE Digital Unit	7	3.50	2.00	30%	2.10	9.10	8:00	16.00	30%	4.80	20.80
Or Vision House Control Electronics 1			nso	N O	2.00	00.4	30%	1.20	5.20	0.00	25.50	30%	0.36	1,56
Intering Reference Serior Total 19,000 30% 5,70 24,70 Total 11,00 30% 11,00 30			UV Discharger Instrument Control Electronics	N -	0.50	0.1.00	% % % % %	1.35	1.30	9 90 8 8	9 00	30%	2.40	7.80 10.40
Institute February Structure Fig. 1 1,00														
Interfit Between Service Trainit Between Service Fig. 1 100 10		Gravitational Reference Sen	nsor Assy		Total	19.00	30%	5.70	24.70	Total	11.00	30%	3.30	14.30
Hotelanderson			Inertial Reference Sensor	,		00.07	2000	000				i	0	000
Mechanisms			Inertial Sensor	N C	000	13.00	30%	3.90	16.90	0.00	00.00	30%	0.00	0.00
According Structure			Mechanisms	v -	800	8 6	% % % %	090	2.60	00.5	1.00	30%	0.30	30 GE
Victorium														
V-Structure		Y-Tube Assembly			Total	15.00	30%	4.50	19.50	Total	1.00	30%	0:30	1.30
Control Blankers			Y-Structure	- ,	13.00	13.00	30%	3.90	16.90	0.00	0.00	30%	0.00	0 0 7
Application Bankets Total Total 1.00 1.00 30% 0.00 1.00 0.0			Mechanisms	-	80.5	8	30%	0.60	7.00	8:	8.	%O%	0.30	<u>.</u>
Paints P		Thermal Control			Total	2.00	30%	09.0	2.60	Total	0.00	#DIV/0!	0.00	00.0
Total 15.90 30% 4.77 20.67 Total 15.90 30% 4.77 20.67 Total 25.00 30% 30% 4.77 20.67 Total 21.240 30% 4.77 20.67 Total 25.00 30% 25.00 30% 25.00 30% 25.00 30% 25.00 30% 25.00 30% 25.00 20% 25.00 30% 25.00 25.00 30% 25.00 25.00 30% 25.00 25.00 20% 25.00 20% 25.00 20% 25.00 20% 25.00 20% 25.00 20% 25.00 20% 25.00 20%			Blankets	-	00.1	0.1	30%	0.30	1,30	0.00	0.00	30%	00:00	0.00
Total 15.90 30% 4.77 20.67 Total 25.00 30% 4.77 20.67 Total 25.00 30% 30% 4.77 20.67 20.67 25.00 30% 30% 25.00 30% 25.00 30% 25.00 30% 25.00 30% 25.00 30% 25.00 30% 25.00 30% 25.00 30% 25.00 30% 25.00 30% 25.00 30% 25.00 25.00 30% 25.00 25.00 30% 25.00 25.00 30% 25.00 25.00 30% 25.00 2			Paints	-	00.	8.	30%	0:30	1.30	0.00	0.00	30%	0.00	0.00
CPS Total 1590 1590 30% 477 2067 2500 2500 30% 30% 477 2067 2500 2500 30% 470		Payload Processor			Total	15.90	30%	4.77	20.67	Total	25.00	30%	7.50	32.50
Total 101,00 30% 63.75 276.24 Total 264,00 30% 63.75 276,24 Total 264,00 30% 63.05 Fulmary Structure			CPS	-	15.90	15.90	30%	4.77	20.67	25.00	25.00	30%	7.50	32.50
Primary Structure 1 Total 101,00 30% 30.30 151,30 Total 0.00 #DIV/OI Propulsion Secondary Structure 1 51.00 51.00 30% 15.30 60.30 60.00 0.00 0.00 30% Propulsion Hydrazine Fuel Tank 2 81.0 16.20 30% 25.62 111.02 10.00 0.00 30% 3	Propulsion M	odule			Total	212.49	30%	63.75	276.24	Total	264.00	30%	79.20	343.20
Primary Structure		Structure			Total	101.00	30%	30.30	131,30	Total	0.00	#DIA/0i	0.00	0.00
High Propulsion			Primary Structure Secondary Structure		50.00	50.00	30%	15.30	96.30	0000	0.00	30%	00.0	9 G 6 C
Presultarian														
Oxidized Tanks 2 567 11.34 30% 4.86 21.05 0.00 0.00 30% Pressurant Tank for Evel 1 1.400 14.00 30% 4.20 14.20 0.00 0.00 0.00 30% Pressurant Tank for Evel 1 7.40 30% 4.20 14.20 0.00 0.00 0.00 30% Pressurant Tank for Evel 1 7.40 4.00 30% 1.20 0.00 0.00 30% Brook Pressurant Tank for Modizer 1 7.40 4.00 30% 1.20 5.00 0.00 0.00 30% Brook Components & Pressurant Tank for Modizer 1 5.70 4.00 30% 1.59 68.00 9.00 30% Flow Components & Pressurant Tank for Modizer 1 5.30 30% 1.59 68.90 9.00 0.00 30% ACS 0.1N thrusters 8 0.50 4.00 30% 2.21 1.20 5.20 1.50 30%		Chemical Propulsion			Total	85.40	30%	25.62	111.02	Total	98.00	30%	29.40	127.40
Pressurant Tank for Fuel 1 14.00 14.00 30% 4.20 16.20 0.00 0.00 30% Pressurant Tank for Doxidizer 1 7.40 14.00 14.00 30% 4.20 16.20 0.00 0.00 30% BP-op Timester 2DT 2 2.00 4.00 30% 1.20 5.00 0.00 0.00 30% Flow Components & Pipework 1 27.16 27.16 27.16 27.16 30% 8.15 58.31 98.00 30% Brackets and Haware 1 5.30 5.30 30% 1.59 68.9 0.00 30% ACS 0.1N thrusters 8 0.50 4.00 30% 1.20 52.0 1.50 30% Isolation Values 17 0.57 9.69 30% 2.40 10.40 0.00 30% Brackets Hamasen 1 0.40 30% 2.40 10.40 0.00 30%			Hydrazine Fuel Tank Oxidizer Tanks	01 0	8.10	11.34	30%	3.40	14.74	0000	0.00	%0% 30%	000	9 6 6
Presurent Tank for Oxidizer			Pressurant Tank for Fuel		14.00	14.00	30%	4.20	18.20	0.00	0.00	30%	00:0	00.0
Birdyep Thrusters 22N Eackers and Harvate Each Components & Pipework 1 2716 27.16 30% 1.50 0.00 0.00 30% Brackers and Harvate 1 2716 27.16 30% 1.50 0.00 0.00 30% Brackers and Harvate 1 5.30 5.30 0.00 0.00 0.00 Brackers and Harvate 1 5.30 5.30 0.00 0.00 0.00 Brackers and Harvate 1 5.30 5.30 0.00 0.00 0.00 Brackers and Harvate 1 5.30 5.30 0.00 0.00 0.00 Brackers Harvate 1 5.30 5.30 Brackers Harvate 1 5.30			Pressurant Tank for Oxidizer	-	7.40	7.40	30%	2.22	9.62	0.00	00:00	30%	00.00	00.00
How Components & Ppework 1 27.16 27.16 30% 8.15 35.31 98.00 98.00 30% 30% Brackets and Hanvare 1 5.30 5.30 30% 1.59 68.0 0.00 0.00 30% 1.50 1.50 1.50 1.50 1.50 1.50 1.50 1.50			BiProp Thrusters 22N	5	5.00	4.00	30%	1.20	5.20	00:00	0.00	30%	0.00	0.00
ACS 0.1N thrusters 8 0.55 0.40 0.30% 1.20 5.20 1.40 0.00 0.00 0.00 0.00 0.00 0.00 0.0			Flow Components & Pipework Brookets and Hanners		5.30	5.30	30%	8.15	35.31	98.00	98.00	30%	29.40	127.40
ACS 0.1N thrusters 8 0.50 4.00 30% 0.63 29.72 Total 131.00 30% 1.20 1.20 9.00% 1.20 9.00			בומכונס מות ומואמוס		2	3	8	200	9	3	200	8	3	3
Nithrusiers 8 050 400 30% 120 520 1.50 12.00 30% 120 12.00 30% 120 12.00 30% 14.00 10.00 0.00 0.00 30% 14.00 10.00 0.00 0.00 30% 14.00 0.00 0.00 0.00 30% 14.00 0.00 0.00 0.00 30% 14.00 0.00 0.00 0.00 30% 15.00 0.00 0.00 30% 15.00 0.00 0.00 30% 15.00 0.00 0.00 30% 15.00 0.00 0.00 30% 15.00 0.00 0.00 0.00 30% 15.00 0.00 0.00 0.00 0.00 30% 15.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0		ACS			Total	55.09	30%	6.63	28.72	Total	131.00	30%	39.30	170.30
Valves 11 U.57 9.09 30% 2.40 10.40 0.00 0.00 30% 30% 0.41 0.40 0.40 0.40 0.00 0.00 30%			ACS 0.1N thrusters	ø !	0.50	0.4	30%	1.20	5.20	1.50	12.00	30%	3.60	15.60
1 040 040 000 000 000 30%			Isolation valves	> -	8.00) () () ()	30%	2.81	10.40	00:00	0.00	30%	000	00.0
			Brackets/Hamessing	. ,	0.70	8 0	30%	0.10	0.52	888	800	808	0000	8 6 6

3						Mass				Power			
Mission Element	Sub-system	Component	Otty	Each (CBE Total	Mass Growth Cont. (%)	Mass Growth Cont. (kg)	CBE +	Each	CBE Total	Power Growth Cont. (%)	Power Growth Cont. (W)	CBE + Contin.
	Thermal Control MLI Bia Heaters Paints	MLI Blankets Heaters Paints		Total 2.00 1.00 1.00	4.00 2.00 1.00	%06 %06 %06 %06	1.20 0.60 0.30 0.30	5.20 2.60 1.30 1.30	Total 0.00 35.00 0.00	35.00 0.00 35.00 0.00	30% 30% 30%	10.50 0.00 10.50 0.00	45.50 0.00 45.50 0.00
	Propellant L/V Adapter (5%) (kg)			Total Total	1359.00 169.63	%0 %0	0.00	1359.00 169.63	Total Total	0.00	%0 %0	0.00	0.00
	Science Module Mass (kg)				926.90	30%	278.07	1204.97	Payload Power	161.60	30%	48.48	210.08
	Prop Module Mass (kg)				637.47	30%	191.24	828.71	S\C Bus Power	170.60	30%	51.18	221.78
	Dry Flight System Mass (kg)				1564.37	30%	469.31	2033,68	Prop + S/C Bus Power	434.60	30%	130.38	564.98
	Propellant (kg)				1359.00	%0	0.00	1359.00	Science Module Power	332.20	30%	99'66	431.86
	Wet Flight System Mass (kg)				2923.37	16%	469.31	3392.68					
_	TOTAL LAUNCH MASS (kg)				3093.00	12%	384.45	3562.31	Science Module Power with 8% loss in wiring harness	8% loss in wiring has	rness	_	466.41
	Science Mass Fraction					22.67%			Square Meters of Array needed (for cruise phase bus power with 8% line loss)	ed with 8% line loss)		_	3.45
LAUNCH VEHIC Delta 4 Medium	LAUNCH VEHICLE CAPABILITY Delta 4 Medium Performance Margin (kg) %	4003 440.69 11.01%							Square Meters of Array needed (for operations phase Science Module power with 8% line loss)	ed e Module power witl	h 8% line loss)	_	2.64

H.1.7 Work Breakdown Structure (WBS)

The LISA Project Work Breakdown Structure (WBS) is a product-oriented family tree division of the hardware, software, services and other unique effort associated with the LISA Project. It represents the way the work will be performed on the project and correlates directly to the LISA Supporting, Detailed Schedules (see Section H.2) and the LISA Formulation and Implementation Budgets (see Section H.1). This WBS is refined during the Pre-Formulation Stage.

1.0 LISA Total Project

1.1 Project Management

- 1.1.1 Project Management
- 1.1.2 Project Scientist
- 1.1.3 Business Management
- 1.1.4 Scheduling
- 1.1.5 Procurement Management
- 1.1.6 Project Support

1.2 System Engineering

- 1.2.1 System Engineering Management
- 1.2.2 Requirements
- 1.2.3 System Validation & Verification
- 1.2.4 Operations Concept Definition
- 1.2.5 System Architecture & Design
- 1.2.6 Interfaces and ICDs
- 1.2.7 Software System Engineering
- 1.2.8 Configuration Management
- 1.2.9 Risk Management
- 1.2.10 System Integration & Test (Support)

1.3 Mission Assurance

- 1.3.1 Mission Assurance Management
- 1.3.2 System Safety
- 1.3.3 Contamination Control
- 1.3.4 IV&V
- 1.3.5 Quality Assurance

1.4 Science

1.4.1	Mission S	Science Office
	1.4.1.1	Mission Scientist
	1.4.1.2	Science Team
	1.4.1.3	Science Management Plan
	1.4.1.4	Science Requirements
	1.4.1.5	Astrophysics & Waveform of Sources
	1.4.1.6	Data Analysis Methods
	1.4.1.7	Science Community Interface
1.4.2	Science	Operations
	1.4.2.1	Operations Support
	1.4.2.2	Scientific Performance Characterization
1.4.3	Science	Data Support
	1.4.3.1	Data Analysis
	1.4.3.2	Data Archiving & Distribution
	1.4.3.3	Participating Scientists

1.5 Payload System

1.4.4

1.5.1 Payload System Engineering / Management

Education & Public Outreach

- 1.5.2 Optical Assembly
 - 1.5.2.1 Optical Bench Assembly
 - 1.5.2.2 CCD Sensor
 - 1.5.2.3 Phase Detector
 - 1.5.2.4 Laser
 - 1.5.2.5 Optical Assembly Structure
 - 1.5.2.6 Star Tracker
 - 1.5.2.7 Telescope
 - 1.5.2.8 Optical Assembly Actuator
- 1.5.3 Gravitational Reference Sensor (GRS)
 - 1.5.3.1 Flight Test Mass
 - 1.5.3.2 Caging
 - 1.5.3.3 Housing & Electronics (Lightweight Test Mass)
 - 1.5.3.4 Charge Management Unit
 - 1.5.3.5 GRS Integration & Test

		1.5.3.5.1	GRS EM I&T
		1.5.3.5.2	GRS FM1 I&T
		1.5.3.5.3	GRS FM2 I&T
		1.5.3.5.4	GRS FM3 I&T
1.5.4	Y-Tube A	Assembly	
	1.5.4.1	Aft Fiber	
	1.5.4.2	Structures	
1.5.5	Payload	Processor / C	Controller
1.5.6	Payload	Integration &	Test
	1.5.6.1	Optical Ber	nch Assembly Initial Integration (Dummy GRS)
		1.5.6.1.1	Set-Up Integration Facility & GSE
		1.5.6.1.2	Optical Bench Assembly EM I&T
		1.5.6.1.3	Optical Bench Assembly FM1 I&T
		1.5.6.1.4	Optical Bench Assembly FM2 I&T
		1.5.6.1.5	Optical Bench Assembly FM3 I&T
	1.5.6.2	Electro-Op	tical Integration (Dummy GRS)
		1.5.6.2.1	Electro-Optical EM I&T
		1.5.6.2.2	Electro-Optical FM1 I&T
		1.5.6.2.3	Electro-Optical FM2 I&T
		1.5.6.2.4	Electro-Optical FM3 I&T
	1.5.6.3	Optical Ass	sembly Integration (Dummy GRS)
		1.5.6.3.1	Optical Assembly EM I&T
		1.5.6.3.2	Optical Assembly FM1 I&T
		1.5.6.3.3	Optical Assembly FM2 I&T
		1.5.6.3.4	Optical Assembly FM3 I&T
	1.5.6.4	Optical Ass	sembly Final Integration (Flight GRS)
		1.5.6.4.1	Optical Assembly EM Final I&T
		1.5.6.4.2	Optical Assembly FM1 Final I&T
		1.5.6.4.3	Optical Assembly FM2 Final I&T
		1.5.6.4.4	Optical Assembly FM3 Final I&T
	1.5.6.5	Y-Tube Inte	egration
		1.5.6.5.1	Y-Tube EM I&T
		1.5.6.5.2	Y-Tube FM1 I&T
		1.5.6.5.3	Y-Tube FM2 I&T
		1.5.6.5.4	Y-Tube FM3 I&T

1.6 Flight System

1.6.1

Spacecraf	t (ESA)				
1.6.1.1	ESA Acquis	ition Cycle			
1.6.1.2	Spacecraft	System Engine	ering / Ma	ınagemen	t
1.6.1.3	Spacecraft I	Bus			
	1.6.1.3.1	Structures / Me	echanism	ıs Subsyst	em
		1.6.1.3.1.1	Structur	e & Mech	anism
		1.6.1.3.1.2	Second	ary Structi	ure
	1.6.1.3.2	Power Subsys	tem		
		1.6.1.3.2.1	Solar A	rray	
		1.6.1.3.2.2	Battery		
		1.6.1.3.2.3	Power Unit	Control	Distribution
	1.6.1.3.3	Command & D	ata Hand	dling Subs	ystem
	1.6.1.3.4	Telecom Subs	ystem		
		1.6.1.3.4.1	Transpo	onders	
		1.6.1.3.4.2	RFDU		
		1.6.1.3.4.3	SSPA		
		1.6.1.3.4.4	TWT		
		1.6.1.3.4.5	High Ga	ain Antenn	а
		1.6.1.3.4.6	Low Ga	in Antenna	a
		1.6.1.3.4.7	Telecon	n Cabling	
	1.6.1.3.5	Attitude Contro	ol Subsys	tem	
		1.6.1.3.5.1	Star Optical	Camera Head & El	Assembly ectronics
		1.6.1.3.5.2	Sun Se	nsor	
		1.6.1.3.5.3	Aft Ano	maly Dete	ctor
		1.6.1.3.5.4	Magnet	ometer	
		1.6.1.3.5.5	FEEP A	ssembly	
		1.6.1.3.5.6		ain Anteni Electronic	na Assembly s
	1.6.1.3.6	Thermal Subs	ystem		
	1.6.1.3.7	Cabling			
	1.6.1.3.8	Spacecraft Into	egration 8	& Test	

1.6.1.3.8.1

Spacecraft Bus EM I&T

			1.6.1.3.8.2	Spacecraft Bus FM1 I&I
			1.6.1.3.8.3	Spacecraft Bus FM2 I&T
			1.6.1.3.8.4	Spacecraft Bus FM3 I&T
	1.6.1.4	Propulsion	Module	
		1.6.1.4.1	Structure	
			1.6.1.4.1.1	Primary Structure
			1.6.1.4.1.2	Secondary Structure
		1.6.1.4.2	Chemical Pro	ppulsion
			1.6.1.4.2.1	Fuel Tanks
			1.6.1.4.2.2	Pressurant Tanks
			1.6.1.4.2.3	Filters, Valves & Regulators
			1.6.1.4.2.4	RCS Thruster Assembly
			1.6.1.4.2.5	Heaters
		1.6.1.4.3	Integration &	Test
			1.6.1.4.3.1	Propulsion Module EM I&T
			1.6.1.4.3.2	Propulsion Module FM1 I&T
			1.6.1.4.3.3	Propulsion Module FM2 I&T
			1.6.1.4.3.4	Propulsion Module FM3 I&T
	1.6.1.5	Spacecraft	: I&T (Bus & Pro	opulsion Module)
		1.6.1.5.1	Spacecraft E	M I&T
		1.6.1.5.2	Spacecraft F	light 1 I&T
		1.6.1.5.3	Spacecraft F	light 2 I&T
		1.6.1.5.4	Spacecraft F	light 3 I&T
1.6.2	Observat	tory Integration	on & Test	
	1.6.2.1	Observator	ry EM I&T	
	1.6.2.2	Observato	ry Flight 1 I&T	
	1.6.2.3	Observator	ry Flight 2 I&T	
	1.6.2.4	Observato	ry Flight 3 I&T	
1.6.3	Constella	ation Testing		
1.6.4	Final Inte	egration (Pre-	Launch)	
1.6.5	Launch C	Campaign		
	1.6.5.1	Constellati	on Launch Site	Operations
	1.6.5.2	Constellati	on / Launch Ve	hicle Operations

1.7 Mission Operations System

- 1.7.1 Mission Ops Management
- 1.7.2 Mission Ops System Engineering
- 1.7.3 GDS Supporting Elements (Non-MOS GDS efforts)
- 1.7.4 Operations (Pre-Launch)
- 1.7.5 Missions Operations System V&V / Readiness Testing

1.8 Launch System

- 1.8.1 Spacecraft-To-Launch Vehicle Interface Definition, Verification & Coordination
- 1.8.2 Launch Vehicle Build, Storage & Pre-Ship Operations
- 1.8.3 Launch Services / Launch Site Operations

1.9 Technology

- 1.9.1 Technology Management
- 1.9.2 Disturbance Reduction System
 - 1.9.2.1 Gravitational Reference Sensor
 - 1.9.2.1.1 Housing Subsystem
 - 1.9.2.1.2 Vacuum Subsystem
 - 1.9.2.1.3 Caging Subsystem
 - 1.9.2.1.4 Proof Mass Subsystem
 - 1.9.2.1.5 Proof Mass Control Subsystem
 - 1.9.2.1.6 Sensing & Forcing Subsystem
 - 1.9.2.1.7 Charge Control Subsystem
 - 1.9.2.1.8 I&T Mechanisms
 - 1.9.2.1.9 Prototype GRS (Lab Verified)
 - 1.9.2.2 Thrusters
 - 1.9.2.2.1 Emitter
 - 1.9.2.2.2 Neutralizer
 - 1.9.2.2.3 Thruster Electronics
 - 1.9.2.2.4 Prototype Thruster
 - 1.9.2.3 DRS Controls
 - 1.9.2.3.1 Algorithms
 - 1.9.2.3.2 Simulator Test

1.9.3	Interferon	netry Measur	ement System
	1.9.3.1	Laser(s)	
	1.9.3.2	Laser/Clock	Noise Cancellation
	1.9.3.3	Stabilization	า
	1.9.3.4	Phase Mete	er
	1.9.3.5	Ultra Stable	Structures
	1.9.3.6	IMS Tester	
	1.9.3.7	Prototype II	MS
1.9.4	System V	erification	
	1.9.4.1	Integrated N	Modeling
		1.9.4.1.1	Modeling Environment
		1.9.4.1.2	Quasi-Static Models
		1.9.4.1.3	Dynamic Models
		1.9.4.1.4	Phase Propagation Models
		1.9.4.1.5	End-To-End Models
	1.9.4.2	Test Bed To	echnology
		1.9.4.2.1	Phase 1 - Approaches
		1.9.4.2.2	Phase 2 - Development

1.10 Mission Software

1.10.1	Software Management
1.10.2	Software Requirements Document
1.10.3	Software Test Plan
1.10.4	Spacecraft Flight Software
1.10.5	Ground Data System (Mission Ops System)
1.10.6	Science Data System Software
1.10.7	IV&V

H.1.8 WBS Dictionary

While the LISA Work Breakdown Structure presented in Section H.1.7 represents a **subdivision** of the total work scope for the LISA Project, the Work Breakdown Structure Dictionary is provided to specifically **define** the work scope for each element. The work scope definition provided by the Work Breakdown Structure Dictionary ensures effective communication and mutual understanding by the LISA Team members of the work to be performed. It is refined during the Pre-Formulation Stage.

Table H-8: WBS Dictionary

WBS	DESCRIPTION	WBS DICTIONARY
1.0	LISA Total Project	This element summarizes the project management, system engineering, mission assurance, science, payload system, flight system, mission operations system, launch system, technology, and mission software effort for the entire LISA project.
1.1	Project Management	This element summarizes the effort for the LISA Project Management Office and includes: project management, project scientist, business management, scheduling, procurement management, and project support.
1.1.1	Project Management	This element contains the project management effort and includes all labor, subcontracts, materials and other direct costs.
1.1.2	Project Scientist	This element contains the effort for the project scientist at GSFC and includes coordination with the mission scientist to ensure the science requirements are met during the life of the mission. Includes all labor, subcontracts, materials and other direct costs.
1.1.3	Business Management	This element contains the business management effort (resource analysis) and includes all labor, subcontracts, materials and other direct costs.
1.1.4	Scheduling	This element contains the effort to develop, maintain and control the end-to-end LISA Project schedules and includes all labor, subcontracts, materials and other direct costs.
1.1.5	Procurement Management	This element contains the effort to manage and maintain external contracts and subcontracts with vendors, suppliers and team members. Includes all labor, subcontracts, materials and other direct costs.
1.1.6	Project Support	This element contains all other project support effort not identified in the other project management WBS elements such as logistics, graphics, ITAR coordination, MIS, etc. It includes all labor, subcontracts, materials and other direct costs.
1.2	System Engineering	This element summarizes the system engineering management, requirements, system validation & verification, operations concept definition, system architecture & design, interfaces & ICDs, S/W system engineering, configuration management, risk management and system I&T support effort.

WBS	DESCRIPTION	WBS DICTIONARY
1.2.1	System Engineering Management	This element contains the effort to manage the LISA system engineering process and to develop and maintain the System Engineering Management Plan, maintain a System Architect Team, conduct major system-level reviews, and develop and maintain communications and system engineering tools. Includes all labor, subcontracts, materials and other direct costs.
1.2.2	Requirements	This element contains the effort to define and maintain Level 1 and System Requirements, develop and maintain a requirements database, prepare the Disposal, Re-Entry & Decommission Plan, and prepare Environmental and Orbital Debris Assessments. Includes all labor, subcontracts, material and other direct costs.
1.2.3	System Validation & Verification	This element contains the effort to develop and maintain a project wide verification process of system interactions and a process to ensure that project requirements are met. Includes all labor, subcontracts, materials and other direct costs.
1.2.4	Operations Concept Definition	This element contains the effort to define and maintain an operations concept and includes all labor, subcontracts, materials and other direct costs.
1.2.5	System Architecture & Design	This element contains the effort to conduct system architecture trade studies, develop the architecture & design specification, and maintain the integrated modeling effort after the conclusion of the technology development program. Includes all labor, subcontracts, materials and other direct costs.
1.2.6	Interfaces and ICDs	This element contains the effort to develop and update ICDs and includes all labor, subcontracts, materials and other direct costs.
1.2.7	Software System Engineering	This element contains the effort to maintain flight software interface control documents between the subsystems on the bus, between the bus and payload, & between the constellation and the ground system. Includes all labor, subcontracts, materials and other direct costs.
1.2.8	Configuration Management	This element contains the effort to maintain configuration control of all drawings and documentation necessary to ensure the proper design, production and operations of the LISA system and includes all labor, subcontracts, materials and other direct costs.
1.2.9	Risk Management	This element contains the effort to develop and maintain a risk management plan and system and includes all labor, subcontracts, materials and other direct costs.
1.2.10	System Integration & Test (Support)	This element contains the effort to design, develop, and implement system level tests that reflect the interaction of observatory and constellation elements. It also includes development of an EMC/EMI plan and development/operations of system test beds after the conclusion of the technology development program. Includes all labor, subcontracts, materials and other direct costs.
1.3	Mission Assurance	This element summarizes the effort to perform mission assurance management, system safety, contamination control, IV&V, and quality assurance.

WBS	DESCRIPTION	WBS DICTIONARY
1.3.1	Mission Assurance Management	This element contains the effort to establish and maintain a flight assurance program and to develop and maintain the mission assurance guidelines and requirements. Includes all labor, subcontracts, materials and other direct costs.
1.3.2	System Safety	This element contains the effort to develop and maintain a safety program, and includes development of a Safety Data Package and a System Safety Plan. Includes all labor, subcontracts, materials and other direct costs.
1.3.3	Contamination Control	This element contains the effort to develop and maintain a contamination control program and includes all labor, subcontracts, materials and other direct costs.
1.3.4	IV&V	This element contains the effort to develop independent verification and validation processes for the onboard software and firmware and includes all labor, subcontracts, materials and other direct costs.
1.3.5	Quality Assurance	This element contains the effort to establish and maintain a project level quality assurance program and includes all labor, subcontracts, materials and other direct costs.
1.4	Science	This element summarizes the effort to maintain a mission science office, science operations, science data support, and education & public outreach.
1.4.1	Mission Science Office	This element summarizes the effort for the mission scientist, the science team, the Science Management Plan, Science Requirements, Astrophysics & Waveform of Sources, Data Analysis Methods and the Science Community Interface.
1.4.1.1	Mission Scientist	This element contains the effort for the mission scientist and includes all labor, subcontracts, materials and other direct costs.
1.4.1.2	Science Team	This element contains the effort to support the science team and includes all labor, subcontracts, materials and other direct costs.
1.4.1.3	Science Management Plan	This element contains the effort to determine a science management approach and develop and maintain a science management plan. Includes all labor, subcontracts, materials, and other direct costs.
1.4.1.4	Science Requirements	This element contains the effort to define and update the science requirements. Includes all labor, subcontracts, materials, and other direct costs.
1.4.1.5	Astrophysics & Waveform of Sources	This element contains the effort to develop the initial calculations of Astrophysics & Waveform of Sources, and provide updates throughout the life of the LISA mission. Includes all labor, subcontracts, materials, and other direct costs.
1.4.1.6	Data Analysis Methods	This element contains the effort to develop methods for analyzing LISA data in order to extract scientific results. Includes all labor, subcontracts, materials, and other direct costs.

WBS	DESCRIPTION	WBS DICTIONARY
1.4.1.7	Science Community Interface	This element contains the effort to develop and manage the participation of the astrophysics and fundamental physics communities in LISA science. Includes all labor, subcontracts, materials, and other direct costs.
1.4.2	Science Operations	This element summarizes the effort to provide operations support and scientific performance characterization.
1.4.2.1	Operations Support	This element contains the effort to receive and archive incoming science and scientific housekeeping data from the mission operations center, and calibrate, correct and format incoming data in units of strain for each polarization of the antenna. It also includes the recording of orbital elements describing detector location and orientation, verifying that incoming data meets the expectations of the science requirements, and ascertaining and advising on the need for adjustments to, or improvements in, the operational state of the LISA constellation. Includes all labor, subcontracts, materials, and other direct costs.
1.4.2.2	Scientific Performance Characterization	This element contains the effort to provide scientific performance characterization and includes all labor, subcontracts, materials and other direct costs.
1.4.3	Science Data Support	This element summarizes the effort to provide data analysis and data archiving & distribution and to support participating scientists.
1.4.3.1	Data Analysis	This element contains the effort to analyze strain data to extract astrophysical parameters of sources and catalog them with their astrophysical parameters. This includes preliminary analyses to enable other astrophysical observing systems to react to forecasts of extraordinary events, and includes all labor, subcontracts, materials and other direct costs.
1.4.3.2	Data Archiving & Distribution	This element contains the effort to archive and distribute intermediate data products and the catalog of gravitational wave sources to develop and maintain the science data management plan and a data archiving & distribution system. Includes all labor, subcontracts, materials and other direct costs.
1.4.3.3	Participating Scientists	This element contains the effort to manage direct participation (e.g., guest investigator program) by members of astrophysics and fundamental physics communities in the analysis and interpretation of LISA science data and includes all labor, subcontracts, materials and other direct costs.
1.4.4	Education & Public Outreach	This element contains the effort to provide education and public outreach. These costs may include public affairs announcements, video and audio type publications, public relation seminars, K-12 classroom presentations, college level seminars and involvement with public television programming. Includes all labor, subcontracts, materials and other direct costs.
1.5	Payload System	This element summarizes the effort to design, fabricate, assemble, integrate and test the payload system.

WBS	DESCRIPTION	WBS DICTIONARY
1.5.1	Payload System Engineering / Management	This element contains the effort to manage the engineering resources necessary to complete the payload systems engineering functions. These costs include personnel management, coordination between subsystems and carrying out of operational activities. Includes all labor, subcontracts, materials, and other direct costs.
1.5.2	Optical Assembly	This element summarizes the effort to design, fabricate, and assemble the optical assembly.
1.5.2.1	Optical Bench Assembly	This element contains the effort to design, fabricate, and assemble the optical bench assembly components such as the optical bench, dummy GRS, fiber positioner, optics, laser stabilization cavity, premounting for CCD & diodes and the integration fixture. Includes all labor, subcontracts, materials, and other direct costs.
1.5.2.2	CCD Sensor	This element contains the effort to design, fabricate, assemble, integrate and test the CCD sensor. It includes the CCDs and the CCD electronics. Also included are all labor, subcontracts, materials, and other direct costs.
1.5.2.3	Phase Detector	This element contains the effort to design, fabricate, assemble, and integrate and test the phase detector. It includes the diodes & hardware, pre-amplifiers, phase readout electronics and ultra stable oscillator. Also included are all labor, subcontracts, materials, and other direct costs.
1.5.2.4	Laser	This element contains the effort to design, fabricate, assemble, and integrate and test the laser. It includes the fiber delivery system, laser stabilization electronics, laser, phase modulator and phase modulator electronics & data interface. Also, included are all labor, subcontracts, materials, and other direct costs.
1.5.2.5	Optical Assembly Structure	This element contains the effort to design, fabricate, assemble, and test the optical assembly structure. Includes all labor, subcontracts, materials, and other direct costs.
1.5.2.6	Star Tracker	This element contains the effort to design, fabricate, assemble, and test the start tracker. Includes all labor, subcontracts, materials, and other direct costs.
1.5.2.7	Telescope	This element contains the effort to design, fabricate, assemble, integrate and test the telescope. It includes the primary optics, secondary optics and the telescope structure. Also included are all labor, subcontracts, materials, and other direct costs.
1.5.2.8	Optical Assembly Actuator	This element contains the effort to design, fabricate, assemble, integrate and test the optical assembly actuator. It includes the actuator mechanism and actuator electronics. Also included are all labor, subcontracts, materials, and other direct costs.
1.5.3	Gravitational Reference Sensor (GRS)	This element summarizes the effort to design, fabricate, assemble, and integrate and test the gravitational reference sensor (GRS).

WBS	DESCRIPTION	WBS DICTIONARY
1.5.3.1	Flight Test Mass	This element contains the effort to design, fabricate, assemble, and test the flight test mass. Includes all labor, subcontracts, materials, and other direct costs.
1.5.3.2	Caging	This element contains the effort to design, fabricate, assemble, and test the caging. Includes all labor, subcontracts, materials, and other direct costs.
1.5.3.3	Housing & Electronics (Lightweight Test Mass)	This element contains the effort to design, fabricate, assemble, and integrate and test the housing & electronics (lightweight test mass). It includes the vacuum & structure, electrode housing, lightweight test mass & special caging, front end electronics and control electronics. It also includes all labor, subcontracts, materials, and other direct costs.
1.5.3.4	Charge Management Unit	This element contains the effort to design, fabricate, assemble, and integrate and test the charge management unit. It includes the UV lamps, UV switch and UV control. Also included are all labor, subcontracts, materials, and other direct costs.
1.5.3.5	GRS I&T	This element summarizes the effort to integrate and test the gravitational reference sensor.
1.5.3.5.1	GRS EM I&T	This element contains the effort to integrate and test the gravitational reference sensor engineering model. Includes all labor, subcontracts, materials, and other direct costs.
1.5.3.5.2	GRS FM1 I&T	This element contains the effort to integrate and test the gravitational reference sensor flight model 1. Includes all labor, subcontracts, materials, and other direct costs.
1.5.3.5.3	GRS FM2 I&T	This element contains the effort to integrate and test the gravitational reference sensor flight model 2. Includes all labor, subcontracts, materials, and other direct costs.
1.5.3.5.4	GRS FM3 I&T	This element contains the effort to integrate and test the gravitational reference sensor flight model 3. Includes all labor, subcontracts, materials, and other direct costs.
1.5.4	Y-Tube Assembly	This element summarizes the effort to design, fabricate, and assemble, the Y-tube assembly components.
1.5.4.1	Aft Fiber	This element contains the effort to design, fabricate, assemble, and test the aft fiber. Includes all labor, subcontracts, materials, and other direct costs.
1.5.4.2	Ultra Stable Structure	This element contains the effort to design, build, assemble, and integrate and test the ultra stable structure. It includes the radiator, Y-tube structure and aft bulkhead. Also included are all labor, subcontracts, materials, and other direct costs.
1.5.5	Payload Processor / Controller	This element contains the effort to design, fabricate, assemble, integrate and test the payload processor/ controller. Includes all labor, subcontracts, materials, and other direct costs.
1.5.6	Payload Integration & Test	This element summarizes the effort to integrate and test the payload.

WBS	DESCRIPTION	WBS DICTIONARY
1.5.6.1	Optical Bench Assembly Initial Integration (Dummy GRS)	This element summarizes the effort to set up the integration facility & GSE and to perform the Optical Bench Assembly Initial Integration & Test of the engineering and flight model Payloads.
1.5.6.1.1	Set-Up Integration Facility & GSE	This element contains the effort to set up the integration facility & GSE and includes all labor, subcontracts, materials, and other direct costs.
1.5.6.1.2	Optical Bench Assembly EM I&T	This element contains the effort to integrate & test and pack, ship and deliver to ESA the Optical Bench Assembly for the Payload Engineering Model. Includes all labor, subcontracts, materials, and other direct costs.
1.5.6.1.3	Optical Bench Assembly FM1 I&T	This element contains the effort to integrate & test and pack, ship and deliver to ESA the Optical Bench Assembly for the Payload Flight Model 1. Includes all labor, subcontracts, materials, and other direct costs.
1.5.6.1.4	Optical Bench Assembly FM2 I&T	This element contains the effort to integrate & test and pack, ship and deliver to ESA the Optical Bench Assembly for the Payload Flight Model 2. Includes all labor, subcontracts, materials, and other direct costs.
1.5.6.1.5	Optical Bench Assembly FM3 I&T	This element contains the effort to integrate & test and pack, ship and deliver to ESA the Optical Bench Assembly for the Payload Flight Model 3. Includes all labor, subcontracts, materials, and other direct costs.
1.5.6.2	Electro-Optical Integration (Dummy GRS)	This element summarizes the effort to perform the Electro-Optical Integration with the Dummy GRS of the engineering and flight model Payloads.
1.5.6.2.1	Electro-Optical EM I&T	This element contains the effort to integrate & test and pack, ship and deliver to NASA the electro-optical engineering model. Includes all labor, subcontracts, materials, and other direct costs.
1.5.6.2.2	Electro-Optical FM1 I&T	This element contains the effort to integrate & test and pack, ship and deliver to NASA the electro-optical flight model 1. Includes all labor, subcontracts, materials, and other direct costs.
1.5.6.2.3	Electro-Optical FM2 I&T	This element contains the effort to integrate & test and pack, ship and deliver to NASA the electro-optical flight model 2. Includes all labor, subcontracts, materials, and other direct costs.
1.5.6.2.4	Electro-Optical FM3 I&T	This element contains the effort to integrate & test and pack, ship and deliver to NASA the electro-optical flight model 3. Includes all labor, subcontracts, materials, and other direct costs.
1.5.6.3	Optical Assembly Integration (Dummy GRS)	This element summarizes the effort to perform the Optical Assembly Integration with the Dummy GRS of the engineering and flight model Payloads.
1.5.6.3.1	Optical Assembly EM I&T	This element contains the effort to initially integrate and test the optical assembly engineering model. Includes all labor, subcontracts, materials, and other direct costs.

WBS	DESCRIPTION	WBS DICTIONARY
1.5.6.3.2	Optical Assembly FM1 I&T	This element contains the effort to initially integrate and test the optical assembly flight model 1. Includes all labor, subcontracts, materials, and other direct costs.
1.5.6.3.3	Optical Assembly FM2 I&T	This element contains the effort to initially integrate and test the optical assembly flight model 2. Includes all labor, subcontracts, materials, and other direct costs.
1.5.6.3.4	Optical Assembly FM3 I&T	This element contains the effort to initially integrate and test the optical assembly flight model 3. Includes all labor, subcontracts, materials, and other direct costs.
1.5.6.4	Optical Assembly Final Integration (Flight GRS)	This element summarizes the effort to perform the Optical Assembly Final Integration with the Flight GRS of the engineering and flight model Payloads.
1.5.6.4.1	Optical Assembly EM Final I&T	This element contains the effort to perform final integration and test of the optical assembly engineering model. Includes all labor, subcontracts, materials, and other direct costs.
1.5.6.4.2	Optical Assembly FM1 Final I&T	This element contains the effort to perform final integration and test of the optical assembly flight model 1. Includes all labor, subcontracts, materials, and other direct costs.
1.5.6.4.3	Optical Assembly FM2 Final I&T	This element contains the effort to perform final integration and test of the optical assembly flight model 2. Includes all labor, subcontracts, materials, and other direct costs.
1.5.6.4.4	Optical Assembly FM3 Final I&T	This element contains the effort to perform final integration and test of the optical assembly flight model 3. Includes all labor, subcontracts, materials, and other direct costs.
1.5.6.5	Y-Tube Integration	This element summarizes the effort to perform the Y-Tube Integration of the engineering and flight model Payloads.
1.5.6.5.1	Y-Tube EM I&T	This element contains the effort to integrate and test and pack, ship and delivery to NASA the Y-tube engineering model. Includes all labor, subcontracts, materials, and other direct costs.
1.5.6.5.2	Y-Tube FM1 I&T	This element contains the effort to integrate and test and pack, ship and delivery to NASA the Y-tube flight model 1. Includes all labor, subcontracts, materials, and other direct costs.
1.5.6.5.3	Y-Tube FM2 I&T	This element contains the effort to integrate and test and pack, ship and delivery to NASA the Y-tube flight model 2. Includes all labor, subcontracts, materials, and other direct costs.
1.5.6.5.4	Y-Tube FM3 I&T	This element contains the effort to integrate and test and pack, ship and delivery to NASA the Y-tube flight model 3. Includes all labor, subcontracts, materials, and other direct costs.
1.6	Flight System	This element summarizes the effort to acquire, design, fabricate, assemble, integrate and test the ESA spacecraft and integrate and test the observatory.
1.6.1	Spacecraft (ESA)	This element summarizes the effort to acquire, design, fabricate, assemble, integrate and test the ESA spacecraft.

WBS	DESCRIPTION	WBS DICTIONARY
1.6.1.1	ESA Acquisition Cycle	This element represents the ESA acquisition process / cycle effort and is included for reference.
1.6.1.2	Spacecraft System Engineering / Management	This element contains the effort to manage the engineering resources necessary to complete the spacecraft systems engineering functions. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.3	Spacecraft Bus	This element summarizes the effort to design, fabricate, and assemble the structures/ mechanisms, power, C&DH, telecom, ACS, thermal and cabling subsystems, and integrate and test the spacecraft bus.
1.6.1.3.1	Structures / Mechanisms Subsystem	This element summarizes the effort to design, fabricate, assemble, and test the structure & mechanism and secondary structure subsystems.
1.6.1.3.1.1	Structure & Mechanism	This element contains the effort to design, fabricate, assemble and test the structure & mechanism subsystem and includes all labor, subcontracts, materials, and other direct costs.
1.6.1.3.1.2	Secondary Structure	This element contains the effort to design, fabricate, assemble, and test the secondary structure subsystem and includes all labor, subcontracts, materials, and other direct costs.
1.6.1.3.2	Power Subsystem	This element summarizes the effort to design, fabricate, assemble, and test the solar array, battery, and power control distribution unit.
1.6.1.3.2.1	Solar Array	This element contains the effort to design, fabricate, assemble, and test the solar array and includes all labor, subcontracts, materials, and other direct costs.
1.6.1.3.2.2	Battery	This element contains the effort to design, fabricate, assemble, and test the battery and includes all labor, subcontracts, materials, and other direct costs.
1.6.1.3.2.3	Power Control Distribution Unit	This element contains the effort to design, fabricate, assemble, and test the power control distribution unit and includes all labor, subcontracts, materials, and other direct costs.
1.6.1.3.3	Command & Data Handling Subsystem	This element contains the effort to design, fabricate, assemble, and test the command and data handling subsystem and includes all labor, subcontracts, materials, and other direct costs.
1.6.1.3.4	Telecom Subsystem	This element summarizes the effort to design, fabricate, assemble, and test the transponders, RFDU, SSPA, TWT, high gain antenna, low gain antenna, and telecom cabling.
1.6.1.3.4.1	Transponders	This element contains the effort to design, fabricate, assemble, and test the transponders and includes all labor, subcontracts, materials, and other direct costs.
1.6.1.3.4.2	RFDU	This element contains the effort to design, fabricate, assemble, and test the RFDU and includes all labor, subcontracts, materials, and other direct costs.
1.6.1.3.4.3	SSPA	This element contains the effort to design, fabricate, assemble, and test the SSPA and includes all labor, subcontracts, materials, and other direct costs.

WBS	DESCRIPTION	WBS DICTIONARY
1.6.1.3.4.4	TWT	This element contains the effort to design, fabricate, assemble, and test the TWT and includes all labor, subcontracts, materials, and other direct costs.
1.6.1.3.4.5	High Gain Antenna	This element contains the effort to design, fabricate, assemble, and test the high gain antenna and includes all labor, subcontracts, materials, and other direct costs.
1.6.1.3.4.6	Low Gain Antenna	This element contains the effort to design, fabricate, assemble, and test the low gain antenna and includes all labor, subcontracts, materials, and other direct costs.
1.6.1.3.4.7	Telecom Cabling	This element contains the effort to design, fabricate, assemble, and test the telecom cabling and includes all labor, subcontracts, materials, and other direct costs.
1.6.1.3.5	Attitude Control Subsystem	This element summarizes the effort to design, fabricate, assemble, and test the star camera assembly optical head & electronics, sun sensor, aft anomaly detector, magnetometer, FEEP assembly, and high gain antenna assembly drive & electronics.
1.6.1.3.5.1	Star Camera Assembly Optical Head & Electronics	This element contains the effort to design, fabricate, assemble, and test the star camera assembly optical head & electronics. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.3.5.2	Sun Sensor	This element contains the effort to design, fabricate, assemble, and test the sun sensor. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.3.5.3	Aft Anomaly Detector	This element contains the effort to design, fabricate, assemble, and test the aft anomaly detector. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.3.5.4	Magnetometer	This element contains the effort to design, fabricate, assemble, and test the magnetometer. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.3.5.5	FEEP Assembly	This element contains the effort to design, fabricate, assemble, and test the FEEP assembly. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.3.5.6	High Gain Antenna Assembly Drive & Electronics	This element contains the effort to design, fabricate, assemble, and test the high gain antenna assembly drive & electronics. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.3.6	Thermal Subsystem	This element contains the effort to design, procure, and build the thermal subsystem. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.3.7	Cabling	This element contains the effort to design, fabricate, assemble, and test the interconnecting cabling for the spacecraft bus. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.3.8	Spacecraft Bus Integration & Test	This element summarizes the effort to integrate and test the spacecraft bus engineering and flight models.

WBS	DESCRIPTION	WBS DICTIONARY
1.6.1.3.8.1	Spacecraft Bus EM Integration & Test	This element contains the effort to integrate and test the spacecraft bus engineering model. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.3.8.2	Spacecraft Bus FM1 Integration & Test	This element contains the effort to integrate and test the spacecraft bus flight model 1. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.3.8.3	Spacecraft Bus FM2 Integration & Test	This element contains the effort to integrate and test the spacecraft bus flight model 2. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.3.8.4	Spacecraft Bus FM3Integtration & Test	This element contains the effort to integrate and test the spacecraft bus flight model 3. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.4	Propulsion Module	This element summarizes the effort to design, fabricate, assemble, and test the structure and chemical propulsion subsystems and integrate and test the propulsion module.
1.6.1.4.1	Structure	This element summarizes the effort to design, fabricate, assemble, and test the primary and secondary structures.
1.6.1.4.1.1	Primary Structure	This element contains the effort to design, fabricate, assemble, and test the primary structure. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.4.1.2	Secondary Structure	This element contains the effort to design, fabricate, assemble, and test the secondary structure. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.4.2	Chemical Propulsion	This element summarizes the effort to design, fabricate, assemble, and test the fuel and pressurant tanks, filters, valves & regulators, RCS thruster assembly, and heaters.
1.6.1.4.2.1	Fuel Tanks	This element contains the effort to design, fabricate, assemble, and test the fuel tanks. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.4.2.2	Pressurant Tanks	This element contains the effort to design, fabricate, assemble, and test the pressurant tanks. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.4.2.3	Filters, Valves & Regulators	This element contains the effort to design, fabricate, assemble, and test the filters, valves & regulators. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.4.2.4	RCS Thruster Assembly	This element contains the effort to design, fabricate, assemble, and test the RCS thruster assembly. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.4.2.5	Heaters	This element contains the effort to design, fabricate, assemble, and test the heaters. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.4.3	Integration & Test	This element summarizes the effort to integrate and test the engineering and flight model propulsion modules.

WBS	DESCRIPTION	WBS DICTIONARY
1.6.1.4.3.1	Propulsion Module EM I&T	This element contains the effort to integrate and test the engineering model propulsion module. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.4.3.2	Propulsion Module FM1 I&T	This element contains the effort to integrate and test the flight model 1 propulsion module. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.4.3.3	Propulsion Module FM2 I&T	This element contains the effort to integrate and test the flight model 2 propulsion module. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.4.3.4	Propulsion Module FM3 I&T	This element contains the effort to integrate and test the flight model 3 propulsion module. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.5	Spacecraft I&T (Bus & Propulsion Module)	This element summarizes the effort to integrate and test the engineering model and flight spacecraft.
1.6.1.5.1	Spacecraft EM I&T	This element contains the effort to integrate and test and pack, ship and deliver the engineering model spacecraft to observatory I&T contractor. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.5.2	Spacecraft Flight 1 I&T	This element contains the effort to integrate and test and pack, ship and deliver the flight model 1 spacecraft to observatory I&T contractor. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.5.3	Spacecraft Flight 2 I&T	This element contains the effort to integrate and test and pack, ship and deliver the flight model 2 spacecraft to observatory I&T contractor. Includes all labor, subcontracts, materials, and other direct costs.
1.6.1.5.4	Spacecraft Flight 3 I&T	This element contains the effort to integrate and test and pack, ship and deliver the flight model 3 spacecraft to observatory I&T contractor. Includes all labor, subcontracts, materials, and other direct costs.
1.6.2	Observatory Integration & Test	This element summarizes the effort to perform observatory testing on engineering and flight models.
1.6.2.1	Observatory EM Integration & Test	This element contains the effort to perform observatory test on the engineering model. Includes all labor, subcontracts, materials, and other direct costs.
1.6.2.2	Observatory Flight 1 Integration & Test	This element contains the effort to perform observatory test on flight 1 and to turnover flight 1 to constellation test. Includes all labor, subcontracts, materials, and other direct costs.
1.6.2.3	Observatory Flight 2 Integration & Test	This element contains the effort to perform observatory test on flight 2 and to turnover flight 2 to constellation test. Includes all labor, subcontracts, materials, and other direct costs.
1.6.2.4	Observatory Flight 3 Integration & Test	This element contains the effort to perform observatory test on flight 3 and to turnover flight 3 to constellation test. Includes all labor, subcontracts, materials, and other direct costs.

WBS	DESCRIPTION	WBS DICTIONARY
1.6.3	Constellation Testing	This element contains the effort to set up the constellation test facility and test the three LISA observatories in the constellation configuration. Includes all labor, subcontracts, materials, and other direct costs.
1.6.4	Final Integration (Pre-Launch)	This element contains the effort to install the payload attach fitting, perform pre-ship functional testing for the launch campaign, and pack, ship & deliver to the launch site. Includes all labor, subcontracts, materials, and other direct costs.
1.6.5	Launch Campaign	This element summarizes the effort to perform launch site operations and conduct the LISA launch.
1.6.5.1	Constellation Launch Site Operations	This element contains the effort to perform constellation launch site operations. Includes all labor, subcontracts, materials, and other direct costs.
1.6.5.2	Constellation / Launch Vehicle Operations	This element contains the effort to perform constellation/ launch vehicle operations. Includes all labor, subcontracts, materials, and other direct costs.
1.7	Mission Operations System	This element summarizes the effort to develop and maintain a support team to provide the following functions: commanding and monitoring the spacecraft subsystems, mission planning activities, mission requirements analysis, flight operations and contingency procedures development/ maintenance, spacecraft trend analysis and anomaly resolution.
1.7.1	Mission Ops Management	This element contains the effort to manage the design, development, integration, and testing of the mission operations system. Includes all labor, subcontracts, materials, and other direct costs.
1.7.2	Mission Ops System Engineering	This element contains the effort to perform system engineering of the mission operations system including management, planning, requirements definition, design, development, integration, evaluation, and testing of the hardware and software components of the mission ops system for the mission prior to launch. Includes all labor, subcontracts, materials, and other direct costs.
1.7.3	GDS Supporting Elements (Non- MOS GDS efforts)	This element contains the effort for non-GDS-specific supporting elements such as developing DSN configurations for LISA and configuring and extending the navigation tools for the LISA specific problems Includes all labor, subcontracts, materials, and other direct costs.
1.7.4	Operations (Pre-Launch)	This element contains the effort for mission operations planning and reviews including the Mission Operations Review and the Operations Readiness Review. Includes all labor, subcontracts, materials, and other direct costs.
1.7.5	Missions Operations System V&V / Readiness Testing	This element contains the effort to perform mission operations systems validation & verification, certification and readiness testing such as End-To-End Tests, mission simulations, compatibility checks and data flows. Includes all labor, subcontracts, materials, and other direct costs.

WBS	DESCRIPTION	WBS DICTIONARY
1.8	Launch System	This element summarizes the effort to integrate with the launch vehicle and includes safety documentation, launch site procedure development, range safety support for hazardous procedure reviews, launch vehicle-to-observatory testing, launch rehearsals and launch site-to-ground control center interface testing.
1.8.1	Spacecraft-To- Launch Vehicle Interface Definition, Verification & Coordination	This element contains the effort for developing, integrating, and verifying spacecraft-to-launch vehicle interface control document. Also includes support to develop the range's operations requirements document, the missile system pre-launch safety plan (MSPSP), the launch service's mission constraints and launch countdown manuals. Includes all labor, subcontracts, materials, and other direct costs.
1.8.2	Launch Vehicle Build, Storage and Pre-Ship Operations	This element contains the effort to build, store, and perform pre-ship ops on the launch vehicle. Includes all labor, subcontracts, materials, and other direct costs.
1.8.3	Launch Services / Launch Site Operations	This element contains the effort for supporting launch services and launch site operations. Includes all labor, subcontracts, materials, and other direct costs.
1.9	Technology	This element summarizes the effort for the technology management, disturbance reduction system, interferometry measurement system, and system verification.
1.9.1	Technology Management	This element contains the effort to manage the LISA technology development effort including responding to TRIP board findings, and supporting annual independent and technology readiness reviews. Includes all labor, subcontracts, materials, and other direct costs.
1.9.2	Disturbance Reduction System	This element summarizes the effort to perform trade studies and prototype design and testing on the gravitational reference sensor, perform trades and prototype build and testing on the thrusters, and develop algorithms and perform simulator testing on the DRS controls.
1.9.2.1	Gravitational Reference Sensor	This element summarizes the effort to perform trade studies and write reports on the housing, vacuum, caging, proof mass, proof mass control, sensing & forcing, charge control, and I&T mechanisms subsystems. This also summarizes the effort to design, fabricate, assemble, integrate and test the prototype GRS.
1.9.2.1.1	Housing Subsystem	This element contains the effort to perform trade studies and write reports for the housing subsystem. Includes all labor, subcontracts, materials, and other direct costs.
1.9.2.1.2	Vacuum Subsystem	This element contains the effort to perform trade studies and write reports for the vacuum subsystem. Includes all labor, subcontracts, materials, and other direct costs.
1.9.2.1.3	Caging Subsystem	This element contains the effort to perform trade studies and write reports for the caging subsystem. Includes all labor, subcontracts, materials, and other direct costs.

WBS	DESCRIPTION	WBS DICTIONARY
1.9.2.1.4	Proof Mass Subsystem	This element contains the effort to perform trade studies and write reports for the proof mass subsystem. Includes all labor, subcontracts, materials, and other direct costs.
1.9.2.1.5	Proof Mass Control Subsystem	This element contains the effort to perform trade studies and write reports for the proof mass control subsystem. Includes all labor, subcontracts, materials, and other direct costs.
1.9.2.1.6	Sensing & Forcing Subsystem	This element contains the effort to perform trade studies and write reports for the sensing & focusing subsystem. Includes all labor, subcontracts, materials, and other direct costs.
1.9.2.1.7	Charge Control Subsystem	This element contains the effort to perform trade studies and write reports for the charge control subsystem. Includes all labor, subcontracts, materials, and other direct costs.
1.9.2.1.8	I&T Mechanisms	This element contains the effort to perform trade studies and write reports for the I&T mechanisms. Includes all labor, subcontracts, materials, and other direct costs.
1.9.2.1.9	Prototype GRS (Lab Verified)	This element contains the effort to design, fabricate, assemble, integrate, and test the prototype GRS in a laboratory environment. Includes all labor, subcontracts, materials, and other direct costs.
1.9.2.2	Thrusters	This element summarizes the effort to develop and test a prototype emitter, perform trade studies and build and test a prototype neutralizer, develop and test prototype thruster electronics, and integrate, assemble, and test a prototype thruster.
1.9.2.2.1	Emitter	This element contains the effort to develop and test a prototype emitter and includes life testing. Includes all labor, subcontracts, materials, and other direct costs.
1.9.2.2.2	Neutralizer	This element contains the effort to perform trade studies, build and test a neutralizer prototype, and integrate with the emitter prototype. This effort also includes writing a test report. Includes all labor, subcontracts, materials, and other direct costs.
1.9.2.2.3	Thruster Electronics	This element contains the effort to develop and test a prototype thruster electronics which includes writing a test report. Include all labor, subcontracts, materials, and other direct costs.
1.9.2.2.4	Prototype Thruster	This element contains the effort to integrate, assemble, and test a prototype thruster which includes writing a test report. Includes all labor, subcontracts, materials, and other direct costs.
1.9.2.3	DRS Controls	This element summarizes the effort to develop control algorithms and develop and test controls simulator hardware & source code.
1.9.2.3.1	Algorithms	This element contains the effort to develop control algorithms and deliver them testing with the simulator. Includes all labor, subcontracts, materials, and other direct costs.
1.9.2.3.2	Simulator Test	This element contains the effort to develop and test controls simulator hardware and source code which includes writing a test report. Includes all labor, subcontracts, materials, and other direct costs.

WBS	DESCRIPTION	WBS DICTIONARY
1.9.3	Interferometry Measurement System	This element summarizes the effort to perform trade studies and develop breadboard and prototype lasers, develop laser/ clock noise cancellation algorithms, develop stabilization and phase meter breadboards, test the ultra stable structures, develop and build an IMS tester, and design, build, integrate, assemble and test a prototype IMS.
1.9.3.1	Laser(s)	This element contains the effort to perform trade studies and develop breadboard and prototype lasers. Includes all labor, subcontracts, materials, and other direct costs.
1.9.3.2	Laser/Clock Noise Cancellation	This element contains the effort to develop laser/ clock noise cancellation algorithms that include writing a test report. Includes all labor, subcontracts, materials, and other direct costs.
1.9.3.3	Stabilization	This element contains the effort to develop stabilization breadboards. Includes all labor, subcontracts, materials, and other direct costs.
1.9.3.4	Phase Meter	This element contains the effort to develop phase meter breadboards. Includes all labor, subcontracts, materials, and other direct costs.
1.9.3.5	Ultra Stable Structures	This element contains the effort to test the ultra stable structures that include writing a test report. Includes all labor, subcontracts, materials, and other direct costs.
1.9.3.6	IMS Tester	This element contains the effort to develop and build an IMS tester. Includes all labor, subcontracts, materials, and other direct costs.
1.9.3.7	Prototype IMS	This element contains the effort to design, build, integrate, assemble and test a prototype IMS which includes writing a test report. Includes all labor, subcontracts, materials, and other direct costs.
1.9.4	System Verification	This element summarizes the effort to perform integrated modeling and select and develop test bed approaches.
1.9.4.1	Integrated Modeling	This element summarizes the effort to define and develop the modeling environment, and to establish quasi-static, dynamic, phase propagation, and end-to-end models. This effort includes conducting trade studies.
1.9.4.1.1	Modeling Environment	This element contains the effort to define the requirements, conduct trade studies, and develop the modeling environment. Includes all labor, subcontracts, materials, and other direct costs.
1.9.4.1.2	Quasi-Static Models	This element contains the effort to establish a baseline, validate requirements, conduct trade studies, and integrate the quasi-static model. This effort includes writing a final trade study report. Includes all labor, subcontracts, materials, and other direct costs.
1.9.4.1.3	Dynamic Models	This element contains the effort to establish a baseline, validate requirements, conduct trade studies, and integrate the dynamic model. This effort includes writing a final trade study report. Includes all labor, subcontracts, materials, and other direct costs.

WBS	DESCRIPTION	WBS DICTIONARY
1.9.4.1.4	Phase Propagation Models	This element contains the effort to establish a baseline, validate requirements, conduct trade studies, and integrate the phase propagation models. This effort includes writing a final trade study report. Includes all labor, subcontracts, materials, and other direct costs.
1.9.4.1.5	End-To-End Models	This element contains the effort to establish a baseline, validate requirements, conduct trade studies, and integrate the end-to-end model. This effort includes writing a final trade study report. Includes all labor, subcontracts, materials, and other direct costs.
1.9.4.2	Test Bed Technology	This element summarizes the effort to develop test bed approaches. Includes the effort to develop LISA test beds.
1.9.4.2.1	Phase 1 - Approaches	This element contains the effort to define requirements, conduct trade studies, and select test bed approaches. Includes all labor, subcontracts, materials, and other direct costs.
1.9.4.2.2	Phase 2 - Development	This element contains the effort to develop test beds and demonstrate concepts. Includes all labor, subcontracts, materials, and other direct costs.
1.10	Mission Software	This element summarizes the effort to manage and design, develop, and test flight software, mission operations software and science operations software.
1.10.1	Software Management	This element contains the effort to develop and maintain a software management plan. Includes all labor, subcontracts, materials, and other direct costs.
1.10.2	Software Requirements Document	This element contains the effort to develop and maintain a software requirements document. Includes all labor, subcontracts, materials, and other direct costs.
1.10.3	Software Test Plan	This element contains the effort to develop and maintain a software test plan. Includes all labor, subcontracts, materials, and other direct costs.
1.10.4	Spacecraft Flight Software	This element contains the effort to define and analyze the requirements, review the software specification, and design, develop, and test the spacecraft flight software. Includes all labor, subcontracts, materials, and other direct costs.
1.10.5	Ground Data System (Mission Ops System)	This element contains the effort to write the requirements and design, build, and test the ground data system. Includes all labor, subcontracts, materials, and other direct costs.
1.10.6	Science Data System Software	This element contains the effort to write the requirements and design, build, and test the science data system software. Includes all labor, subcontracts, materials, and other direct costs.
1.10.7	IV&V	This element contains the effort to plan and coordinate the verification of the mission software. Includes all labor, subcontracts, materials, and other direct costs.

H.2 Supporting, Detailed Schedules

The LISA team has developed a comprehensive, integrated schedule for the LISA mission that:

- Identifies all work scope and is correlated to the preliminary WBS
- Incorporates critical programmatic interdependencies among the activities
- Contains engineering and planning estimates for activity durations
- Includes schedule reserves at critical points in the work flow

The integrated schedule consists of a single Microsoft Project 2000 file of over 1400 nodes that is organized by WBS. It also includes key elements such as the technology development efforts for the LISA Test Package (LTP) and ST-7 Disturbance Reduction System (DRS) that are part of ESA's SMART-2 mission.

While preliminary, the LISA integrated schedule represents an excellent starting point for further detailed planning, budgeting, and risk assessment during the Pre-Formulation and Formulation Stages.

The terms "schedule reserve," "total slack," "free slack," and "critical path" are defined below. The LISA project uses generally accepted definitions of these concepts.

Critical Path	The longest sequential path through the activities in the logic network, from beginning to end, that define the earliest LISA can launch.
	- Path with the longest overall duration
	- Path with the least amount of total slack
Slack	The difference between the "early" and late "dates" of activities calculated by running a time analysis of the logic network
	 Free Slack: the amount of duration an activity can be delayed before it impacts the early start date of an activity that succeeds it.
	 Total Slack: the amount of time an activity can be delayed from its early finish date without delaying the planned completion or end date of the project. Total slack can be positive, zero or negative
Schedule Reserve	A pre-planned amount of schedule duration incorporated into the logic network at critical points and/or prior to the completion point of the project.

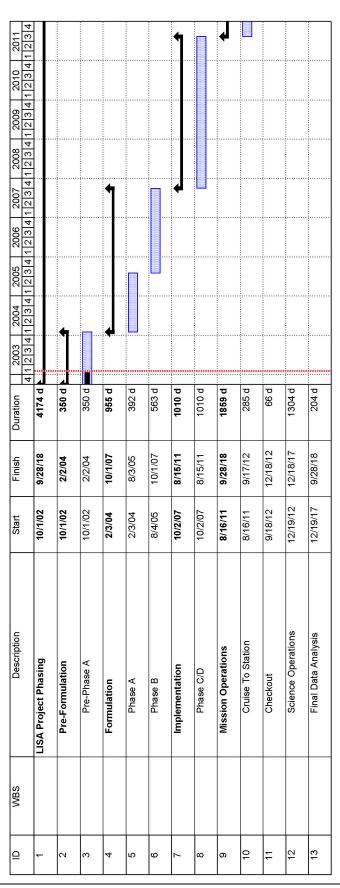
The following detailed schedules are produced directly from the MS Project file that comprises the LISA integrated schedule. They represent the total project effort at the lowest level of the WBS and indicate the planned duration and early start and early finish dates for each activity. This appendix is divided into the following categories:

- LISA Project Phasing
- NASA Headquarters Milestones
- Major Mission Reviews
- LISA System Engineering & Integration Contractor Acquisition Cycle
- LISA/ST-7 Disturbance Reduction System (SM2)
- LISA Test Package (SM2)
- Key Formulation Stage Products

- WBS 1.1 Project Management
- WBS 1.2 System Engineering
- WBS 1.3 Mission Assurance
- WBS 1.4 Science
- WBS 1.5 Payload System
- WBS 1.6 Flight System
- WBS 1.7 Mission Operations System
- WBS 1.8 Launch System
- WBS 1.9 Technology
- WBS 1.10 Mission Software

Finally, the MS Project file from which the schedules were produced is contained on the CD that accompanies this TRIP Report.

LISA PROJECT PHASING



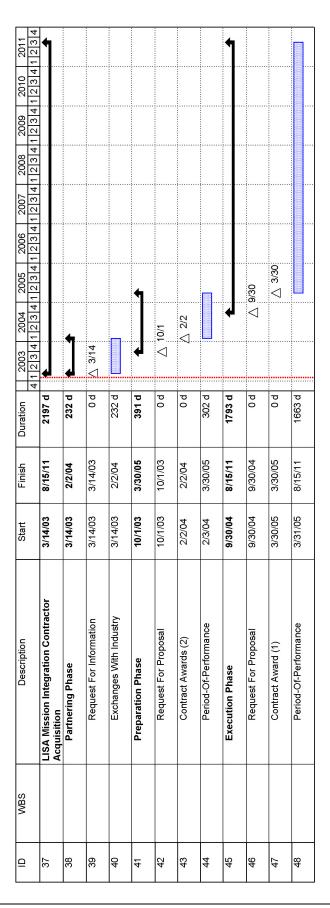
NASA HEADQUARTERS MILESTONES

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WBS Description Start Finish Duration NASA HQ Milestones 2/2/04 1/1/08 1020 d Formulation Authorization Document Signed / ATP Phase A Signed / ATP Phase A Confirmation Review & B/3/05 8/3/05 8/3/05 0 d NASA/ESA Memorandum Of Understanding (Preliminary) 4/12/07 4/12/07 0 d Non-Advocate Review 4/12/07 4/12/07 0 d NASA/ESA Memorandum Of Understanding (Final) 1/1/08 1/1/08 0 d Program Commitment Agreement Approval / ATP Phase C/D 10/1/07 0 d 0 d	ription Start Finish Duration rization Document 2/2/04 1/1/08 1020 d se A 1/2/04 2/2/04 0 d se A Sment / ATP Phase B senent / ATP Phase B eliminary) 8/3/05 8/3/05 0 d eliminary) 4/12/07 4/12/07 0 d riew 4/12/07 4/12/07 0 d randum Of 1/1/08 1/1/08 0 d rent Agreement 10/1/07 10/1/07 0 d ase C/D 0 d 0 d 0 d	2003	1234	•	7					
WBS Description Start Finish NASA HQ Milestones 2/2/04 1/1/08 Formulation Authorization Document Signed / ATP Phase A 2/2/04 2/2/04 Signed / ATP Phase A 8/3/05 8/3/05 Confirmation Review & Confirmation Assessment / ATP Phase B 8/3/05 8/3/05 NASA/ESA Memorandum Of Understanding (Preliminary) 4/12/07 4/12/07 Non-Advocate Review NASA/ESA Memorandum Of Understanding (Final) 1/1/08 1/1/08 Understanding (Final) Program Commitment Agreement Non-Advocate Review Understanding (Final) 1/1/08 1/1/08	ription Start Finish 2/2/04 1/1/08 rization Document 2/2/04 2/2/04 se A	F	4		_	_	_	_	_	_
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WBS Description Start NASA HQ Milestones 2/2/04 Formulation Authorization Document 2/2/04 Signed / ATP Phase A 8/3/05 Confirmation Review & Confirmation Review & Confirmation Assessment / ATP Phase B 8/3/05 NASA/ESA Memorandum Of Understanding (Preliminary) 4/12/07 Non-Advocate Review 4/12/07 Understanding (Final) 1/1/08 Understanding (Final) 1/1/08 Program Commitment Agreement 10/1/07 Approval / ATP Phase C/D 10/1/07	ription Start 2/2/04 2/2/04 se A			1/1/08	2/2/04	8/3/05	4/12/07	4/12/07	1/1/08	10/1/07
WBS Description NASA HQ Milestones Formulation Authorization Doc Signed / ATP Phase A Initial Confirmation Review & Confirmation Assessment / AT NASA/ESA Memorandum Of Understanding (Preliminary) Non-Advocate Review NASA/ESA Memorandum Of Understanding (Final) Program Commitment Agreem Approval / ATP Phase C/D	ription rization Doc se A Review & Ssment / AT randum Of eliminary) riew randum Of randum Of nent Agreem ase C/D	Start		2/2/04	2/2/04	8/3/05	4/12/07	4/12/07	1/1/08	10/1/07
		Description		NASA HQ Milestones	Formulation Authorization Document Signed / ATP Phase A	Initial Confirmation Review & Confirmation Assessment / ATP Phase B	NASA/ESA Memorandum Of Understanding (Preliminary)	Non-Advocate Review	NASA/ESA Memorandum Of Understanding (Final)	Program Commitment Agreement Approval / ATP Phase C/D
D	WBS	WBS								
		₽		4	15	16	17	18	19	20

LISA MISSION REVIEWS

2011	1											△ 2/15	PSR △ 4/2	// V	8	◁
2010 2011 1 2 3 4 1 2 3										10/28	△ 2/15	<u>.</u>	PSR			
2008 2009 1 2 3 4 1 2 3 4									△ 2/25	Obs1 PER 🛆 10/28						
2008								△ 3/13		Obs1						
2006 2007 1 2 3 4 1 2 3 4						△ 11/15	△ 4/12									
			.	08/6 🗸	△ 12/30	7										
2005			△ 5/16	◁	7											
2004		△ 12/15														
2003		7														
Duration 4	2000 d	po	p o	po	po	po	po	PO	p O	po	po	p o	po	po	po	p o
Finish	8/14/11	12/15/03	5/16/05	9/30/02	12/30/05	11/15/06	4/12/07	3/13/08	2/25/09	10/28/09	2/15/10	2/15/11	4/27/11	7/15/11	8/12/11	8/14/11
Start	12/15/03	12/15/03	5/16/05	9/30/02	12/30/05	11/15/06	4/12/07	3/13/08	2/25/09	10/28/09	2/15/10	2/15/11	4/27/11	7/15/11	8/12/11	8/14/11
Description	LISA Major Mission Reviews	Mission Concept Review	Mission Definition Review	System Requirements Review	System Concept Review	Technology Readiness Review	Mission Preliminary Design Review (with Spacecraft PDR)	Mission Critical Design Review (with Spacecraft CDR)	Spacecraft Bus Flight 1 PER	Observatory Flight 1 PER	Mission Operations Review	Operational Readiness Review	Constellation Pre-Ship Review	Mission Readiness Review	Flight Readiness Review	Launch Readiness Review
WBS																
Q	21	22	23	24	22	26	27	78	23	30	31	32	33	34	35	98

LISA SYSTEM ENGINEERING & INTEGRATION CONTRACTOR ACQUISITION CYCLE



LISA ST-7 DISTURBANCE REDUCTION SYSTEM (SM2)

SECTION H: APPENDICES

2007 2008 2009 2010 2011 1 2 3 4 1 2 3 4 1 2 3 4 1 2 3 4 2010 2011																								
16 2007 2008 3 4 1 2 3 4 1 2 3 4 1	4											1											△ 7/18	
2003 2004 2005 2006 1 2 3 4 1 2 3						2	7	0	-	/12				√ 3/1	_			△ 2/28		∇ 4/25		0 6/29	Q	
2003		4				△ 5/2	△ 3/17	△ 4/10	△ 4/11	△ 6/12		4												
Duration 4	1293 d	325 d	120 d	315 d	167 d	p 0	рO	p o	p 0	p o	58 d	p 896	161 d	p o	5 d	303 d	41 d	P 0	76 d	p o	45 d	p o	ро	67 d
Finish	3/14/07	6/29/03	7/28/02	6/8/03	5/2/03	5/2/03	3/17/03	4/10/03	4/11/03	6/12/03	6/29/03	3/14/07	2/9/04	3/1/04	3/5/04	1/2/05	2/28/05	2/28/05	5/15/05	4/25/05	6/29/05	6/29/05	7/18/06	3/14/07
Start	3/31/02	3/31/02	3/31/02	7/29/02	9/12/02	5/2/03	3/17/03	4/10/03	4/11/03	6/12/03	5/3/03	6/30/03	6/30/03	3/1/04	3/1/04	3/6/04	1/3/05	2/28/05	3/1/05	4/25/05	5/16/05	6/29/05	7/18/06	12/12/06
Description	ST7 Disturbance Reduction System (SM2)	Formulation Refinement	Conceptual Design & Requirements Definition	Preliminary Design & Breadboard Development	STM/Simulator Development	STM/Simulator Delivery	Technology Readiness Review	Preliminary Design Review	Confirmation Readiness Review	Confirmation Review	Schedule Reserve	Implementation	Detailed Design & Brassboard Development	Critical Design Review	Schedule Reserve	Flight Hardware Development	Schedule Reserve	Subsystem Delivery	DRS I&T and Environmental Test	Environmental Test Readiness Review	Schedule Reserve	DRS Delivery to SMART 2	SM2 Launch	DRS Demo
WBS																								
Ω	49	20	51	52	53	54	22	26	22	28	29	09	61	62	63	64	65	99	29	89	69	70	71	72

LISA TEST PACKAGE (SM2)

2011																								
2003 2004 2005 2006 2007 2008 2009 2010 2011 41 23 41 24 24 <td></td>																								
2009																								
2008																								
2007																			<u>.</u>					
2006	1	4																				4		
2005																								
2004																							_	
2003									വ	1	7 1/24		△ 3/10		_	_		_	△ 4/25	△ 5/14	△ 6/18	1	∨ 6/20	
	_	2p 0	1 d	1 d	208 d?	67 d?	65 d?	¿p 09	1 d 7/15	105 d	1 d	30 d	1 d	14 d	5 d	1 d	3 d	10 d	1 d	2 d	2 d	776 d?	1 d	100 d
Duration	1480 d?	1349 d?			208	9	99)9		5												776		5
Finish	12/11/06	90/6/9	4/10/01	9/26/01	7/15/02	12/28/01	3/29/02	7/12/02	7/15/02	6/19/03	1/24/03	3/7/03	3/10/03	3/28/03	4/4/03	4/7/03	4/10/03	4/24/03	4/25/03	5/15/03	6/19/03	90/6/9	6/20/03	11/7/03
Start	4/10/01	4/10/01	4/10/01	9/26/01	9/27/01	9/27/01	12/31/01	4/22/02	7/15/02	1/24/03	1/24/03	1/27/03	3/10/03	3/11/03	3/31/03	4/7/03	4/8/03	4/11/03	4/25/03	5/14/03	6/18/03	6/20/03	6/20/03	6/23/03
Description	LISA Test Package (SM2)	SMART-2 S/C	Definition Phase ITT	Definition Phase Kick-off	Definition Phase	Preliminary Mission Architecture	S/C & Technology Concept	Mission Definition & Design	SRR	Implementation Phase ITT	ITT Issue	Bids Preparation	Receive Offers	Evaluation by panels	Report preparation	Technical Evaluation Board	Report finalization	Contract proposal preparation	AC/IPC contr. prop. subm.	AC	Contract Proposal Approval by Industrial Policy Committee	Implementation Phase	Implementation Phase Kick-off	Design finalization
WBS																								
<u>□</u>	73	74	75	9/	22	78	62	08	81	82	83	84	82	98	87	88	68	06	91	92	93	96	95	96

2003 2004 2005 2006 2007 2008 2009 2010 2011 4112341234123411234112341123411234412344	71/10		\(\sigma\) 107		△ 10/11		6/9 🗸	4				717			∇ 6/30	1		△ 4/1	∇ 6/30	1		Δ 7/1/8			
Duration 2003 4 1 2 3 4	10	237 d	1 d?	262 d	1 d	172 d	1 d	984 d	549 d	p 89	197 d	1 d	262 d	215 d	1 d	861 d	76 d	1 d	1 d	131 d	27 d	p 0	30 d	7 d	p /9
Finish	11/10/03	10/6/04	10/7/04	10/10/05	10/11/05	90/8/9	90/6/9	90/06/9	10/30/03	1/31/03	12/31/03	1/1/04	9/1/04	6/29/05	90/08/9	6/30/05	6/27/02	4/1/04	6/30/05	12/11/06	7/18/06	7/18/06	8/29/06	90/2/6	12/11/06
Start	11/10/03	11/11/03	10/7/04	10/8/04	10/11/05	10/12/05	90/6/9	9/24/01	9/24/01	10/30/02	4/1/03	1/1/04	9/2/03	9/2/04	90/06/9	3/14/02	3/14/02	4/1/04	90/06/9	6/12/06	6/12/06	7/18/06	7/19/06	90/06/8	90/8/6
Description	PDR	EM Manufacturing	CDR	FM Manufacturing	S/S Acc. Revs	AIT	FAR - S/C Delivery	LΤР	Design & S/A EM	LTP STPM	LTP EM	CDR	FM manufacturing	AIT	LTP Delivery	DRS (ESA Reqs)	Reqs & I/F def.	CDR	Receive DRS From JPL	Phase E/F	Launch campaign	Launch	Transfer to final orbit	Commissioning	LISA demo
WBS																									
Ω	26	86	8	100	101	102	103	104	105	106	107	108	109	110	111	112	113	114	115	116	117	118	119	120	121

KEY FORMULATION STAGE PRODUCTS

9 2010 2011																								
2007 2008 2009 2010 1 2 3 4 1 2	4															2	2		△ 8/1					12
2006 2007 2 3 4 1 2 3		1									◁					△ 3/12	△ 4/12		◁	1			08/6	△ 4/12
2005	-									◁				√ 7/30	7/6 ♥							△ 7/15	6	
2003 2004 4 1 2 3 4 1 2 3 4			1	△ 2/3	√ 9/1	4	Complete	Complete	◁			Complete	√ 5/30	7				08/9 🗸			△ 5/15	7		
Duration	1532 d	1261 d	150 d	P0	p0	1261 d	po	ро	ΡO	PO	PO	PO	p0	p ₀	p0	PO	P O	po	po	1020 d	PO	PO	ро	0 d
Finish	3/13/08	8/1/07	9/1/03	2/3/03	9/1/03	8/1/07	10/1/02	10/1/02	5/1/03	8/1/05	4/2/07	12/16/02	5/30/03	7/30/04	9/1/05	3/12/07	4/12/07	60/06/9	8/1/07	4/12/07	5/15/03	7/15/04	9/30/02	4/12/07
Start	4/30/02	10/1/02	2/3/03	2/3/03	9/1/03	10/1/02	10/1/02	10/1/02	5/1/03	8/1/05	4/2/07	12/16/02	2/30/03	7/30/04	9/1/05	3/12/07	4/12/07	6/30/03	8/1/07	5/15/03	5/15/03	7/15/04	9/30/02	4/12/07
Description	Key Formulation Products Summary	Project Management Formulation Products	Formulation Plan	Initial (TRIP Report Submission)	Update	Management Agreements	NASA/ESA Definition Phase LOA	GSFC/JPL Management Agreement	GSFC/JPL Management Agreement - Update	GSFC/JPL Management Agreement - Update	GSFC/JPL Management Agreement - Update	NASA/ESA Pathfinder Agreement - Initial Draft for	NASA/ESA Pathfinder Agreement - Update	NASA / ESA MOU - Draft	JPL/ESA SMART-2 MOU	Agreement with NASA HQ on Reporting Frequency & Content		Project Plan Outline - Initial	Project Plan Outline - Update	Project Plan - Initial	Project Plan - Update			
WBS																								
Ω	122	123	124	125	126	127	128	129	130	131	132	133	134	135	136	137	138	139	140	141	142	143	44	145

2005 2006 1 2 3 4 1 2 3 4 1	△ 4/12	4		√ 9/30	1		√ 9/30	1		06/30	1		08/30	1		06/30	1		√ 9/30	1		₩ 29/30	4	
2003 2004 1 2 3 4 1 2 3 4		4	△ 3/28			△ 4/30			√ 5/30		ا	A 2/3		4	2/3		4	△ 3/18		4	A 2/3			08/6 🗸
Duration 4	рO	929 p	po	p 0	632 d	p o	p 0	610 d	p o	p 0	694 d	p 0	ро	694 d	p ₀	p 0	963 d	p 0	P 0	694 d	p 0	p 0	328 d	p 0
Finish	4/12/07	9/30/02	3/28/03	9/30/02	9/30/02	4/30/03	9/30/02	9/30/02	5/30/03	9/30/02	9/30/02	2/3/03	9/30/02	9/30/02	2/3/03	9/30/02	9/30/02	3/18/03	9/30/02	9/30/02	2/3/03	9/30/02	12/31/04	6/30/03
Start	4/12/07	3/28/03	3/28/03	9/30/02	4/30/03	4/30/03	9/30/02	5/30/03	5/30/03	9/30/02	2/3/03	2/3/03	9/30/02	2/3/03	2/3/03	9/30/02	3/18/03	3/18/03	9/30/02	2/3/03	2/3/03	9/30/02	6/30/03	6/30/03
Description	Program Plan	Organization Approach	Organizational Approach - White Paper	Organizational Approach - Update For Project Plan	Development Strategy	Development Strategy - White Paper	Development Strategy - Update For Project Plan	Acquistition Strategy	Acquisition Strategy - White Paper	Acquisition Strategy - Update For Project Plan	Work Breakdown Structure	WBS - TRIP Report	WBS - Update for Project Plan	Budget & Life Cycle Cost Estimates	Budget & LCC - Preliminary Estimate	Budget & LCC - Update For Project Plan	Reserves Management	Reserves Mgt. Approach - White Paper	Reserves Mgt. Approach - Update For Project Plan	Integrated Schedule	Integrated Schedule - TRIP	Integrated Schedule - Update For Project Plan	Independent External Review Plan	Independent External Review Plan - Draft
WBS																								
Ω	146	147	148	149	150	151	152	153	154	155	156	157	158	159	160	161	162	163	164	165	166	167	168	169

2010 2011 1 2 3 4 1 2 3 4																								
2009																								
2008							2							2	CI	2	CI	6	4					
2007					1		△ 4/12							△ 4/12	△ 4/12	△ 4/12	△ 4/12	△ 4/12						
2006				9/30		9/30				9/30			/30											/30
2005	△ 12/31	1		◁	4	◁		1		◁	1		08/9 🗸		ļ					1				∇ 6/30
2004	,								\triangle 2/27		1	△ 12/31											08/6	
2003			△ 2/3									,									30	△ 2/3	◁	
Duration 2	p 0	694 d	po	p O	299 d	PO	PO	415 d	ρO	p o	391 d	PO	рo	рo	p O	po	P O	p O	1532 d	827 d	0 0	P 0	p 0	p 0
Finish	12/31/04	9/30/02	2/3/03	9/30/02	4/12/07	9/30/02	4/12/07	9/30/02	2/27/04	9/30/02	90/08/9	12/31/03	90/08/9	4/12/07	4/12/07	4/12/07	4/12/07	4/12/07	3/13/08	90/08/9	4/30/02	2/3/03	6/30/03	90/02/9
Start	12/31/04	2/3/03	2/3/03	9/30/02	9/30/02	30/02/6	4/12/07	2/27/04	2/27/04	9/30/02	12/31/03	12/31/03	90/08/9	4/12/07	4/12/07	4/12/07	4/12/07	4/12/07	4/30/02	4/30/02	4/30/02	2/3/03	6/30/03	6/30/05
Description	Independent External Review Plan - Update For Project Plan	Staffing Plan	Staffing Plan - TRIP	Staffing Plan - Update For Project Plan	Data Management Plan	Initial	Update	Scientific, Technical & Economic Benefits Definition	Scientific, Technical & Economic Benefits Definition -	Scientific, Technical & Economic Benefits Definition -	Facility Requirements	Facility Rqts White Paper	Facility Rqts Update	Technology and Commercialization Plan	Descope Plan	Notices of Intent for Environmental Impact	Cancellation Review Criteria	National Environmental Policy Act Compliance Documentation - Draft	System Engineering Formulation Products	Requirements	Level 1 Requirements Document - 1st Draft	Level 1 Requirements - TRIP Update	Level 1 Requirements Document - 2nd Draft	Level 1 Requirements Document - Final
WBS																								
Ω	170	171	172	173	174	175	176	177	178	179	180	181	182	183	184	185	186	187	188	189	190	191	192	193

2006 2007 2008 2009 2010 2011 4 2 3 4 2 3 4 2 3 4 2 3 4 1 2 3 4 1 2 3 4 3 3 4 3 4 3 4 3 4 3 4 3 4 3 4 3 4 4		∇ 6/30	4				△ 4/12			∇ 6/30				4						16	△ 4/12	4		
2003 2004 2005 2006 1 2 3 4 1 2 3 4 1 2 3 4 1 2 3 4	08/9 🗸	₫			△ 4/30	△ 4/30		4	△ 3/31	ð	1	∑ 5/30	∇ 4/30	4	2/3	△ 4/30	△ 7/31	√ 4/30	△ 9/15	2/16		4	08/6	√ 4/30
Duration 4	p 0	p O	1292 d	0 d 30	p o	p o	p ₀	288 d	p 0	p o	240 d	p o	p 0	1093 d	7 po	p o	p O	p O	p o	p 0	p o	922 d	p o	p 0
Finish	6/30/04	90/06/9	4/12/07	4/30/02	4/30/03	4/30/04	4/12/07	90/06/9	3/31/03	90/06/9	4/30/04	2/30/03	4/30/04	4/12/07	2/3/03	4/30/03	7/31/03	4/30/04	9/15/03	5/16/05	4/12/07	4/12/07	£0/0£/6	4/30/04
Start	6/30/04	9/30/02	4/30/02	4/30/02	4/30/03	4/30/04	4/12/07	3/31/03	3/31/03	9/30/02	5/30/03	2/30/03	4/30/04	2/3/03	2/3/03	4/30/03	7/31/03	4/30/04	9/15/03	5/16/05	4/12/07	9/30/03	6/30/03	4/30/04
Description	System Requirements Document - 1st Draft	System Requirements Document - Final	System Engineering Management Plan (SEMP)	SEMP - 1st draft	SEMP - 2nd draft	SEMP - 3rd Draft	SEMP - Update	Traceability Methodology	Traceability Methodology - White Paper	Traceability Methodology - Update for Mission Rgts. Draft	Organizational Approach	Org Approach - White Paper	Org Approach - Update for SEMP	Reference Design	Reference Design - TRIP Update	Reference Design - Update	Reference Design - Update	Reference Design - Update for SEMP	Architecture & Design Specification - Draft	Architecture & Design Specification - Update	Architecture & Design Specification - Update	Interface Control Documents Definition	ICD Definition - Initial	ICD Definition - Update for SEMP
WBS																								
Ω	194	195	196	197	198	199	200	201	202	203	204	205	506	207	208	209	210	211	212	213	214	215	216	217

2007 2008 2009 2010 2011 1 2 3 4 1 2 3 4 1 2 3 4 1 2 3 4 2011 2011						3/13																		
2007	△ 4/12	1			△ 4/12	◁	△ 4/12	4				△ 4/12												
2003 2004 2005 2006 1 2 3 4 1 2 3 4 1 2 3 4 2005 2006				Q						6/30	08/9 🗸				Q							∑ 5/16	08/8	08/6 ▽
2003 2004 2 3 4 1 2 3 4		4	⊘ 6/30	√ 4/30					√ 6/30	9			1	↑ 12/31	∇ 4/30		△ 12/15	△ 12/15	△ 12/15	△ 12/15	△ 12/15			
Duration 4 1	PO	1228 d	рo	рo	p o	po	рo	p 886	рo	P O	рo	рo	87 d	рo	рo	784 d	рo	P0	p o	po	p o	рo	рo	p O
Finish	4/12/07	3/13/08	6/30/03	4/30/04	4/12/07	3/13/08	4/12/07	4/12/07	6/30/03	6/30/04	90/08/9	4/12/07	4/30/04	12/31/03	4/30/04	12/15/06	12/15/03	12/15/03	12/15/03	12/15/03	12/15/03	5/16/05	9/30/02	9/30/02
Start	4/12/07	6/30/03	6/30/03	4/30/04	4/12/07	3/13/08	4/12/07	6/30/03	6/30/03	6/30/04	90/08/9	4/12/07	12/31/03	12/31/03	4/30/04	12/15/03	12/15/03	12/15/03	12/15/03	12/15/03	12/15/03	5/16/05	9/30/02	9/30/02
Description	Preliminary ICDs	Integation & Test Flow / Plan	I&T Strategy/Flow - Update	I&T Strategy/Flow - Update for SEMP	I&T Plan - Initial (PDR)	I&T Plan - Update (CDR)	EMC Plan - Initial (PDR)	Verification	Verfication Concept - White Paper	Verification Matrix - Initial	Verification Matrix - Update	Verification Plan - Initial	End-To-End (Constellation) Testing	ETE Test - White Paper	ETE Test - Update for SEMP	Primary Trade Studies	Phased Array vs. High Gain Antenna (at MCR)	Star Tracker Mounting (at MCR)	Telescope Mechanism (at MCR)	Design Optimization to Simplify Interfaces (at MCR)	Location of Observatory I&T (at MCR)	Ranging and S/C-to-S/C Communication (at MDR)	Telescope vs. Laser Power (at SRR)	Redundancy vs. Lifetime at (SRR)
WBS																								
Ω	218	219	220	221	222	223	224	225	526	227	228	529	230	231	232	233	234	235	236	237	238	239	240	241

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2003 2004 2005 2006 2007 2008 2009 2010 2011 41 2 3 4 4 2 3 4 4 2 4 4 2 3 4 4 2 3 4 4 2 3 4 4 2 3 4 4 2 3 4 4 2 3 4 4 2 3 4 4 2 3 4 4 2 3 4 4 2 3 4 4 2 3 4 4 2 3 4 4 2 3 4 4 2 3 4 4 2 3 4 4 2 3 4 4 3 4 4 3 4 4 3 4 4 3 4 4 3 4 4 3 4 4 3 4 4 3 4 4 3 4 4 3 4 4 4 4 3 4		△ 9/30	Å 12/15	1	∇ 8/29	△ 4/30	∇ 4/30	1	A 23	△ 7/31	△ 4/30	4	A 23	∇ 4/30	∇ 4/30	√ 9/30	△ 4/12	4	∇ 6/30	∇ 4/30	1	△ 731	△ 4/30	4
Duration	РO	p o	p o	175 d	P O	P O	p 0	324 d	PO	p o	P0	1093 d	PO	PO	P0	PO	p ₀	219 d	P O	p O	196 d	P O	P0	324 d
Finish	9/30/02	9/30/02	12/15/06	4/30/04	8/29/03	4/30/04	4/30/04	4/30/04	2/3/03	7/31/03	4/30/04	4/12/07	2/3/03	4/30/03	4/30/04	9/30/02	4/12/07	4/30/04	6/30/03	4/30/04	4/30/04	7/31/03	4/30/04	4/30/04
Start	9/30/02	9/30/02	12/15/06	8/29/03	8/29/03	4/30/04	4/30/04	2/3/03	2/3/03	7/31/03	4/30/04	2/3/03	2/3/03	4/30/03	4/30/04	9/30/02	4/12/07	6/30/03	6/30/03	4/30/04	7/31/03	7/31/03	4/30/04	2/3/03
Description	Gravity Balancing (at SRR)	Optical Bench Material Selection (at SRR)	Strategy for Integration of Flight GRS (at Payload PDR)	Technical Resource Allocation Process	TRAP - White Paper	TRAP - Update for SEMP	Resource Allocations to Demonstrate Feasibility - Draft	Error Budget	Error Budget - TRIP Update	Error Budget - Update	Error Budget - Update for SEMP	Risk Management Plan	Risk Mgt. Plan - TRIP	Risk Mgt. Plan - Update	Technical Documentation Approach	Tech Doc Approach - White Paper	Tech Doc Approach - Update for SEMP	Documentation Tree	Doc Tree - Initial	Doc Tree - Update	Equipment List			
WBS																								
Ω	242	243	244	245	246	247	248	249	250	251	252	253	254	255	256	257	258	259	260	261	262	263	264	265

2004 2005 2006 2007 2008 2009 2010 2011 1 2 3 4 4 2 4 3 4 3 4 4 3 4 4 3 4 4 3 4 4 3 4 4 4 3 4 4 4 3 4 4 4 3 4 4 4 4																								
2007 20			4				△ 4/12	4		△ 4/12		△ 4/12	4				△ 4/12	△ 4/12	4			△ 4/12	4	
2006						9/30					△ 12/30					9/30					△ 12/30		4	△ 12/30
2005					△ 12/31	\triangleleft		1	△ 5/16		7			1		◁			1	△ 5/16	7			
2004		△ 4/30		œ	7										⊘ 8/29									
2003	2/3		Ţ	0€/9 ▽									4	4	◁									
Duration	ρO	p o	p 886	P O	PO	ро	рo	498 d	p 0	рO	p O	ρO	944 d	545 d	po	p O	PO	рO	498 d	ρO	PO	ρO	334 d	p O
Finish	2/3/03	4/30/04	4/12/07	6/30/03	12/31/04	90/08/6	4/12/07	4/12/07	5/16/05	4/12/07	12/30/05	4/12/07	4/12/07	9/30/02	8/29/03	9/30/02	4/12/07	4/12/07	4/12/07	5/16/05	12/30/05	4/12/07	4/12/07	12/30/05
Start	2/3/03	4/30/04	6/30/03	6/30/03	12/31/04	9/30/02	4/12/07	5/16/05	5/16/05	4/12/07	12/30/05	4/12/07	8/29/03	8/29/03	8/29/03	9/30/02	4/12/07	4/12/07	5/16/05	5/16/05	12/30/05	4/12/07	12/30/05	12/30/05
Description	Equipment List - TRIP	Equipment List - Update	Configuration Management Approach	CM Approach - White Paper	CM Approach - Update	CM Plan - Initial	CM Plan - Update	Disposal, Re-Entry & Decommision Plan	DR&D Plan - Initial (MDR)	DR&D Plan - Update (PDR)	Environmental Assessment (SCR)	Orbital Debris Assessment - Initial (PDR)	Mission Assurance Formulation Products	Flight Assurance Approach	Flight Assurance Approach - White Paper	Flight Assurance Approach - Update For Project Plan	Mission Assurance Guidelines (MAG) - Initial	Mission Assurance Requirements (MAR) - Initial	Safety Data Package	SDP - Initial (MDR)	SDP - Update (SCR)	SDP - Update (PDR)	System Safety Plan	System Safety Plan - Initial
WBS																								
<u> </u>	566	267	268	569	270	271	272	273	274	275	276	277	278	279	280	281	282	283	284	285	286	287	288	289

2003 2004 2005 2006 2007 2008 2009 2010 2011 12 3 4 12 3 4 12 3 4 12 3 4 12 3 4 12 3 4 12 3 4 2009 2010 2011	△ 4/12	△ 4/12	4	4	233	△ 11/3	△ 3/15	4	12/18	233	△ 7/31	∇ 4/29	△ 12/15	4	∇ 630	↑ 12/31	△ 1231	△ 1231	△ 1231	4	△ 3/31	∇ 6/30	∇ 6/30	4
Duration 4 1	ро	РO	661 d	551 d	РO	РО	рo	617 d	▼ po	PO PO	ΡO	рO	рO	394 d	p o	p o	рO	p O	рo	p 889	РO	рo	рO	266 d
Finish	4/12/07	4/12/07	6/30/05	3/15/05	2/3/03	11/3/03	3/15/05	4/29/05	12/18/02	2/3/03	7/31/03	4/29/05	12/15/03	12/31/04	6/30/03	12/31/04	12/31/03	12/31/03	12/31/04	90/08/9	3/31/03	6/30/04	6/30/05	6/30/05
Start	4/12/07	4/12/07	12/18/02	2/3/03	2/3/03	11/3/03	3/15/05	12/18/02	12/18/02	2/3/03	7/31/03	4/29/05	12/15/03	6/30/03	6/30/03	12/31/04	12/31/03	12/31/03	12/31/04	3/31/03	3/31/03	6/30/04	90/06/9	4/30/03
Description	System Safety Plan - Update	Contamination Control Plan - Initial	Science Formulation Products	Mission Concept	Mission Concept - TRIP Update	Mission Concept - Update	Mission Concept - Update	Science Requirements	Science Rqts Initial	Science Rqts TRIP	Science Rqts Update	Science Rqts Update	Education & Public Outreach Plan	Science Management	Science Mgt. Approach - White Paper	Science Mgt. Plan - Initial	Science Data Management Plan - Draft	Science Data Center Plan - Draft	Science Data Mgt Plan - Update	Mission Success Criteria	Mission Success Criteria - Initial	Mission Success Criteria - Update	Mission Success Criteria - Update	Minimal Mission
WBS																								
Ω	290	291	292	293	294	295	596	297	298	599	300	301	302	303	304	305	306	307	308	309	310	311	312	313

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2003 2004 2005 2006 2007 2008 2009 2010 2011 1 2 3 4	□ △ 4/30	\(\rangle \text{ 6/30}\)	∇ 6/30	<u> </u>	△ 12/15	1	4	A 2/3	△ 4/30	1	A 2/28	△ 9/30	4	4	\$\tilde{\Phi}\$ 2/3	∇ 4/30	1	₹ 2/28	△ 9/30	4	4	△ 9/15	∑ 5/16	△ 4/12
Duration 4	ΡO	P O	P O	P O	p O	694 d	324 d	P O	P O	675 d	P O	p O	694 d	324 d	PO	PO	675 d	p O	P O	1074 d	933 d	PO	po	p O
Finish	4/30/03	6/30/04	90/06/9	12/31/03	12/15/03	9/30/02	4/30/04	2/3/03	4/30/04	9/30/02	2/28/03	9/30/02	9/30/02	4/30/04	2/3/03	4/30/04	9/30/02	2/28/03	9/30/02	4/12/07	4/12/07	9/15/03	5/16/05	4/12/07
Start	4/30/03	6/30/04	9/30/02	12/31/03	12/15/03	2/3/03	2/3/03	2/3/03	4/30/04	2/28/03	2/28/03	9/30/02	2/3/03	2/3/03	2/3/03	4/30/04	2/28/03	2/28/03	9/30/02	2/28/03	9/15/03	9/15/03	5/16/05	4/12/07
Description	Minimal Mission - Initial	Minimal Mission - Update	Minimal Mission - Update	Astrophysics & Waveform of Sources - Initial Calculations	Initial Data Analysis Methods for Modeling	Payload System Formulation Products	Payload Concept	P/L Concept - TRIP Update	P/L Concept - Update for SEMP	Flight Segment (Payload) Performance Requirements &	Flight Segment (Payload) Performance Rqts. & Metrics -	Flight Segment (Payload)Performance Rqts. &	Flight System Formulation Products	Spacecraft Concept	S/C Concept - TRIP Update	S/C Concept - Update for SEMP	Flight Segment Performance Requirements & Metrics	Flight Segment Performance Rqts. & Metrics - Initial	Flight Segment Performance Rqts. & Metrics Update	Mission Operations System Formulation Products	Operations Concept	Ops Concept Definition - Initial	Ops Concept Definition - Update	Ops Concept Definition - Update
WBS																								
□	314	315	316	317	318	319	320	321	322	323	324	325	326	327	328	329	330	331	332	333	334	335	336	337

2006 2007 2008 2009 2010 2011 1 2 3 4			9						0			9			0			9			0			V 12/22
2005	4		△ 5/16	4		△ 5/16	1		△ 5/16	4		△ 5/16	1		△ 5/16	4		△ 5/16	1		△ 5/16			V
2004											30			9/15			9/15			△ 9/15			△ 9/15	
2003	4	√ 5/1		4	√ 2/28		1	△ 3/31		4	08/9 🗸		4	◁		4	◁		4	◁		4	◁	
Duration	532 d	PO	PO	p 929	PO	PO	p 999	po	PO	490 d	PO	ρO	435 d	PO	p O	435 d	PO	ρO	435 d	ρO	p O	p 669	ρO	рO
Finish	5/16/05	5/1/03	5/16/05	5/16/05	2/28/03	5/16/05	5/16/05	3/31/03	5/16/05	5/16/05	6/30/03	5/16/05	5/16/05	9/15/03	5/16/05	5/16/05	9/15/03	5/16/05	5/16/05	9/15/03	5/16/05	12/22/05	9/15/03	12/22/05
Start	5/1/03	5/1/03	5/16/05	2/28/03	2/28/03	5/16/05	3/31/03	3/31/03	5/16/05	6/30/03	6/30/03	5/16/05	9/15/03	9/15/03	5/16/05	9/15/03	9/15/03	5/16/05	9/15/03	9/15/03	5/16/05	9/15/03	9/15/03	12/22/05
Description	Organizational Approach	Org Approach - White Paper	Org Approach - Update	Driving Requirements	Driving Rqts White Paper	Driving Rqts Update	Orbital Parameters	Orbital Parameters - Initial	Orbital Parameters - Update	Communications Strategy	Comm Strategy - White Paper	Comm Strategy - Update	Data Collection Approach	Data Collection Approach - White Paper	Data Collectin Approach - Update	Data Policy	Data Policy - Initial	Data Policy - Update	Data Reduction Approach	Data Reduction Approach - White Paper	Data Reduction Approach - Update for Mission Ops Rqts	System Sizing	System Sizing - White Paper	System Sizing - Update
WBS																								
Ω	338	339	340	341	342	343	344	345	346	347	348	349	350	351	352	353	354	355	356	357	358	329	360	361

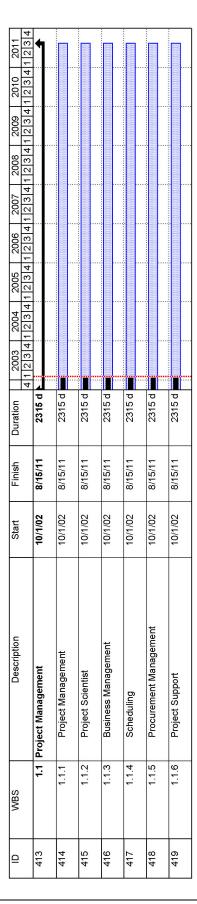
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WBS	Description Ground Segment Performance	Start 12/15/03	Finish 5/16/05	Duration 370 d	2003 4 1 2 3 4	2004	2005	2006	2007	2008 20	2009 20	2005 2006 2007 2008 2009 2010 2011 12 3 4 1 2 3 4 4 4 2 4 4 4 4 4 4 4 4 4 4 4 4 4 4
	Rqts. & Metrics Ground Segment Performance	12/15/03	12/15/03	P 0		∑ 12/15	1					
	Rgts. & Metrics - Initial	5/16/05	5/16/05	0			\ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \					
	Rqts. & Metrics - Update			3			2 7 1					
	Launch System Formulation Products	6/30/03	5/16/05	490 d	1		4					
	Launch Vehicle Options	6/30/03	4/30/04	219 d	4	1						
	LV Options - White Paper	6/30/03	6/30/03	p 0	06/9 🗸	õ						
	LV Options - Initial for SEMP	4/30/04	4/30/04	00		△ 4/30						
	Launch Segment Performance Requirements & Metrics	12/22/03	5/16/05	365 d			4					
	Launch Segment Performance Rqts. & Metrics - Initial	12/22/03	12/22/03	P 0	7	∆ 12/22						
	Launch Segment Performance Rqts. & Metrics Update	5/16/05	5/16/05	P 0			△ 5/16					
	Technology Development Formulation Products	10/1/02	6/30/05	717 d	4		4					
	Description/Rationale for Required Technologies	2/28/03	1/31/05	501 d	4		4					
	Description/Rationale - Draft for TDIP	2/28/03	2/28/03	p 0	∑ 2/28							
	Description/Rationale - Update for TDIP Update	1/31/05	1/31/05	P 0		7	√ 1/31					
	Alignment With Enterprise Roadmaps	2/28/03	1/31/05	501 d	4							
	AWER - Draft for TDIP	2/28/03	2/28/03	p 0	△ 2/28							
	AWER - Update for TDIP Update	1/31/05	1/31/05	p 0		7	△ 1/31					
	Definition of Technology Products	2/28/03	1/31/05	501 d	4		4					
	DoTP - Draft for TDIP	2/28/03	2/28/03	PO	∑ 2/28							
	DoTP - Update for TDIP Update	1/31/05	1/31/05	p 0			△ 1/31					
	Test Bed Definition	2/28/03	1/31/05	501 d	4							
	Test Bed Definition - Draft for TDIP	2/28/03	2/28/03	0 0	△ 2/28							
	Test Bed Definition - Update for TDIP Update	1/31/05	1/31/05	P 0		7	△ 1/31					
	Maturity Milestones/Investment Plan	2/28/03	1/31/05	501 d	4		4					

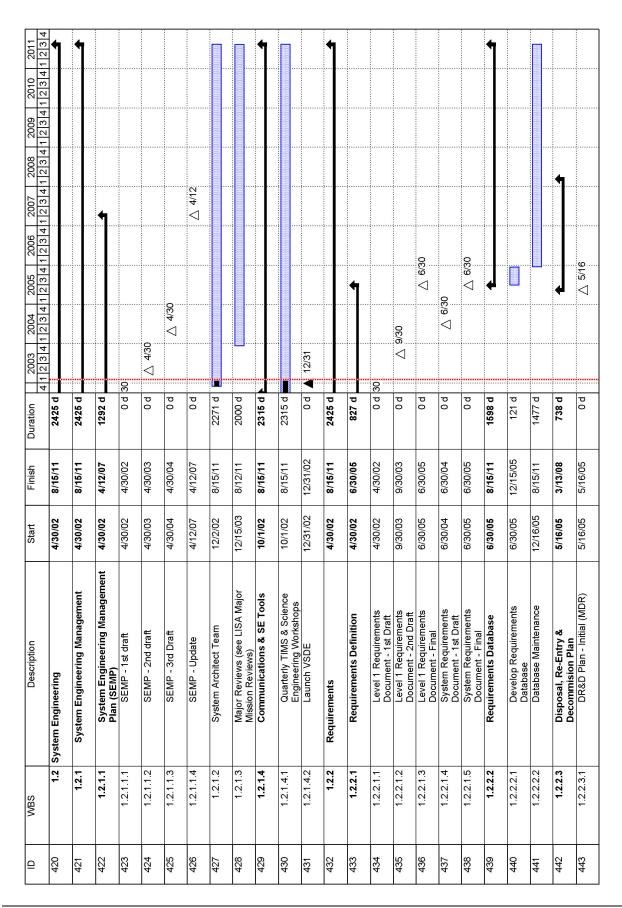
> <u> </u>	386	387	388	389	390	391	392	393	394	395	396	397	398	399	400	401	402	403	404	405	406	407	408	409
WBS																								
Description	MMIP - Draft for TDIP	MMIP - Update for TDIP Update	Integrated Modeling Approach	IM Approach - Draft for TDIP	IM Approach - Update for TDIP Update	Role of Technolgy Validation Mission	Role of TVM - Draft	Role of TVM - Update for TDIP Update	Independent Technology Readiness Assessment	TRIP Report Findings	2004 Independent Tech Review	2005 Independent Tech Review	Mission Software Formulation Products	Software Architecture/Developmental	S/W Architecture - White Paper	S/W Architecture - Update for SEMP	Role of IV&V	IV&V Role - White Paper	IV&V Role - Update for SEMP	Software Management Plan	Software Management Plan - Initial (MDR)	Software Management Plan - Update (FSW PDR)	Software Requirements Document	SRD - Initial (FSW/SSR)
Start	2/28/03	1/31/05	10/1/02	10/1/02	1/31/05	3/31/03	3/31/03	1/31/05	4/30/03	4/30/03	6/30/04	9/30/02	4/30/03	4/30/03	4/30/03	4/30/04	8/29/03	8/29/03	4/30/04	5/16/05	5/16/05	3/2/07	5/26/06	5/26/06
Finish	2/28/03	1/31/05	1/31/05	10/1/02	1/31/05	1/31/05	3/31/03	1/31/05	6/30/05	4/30/03	6/30/04	90/06/9	3/2/07	4/30/04	4/30/03	4/30/04	4/30/04	8/29/03	4/30/04	3/2/07	5/16/05	3/2/07	3/2/07	5/26/06
Duration	P O	P O	p 609	PO	РО	480 d	0 0	рO	266 d	0 0	P O	p O	1002 d	262 d	PO	PO	175 d	P O	p o	469 d	PO	P O	200 d	0
2003 2004 2 4 1 2 3 4 1 2 3 4 1	2/28	4	4	_ complete	ζ		△ 3/31	ζ	4	△ 4/30	06/30		1	1	√ 4/30	∇ 4/30	1	0 8/29	∇ 4/30	,	7			
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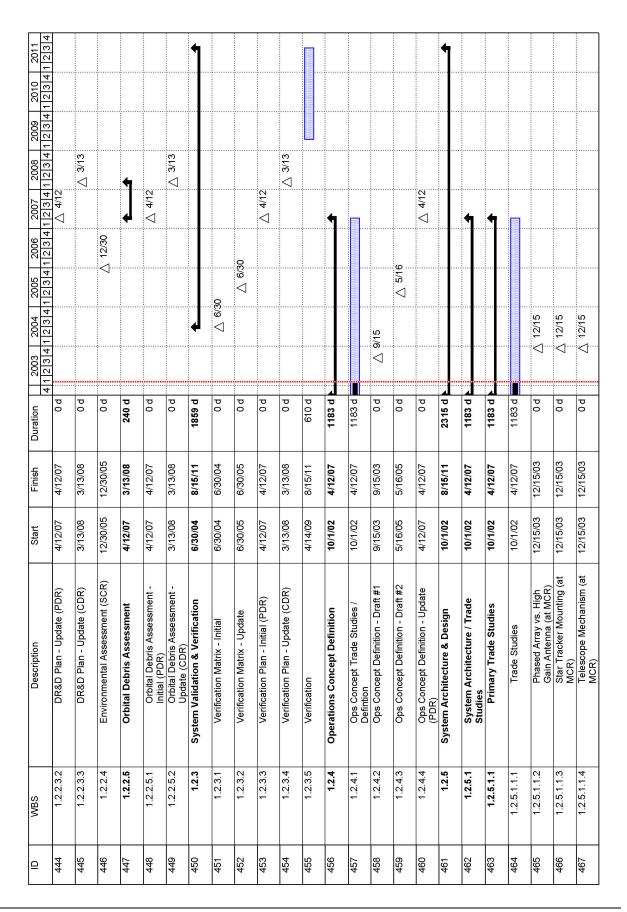
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Duration		P 0	P 0	p 0
Finish		3/2/07	3/2/07	3/2/07
듄		3/2	3/2	3/2
Start		3/2/07	3/2/07	3/2/07
Description		SRD - Update (FSW PDR)	Software Test Plan	Software Test Plan - Initial (FSW PDR)
WBS				
₹				
		410	411	412

LISA PROJECT MANAGEMENT WBS 1.1

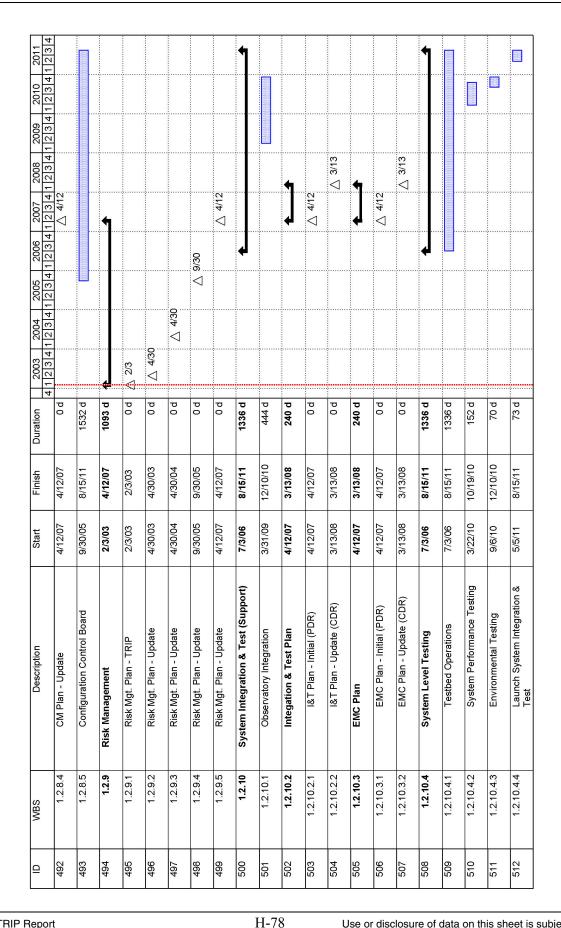


SYSTEM ENGINEERING WBS 1.2



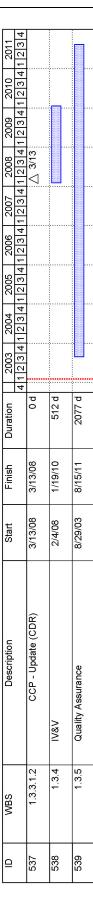


2003 2004 2005 2006 2007 2008 2009 2010 2011 2341234123412341234123412341234 234123412341234		△ 12/15	△ 5/16	02,8/30	₩ 29/30	9/30 △ (+ongoing updates)	△ 9/30	△ 12/15	△ 9/15	△ 5/16	△ 4/12		4		△ 4/12	△ 3/13	4				4	∇ 6/30	△ 1231	\(\nabla\) 9/30
Duration 2	P 0	P 0	p 0	p 0	p 0	рo	p 0	РO	p 0	P O	p 0	1120 d	1281 d	1281 d	РO	P O	2315 d	2271 d	2315 d 🚾	512 d	2121 d	p 0	рo	p 0
Finish	12/15/03	12/15/03	5/16/05	9/30/02	9/30/02	9/30/02	9/30/02	12/15/06	9/15/03	5/16/05	4/12/07	8/15/11	8/11/08	8/11/08	4/12/07	3/13/08	8/15/11	8/15/11	8/15/11	1/19/10	8/15/11	6/30/03	12/31/04	9/30/02
Start	12/15/03	12/15/03	5/16/05	9/30/02	9/30/02	9/30/02	9/30/02	12/15/06	9/15/03	5/16/05	4/12/07	5/1/07	9/15/03	9/15/03	4/12/07	3/13/08	10/1/02	12/2/02	10/1/02	2/4/08	6/30/03	6/30/03	12/31/04	9/30/02
Description	Design Optimization to Simplify Interfaces (at	Location of Observatory I&T (at MCR)	Ranging and S/C-to-S/C Communication (at MDR)	Telescope vs. Laser Power (at SRR)	Redundancy vs. Lifetime at (SRR)	Gravity Balancing (at SRR)	Optical Bench Material Selection (at SRR)	Strategy for Integration of Flight GRS (at Payload	Architecture & Design Specification - Draft	Architecture & Design Specification - Update	Architecture & Design Specification - Update	Integrated Modeling (Post-Technology Development)	Interfaces and ICDs	ICD Development	Preliminary ICDs (PDR)	ICD Updates (CDR)	Software System Engineering	Software System Management	Software Architecture	IV&V	Configuration Management	CM Approach - White Paper	CM Approach - Update	CM Plan - Initial
WBS	1.2.5.1.1.5	1.2.5.1.1.6	1.2.5.1.1.7	1.2.5.1.1.8	1.2.5.1.1.9	1.2.5.1.1.10	1.2.5.1.1.11	1.2.5.1.1.12	1.2.5.2	1.2.5.3	1.2.5.4	1.2.5.5	1.2.6	1.2.6.1	1.2.6.2	1.2.6.3	1.2.7	1.2.7.1	1.2.7.2	1.2.7.3	1.2.8	1.2.8.1	1.2.8.2	1.2.8.3
Ω	468	469	470	471	472	473	474	475	476	477	478	479	480	481	482	483	484	485	486	487	488	489	490	491

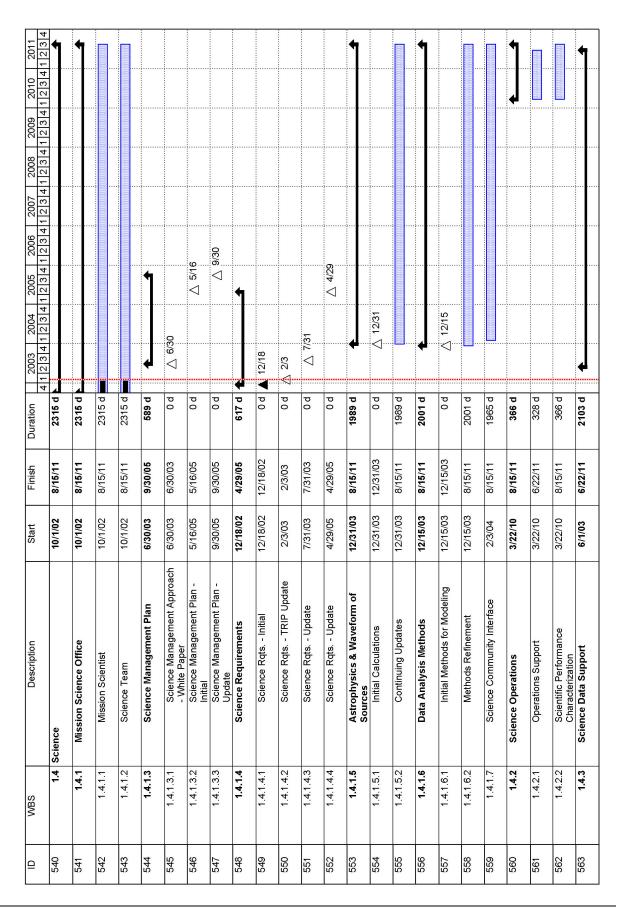


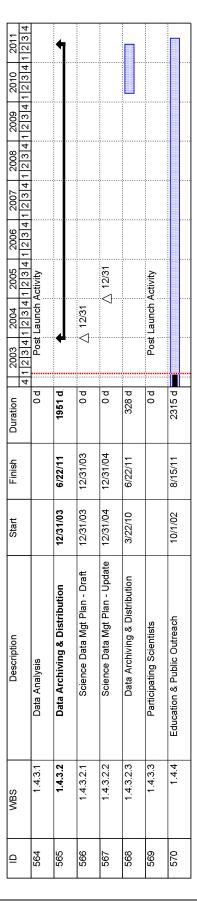
MISSION ASSURANCE WBS 1.3

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Duration 4	2077 d	1184 d	545 d	p ₀	P0	240 d	P O	PO	240 d	P O	PO	1629 d	1629 d	PO	PO	PO	PO	P O	334 d	PO	PO	240 d	240 d	p 0
Finish	8/15/11	3/13/08	9/30/02	8/29/03	9/30/02	3/13/08	4/12/07	3/13/08	3/13/08	4/12/07	3/13/08	8/12/11	8/12/11	5/16/05	12/30/05	4/12/07	3/13/08	8/12/11	4/12/07	12/30/05	4/12/07	3/13/08	3/13/08	4/12/07
Start	8/29/03	8/29/03	8/29/03	8/29/03	9/30/02	4/12/07	4/12/07	3/13/08	4/12/07	4/12/07	3/13/08	5/16/05	5/16/05	5/16/05	12/30/05	4/12/07	3/13/08	8/12/11	12/30/05	12/30/05	4/12/07	4/12/07	4/12/07	4/12/07
Description	1.3 Mission Assurance	Mission Assurance Management	Flight Assurance Approach	Flight Assurance Approach - White Paper	Flight Assurance Approach - Update For Project Plan	Mission Assurance Guidelines (MAG)	MAG - Initial (PDR)	MAG - Update (CDR)	Mission Assurance Requirements Document	MAR - Initial (PDR)	MAR - Update (CDR)	System Safety	Safety Data Package	SDP - Initial (MDR)	SDP - Update (SCR)	SDP - Update (PDR)	SDP - Update (CDR)	SDP - Update (FRR)	System Safety Plan	SSP - Initial (SCR)	SSP - Update (PDR)	Contamination Control	Contamination Control Plan	CCP - Initial (PDR)
WBS	1.3	1.3.1	1.3.1.1	1.3.1.1.1	1.3.1.1.2	1.3.1.2	1.3.1.2.1	1.3.1.2.2	1.3.1.3	1.3.1.3.1	1.3.1.3.2	1.3.2	1.3.2.1	1.3.2.1.1	1.3.2.1.2	1.3.2.1.3	1.3.2.1.4	1.3.2.1.5	1.3.2.2	1.3.2.2.1	1.3.2.2.2	1.3.3	1.3.3.1	1.3.3.1.1
Ω	513	514	515	516	517	518	519	520	521	522	523	524	525	526	527	528	529	530	531	532	533	534	535	536

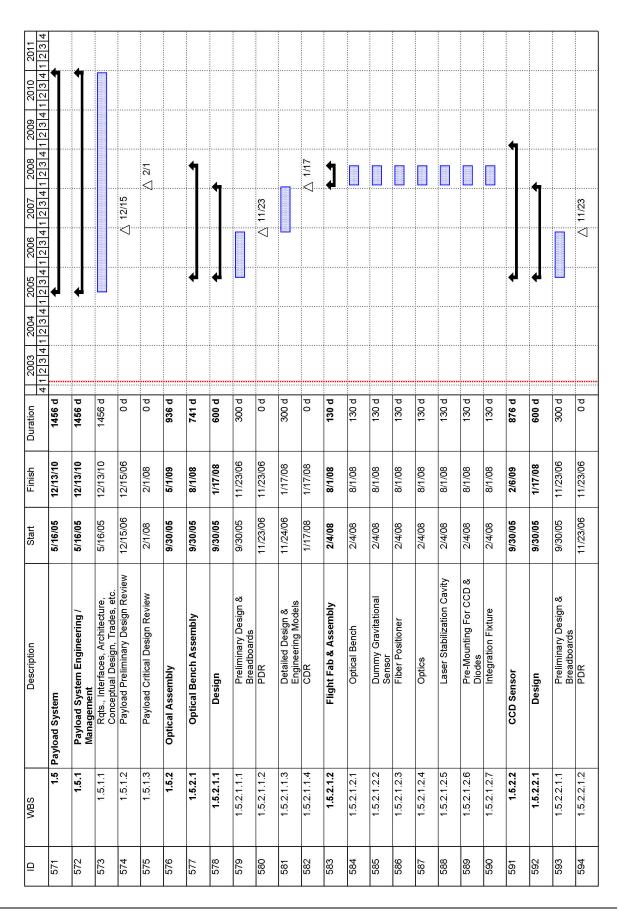


SCIENCE WBS 1.4

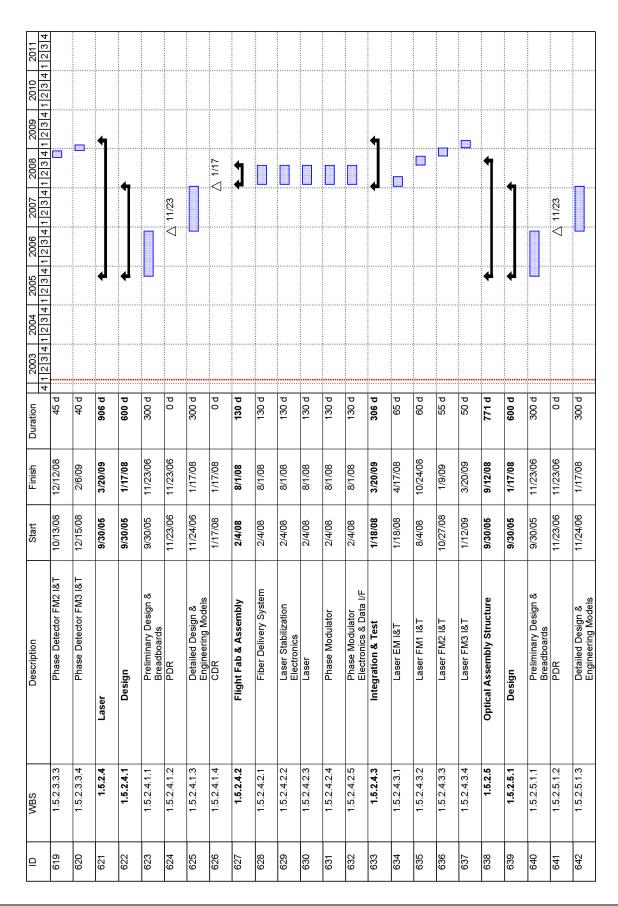




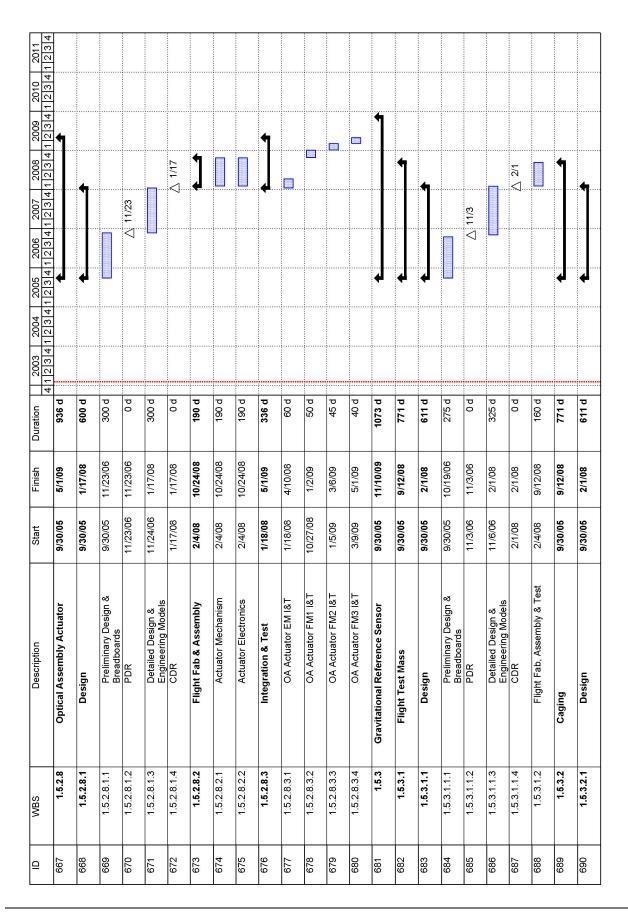
PAYLOAD SYSTEM WBS 1.5

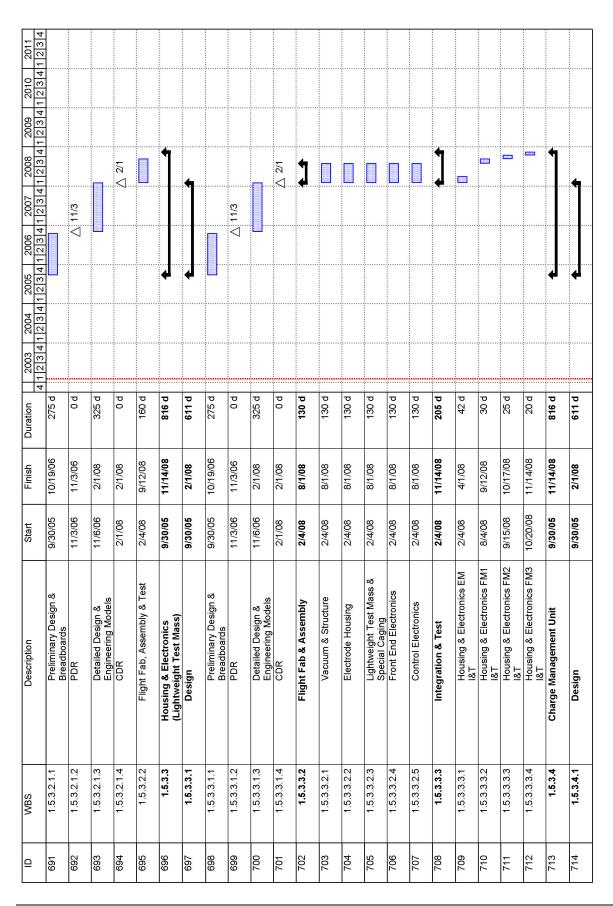


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Duration	900E	P O	130 d	130 d	130 d	276 d	p 09	50 d	45 d	40 d	876 d	P 009	300 d	P0	300 d	P0	130 d	130 d	130 d	130 d	130 d	276 d	p 09	50 d
Finish	1/17/08	1/17/08	8/1/08	8/1/08	8/1/08	2/6/09	4/10/08	10/10/08	12/12/08	2/6/09	2/6/09	1/17/08	11/23/06	11/23/06	1/17/08	1/17/08	8/1/08	8/1/08	8/1/08	8/1/08	8/1/08	2/6/09	4/10/08	10/10/08
Start	11/24/06	1/17/08	2/4/08	2/4/08	2/4/08	1/18/08	1/18/08	8/4/08	10/13/08	12/15/08	9/30/02	9/30/02	9/30/02	11/23/06	11/24/06	1/17/08	2/4/08	2/4/08	2/4/08	2/4/08	2/4/08	1/18/08	1/18/08	8/4/08
Description	Detailed Design & Engineering Models	CDR	Flight Fab & Assembly	CCD	CCD Electronics	Integration & Test	CCD Sensor EM I&T	CCD Sensor FM1 I&T	CCD Sensor FM2 I&T	CCD Sensor FM3 I&T	Phase Detector	Design	Preliminary Design & Breadboards	PDR	Detailed Design & Engineering Models	CDR	Flight Fab & Assembly	Diodes & Hardware	Pre-Amplifiers	Phase Readout Electronics	Ultra Stable Oscillator	Integration & Test	Phase Detector EM I&T	Phase Detector FM1 I&T
WBS	1.5.2.2.1.3	1.5.2.2.1.4	1.5.2.2.2	1.5.2.2.2.1	1.5.2.2.2.2	1.5.2.2.3	1.5.2.2.3.1	1.5.2.2.3.2	1.5.2.2.3.3	1.5.2.2.3.4	1.5.2.3	1.5.2.3.1	1.5.2.3.1.1	1.5.2.3.1.2	1.5.2.3.1.3	1.5.2.3.1.4	1.5.2.3.2	1.5.2.3.2.1	1.5.2.3.2.2	1.5.2.3.2.3	1.5.2.3.2.4	1.5.2.3.3	1.5.2.3.3.1	1.5.2.3.3.2
Ω	295	296	297	298	299	009	601	602	603	604	909	909	209	809	609	610	611	612	613	614	615	616	617	618



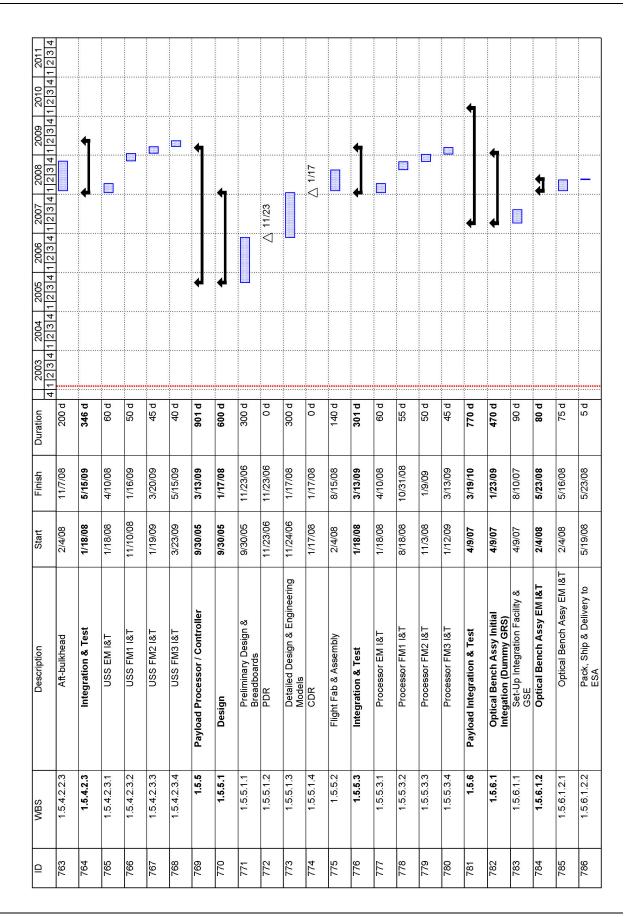
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Duration	рo	160 d	811 d	p 009	300 d	P O	300 d	ρO	200 d	911 d	P 009	300 d	0 0	900E	PO	150 d	150 d	150 d	150 d	311 d	p 09	55 d	50 d	45 d
Finish	1/17/08	9/12/08	11/7/08	1/17/08	11/23/06	11/23/06	1/17/08	1/17/08	11/7/08	3/27/09	1/17/08	11/23/06	11/23/06	1/17/08	1/17/08	8/53/08	8/29/08	8/29/08	8/29/08	3/27/09	4/10/08	11/14/08	1/23/09	3/27/09
Start	1/17/08	2/4/08	9/30/02	9/30/02	9/30/02	11/23/06	11/24/06	1/17/08	2/4/08	9/30/02	9/30/02	9/30/02	11/23/06	11/24/06	1/17/08	2/4/08	2/4/08	2/4/08	2/4/08	1/18/08	1/18/08	9/1/08	11/17/08	1/26/09
Description	CDR	Flight Fab, Assembly & Test	Star Tracker	Design	Preliminary Design & Breadboards	PDR	Detailed Design & Engineering Models	CDR	Flight Fab, Assembly & Test	Telescope	Design	Preliminary Design & Breadboards	PDR	Detailed Design & Engineering Models	CDR	Flight Fab & Assembly	Primary Optics	Secondary Optics	Telescope Structure	Integration & Test	Telescope EM I&T	Telescope FM1 I&T	Telescope FM2 I&T	Telescope FM3 I&T
WBS	1.5.2.5.1.4	1.5.2.5.2	1.5.2.6	1.5.2.6.1	1.5.2.6.1.1	1.5.2.6.1.2	1.5.2.6.1.3	1.5.2.6.1.4	1.5.2.6.2	1.5.2.7	1.5.2.7.1	1.5.2.7.1.1	1.5.2.7.1.2	1.5.2.7.1.3	1.5.2.7.1.4	1.5.2.7.2	1.5.2.7.2.1	1.5.2.7.2.2	1.5.2.7.2.3	1.5.2.7.3	1.5.2.7.3.1	1.5.2.7.3.2	1.5.2.7.3.3	1.5.2.7.3.4
Ω	643	644	645	646	647	648	649	650	651	652	653	654	655	656	657	658	629	099	661	662	993	664	999	999





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Duration	275 d	p o	325 d	p o	130 d	130 d	130 d	130 d	130 d	205 d	42 d	30 d	25 d	20 d	420 d	135 d	p 09	p 0/	5 d	100 d	30 d	65 d	5 d	100 d
Finish	10/19/06	11/3/06	2/1/08	2/1/08	8/1/08	8/1/08	8/1/08	8/1/08	8/1/08	11/14/08	4/1/08	9/12/08	10/17/08	11/14/08	11/10/09	10/7/08	6/24/08	80/06/6	10/7/08	2/17/09	11/11/08	2/10/09	2/17/09	6/30/08
Start	9/30/02	11/3/06	11/6/06	2/1/08	2/4/08	2/4/08	2/4/08	2/4/08	2/4/08	2/4/08	2/4/08	8/4/08	9/15/08	10/20/08	4/2/08	4/2/08	4/2/08	6/25/08	10/1/08	10/1/08	10/1/08	11/12/08	2/11/09	2/11/09
Description	Preliminary Design & Breadboards	PDR	Detailed Design & Engineering Models	CDR	Flight Fab & Assembly	Fibers	UV Lamps	UV Switch	UV Control	Integration & Test	CMU EM I&T	CMU FM1 I&T	CMU FM2 I&T	CMU FM3 I&T	GRS I&T	GRS EM I&T	Intial Flight Sensor Integration	Final Flight Sensor I&T	Pack, Ship & Delivery to NASA	GRS FM1 I&T	Intial Flight Sensor Integration	Final Flight Sensor I&T	Pack, Ship & Delivery to NASA	GRS FM2 I&T
WBS	1.5.3.4.1.1	1.5.3.4.1.2	1.5.3.4.1.3	1.5.3.4.1.4	1.5.3.4.2	1.5.3.4.2.1	1.5.3.4.2.2	1.5.3.4.2.3	1.5.3.4.2.4	1.5.3.4.3	1.5.3.4.3.1	1.5.3.4.3.2	1.5.3.4.3.3	1.5.3.4.3.4	1.5.3.5	1.5.3.5.1	1.5.3.5.1.1	1.5.3.5.1.2	1.5.3.5.1.3	1.5.3.5.2	1.5.3.5.2.1	1.5.3.5.2.2	1.5.3.5.2.3	1.5.3.5.3
Ω	715	716	717	718	719	720	721	722	723	724	725	726	727	728	729	730	731	732	733	734	735	736	737	738

2003 2004 2005 2006 2007 2008 2009 2010 2011 1 2 3 4 1 2 3 4 1 2 3 4 1 2 3 4 1 2 3 4 1 2 3 4 1 2 3 4 1 2 3 4 1 2 3 4 1 2 3 4 1 2 3 4 3 3 4 3 3 4 3 3				1				4	1	1		∑ 11/23		△ 1/17		4	1		<u>\(\frac{1}{2}\) 11/23</u>		△ 1/17	1		
Duration 4 1	30 d	65 d	5 d	100 d	30 d	92 d	5 d	946 d	771 d	p 009	300 d	рo	300 d	рo	160 d	946 d	p 009	300 d	рo	300 d	рo	200 d	200 d	200 d
Finish	3/24/09	6/23/09	60/08/9	11/10/09	8/4/09	11/3/09	11/10/09	5/15/09	9/12/08	1/17/08	11/23/06	11/23/06	1/17/08	1/17/08	9/12/08	5/15/09	1/17/08	11/23/06	11/23/06	1/17/08	1/17/08	11/7/08	11/7/08	11/7/08
Start	2/11/09	3/25/09	6/24/09	6/24/09	6/24/09	8/2/06	11/4/09	9/30/02	9/30/02	9/30/02	9/30/02	11/23/06	11/24/06	1/17/08	2/4/08	9/30/02	9/30/02	9/30/02	11/23/06	11/24/06	1/17/08	2/4/08	2/4/08	2/4/08
Description	Intial Flight Sensor Integration	Final Flight Sensor I&T	Pack, Ship & Delivery to NASA	GRS FM3 I&T	Intial Flight Sensor Integration	Final Flight Sensor I&T	Pack, Ship & Delivery to NASA	Y-Tube Assembly	Aff Fiber	Design	Preliminary Design & Breadboards	PDR	Detailed Design & Engineering Models	CDR	Flight Fab, Assembly & Test	Ultra Stable Structure	Design	Preliminary Design & Breadboards	PDR	Detailed Design & Engineering Models	CDR	Flight Fab & Assembly	Radiator	Y-tube Structure
WBS	1.5.3.5.3.1	1.5.3.5.3.2	1.5.3.5.3.3	1.5.3.5.4	1.5.3.5.4.1	1.5.3.5.4.2	1.5.3.5.4.3	1.5.4	1.5.4.1	1.5.4.1.1	1.5.4.1.1.1	1.5.4.1.1.2	1.5.4.1.1.3	1.5.4.1.1.4	1.5.4.1.2	1.5.4.2	1.5.4.2.1	1.5.4.2.1.1	1.5.4.2.1.2	1.5.4.2.1.3	1.5.4.2.1.4	1.5.4.2.2	1.5.4.2.2.1	1.5.4.2.2.2
Ω	739	740	741	742	743	744	745	746	747	748	749	750	751	752	753	754	755	756	757	758	759	760	761	762



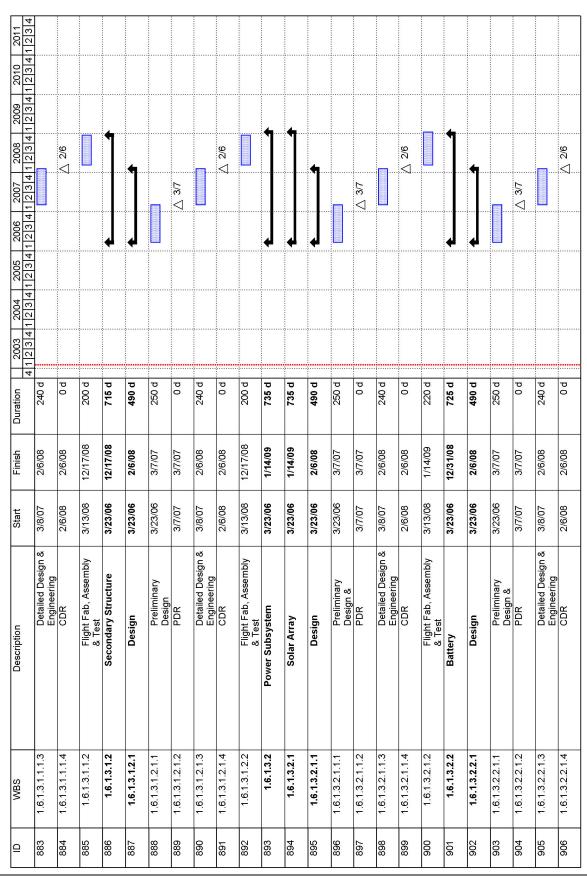
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Duration	P 09	45 d	5 d	45 d	40 d	5 d	40 d	35 d	ρς	260 d	p 9 9	p 09	5 d	p 9 9	20 d	5 d	P 09	45 d	ρς	45 d	40 d	5 d	225 d	p 09
Finish	10/10/08	10/3/08	10/10/08	12/5/08	11/28/08	12/5/08	1/23/09	1/16/09	1/23/09	5/22/09	8/22/08	8/15/08	8/22/08	1/16/09	1/9/09	1/16/09	3/20/09	3/13/09	3/20/09	5/22/09	5/15/09	5/22/09	7/3/09	11/14/08
Start	8/4/08	8/4/08	10/6/08	10/6/08	10/6/08	12/1/08	12/1/08	12/1/08	1/19/09	5/26/08	2/26/08	5/26/08	8/18/08	11/3/08	11/3/08	1/12/09	1/12/09	1/12/09	3/16/09	3/23/09	3/23/09	5/18/09	8/25/08	8/25/08
Description	Optical Bench Assy FM1 I&T	Optical Bench Assy FM1	Pack, Ship & Deliver to ESA	Optical Bench Assy FM2 I&T	Optical Bench Assy FM2 I&T	Pack, Ship & Deliver to ESA	Optical Bench Assy FM3 I&T	Optical Bench Assy FM3	Pack, Ship & Deliver to ESA	Electro-Optical Integration (Dummy GRS)	Electro-Optical EM I&T	Electro-Optical EM I&T	Pack, Ship & Delivery to NASA	Electro-Optical FM1 I&T	Electro-Optical FM1 I&T	Pack, Ship & Delivery to NASA	Electro-Optical FM2 I&T	Electro-Optical FM2 I&T	Pack, Ship & Delivery to NASA	Electro-Optical FM3 I&T	Electro-Optical FM3 I&T	Pack, Ship & Delivery to NASA	Optical Assembly Integration (Dummy GRS)	Optical Assy EM I&T
WBS	1.5.6.1.3	1.5.6.1.3.1	1.5.6.1.3.2	1.5.6.1.4	1.5.6.1.4.1	1.5.6.1.4.2	1.5.6.1.5	1.5.6.1.5.1	1.5.6.1.5.2	1.5.6.2	1.5.6.2.1	1.5.6.2.1.1	1.5.6.2.1.2	1.5.6.2.2	1.5.6.2.2.1	1.5.6.2.2.2	1.5.6.2.3	1.5.6.2.3.1	1.5.6.2.3.2	1.5.6.2.4	1.5.6.2.4.1	1.5.6.2.4.2	1.5.6.3	1.5.6.3.1
□	787	788	789	790	791	792	793	794	795	962	797	798	799	800	801	802	803	804	805	806	807	808	808	810

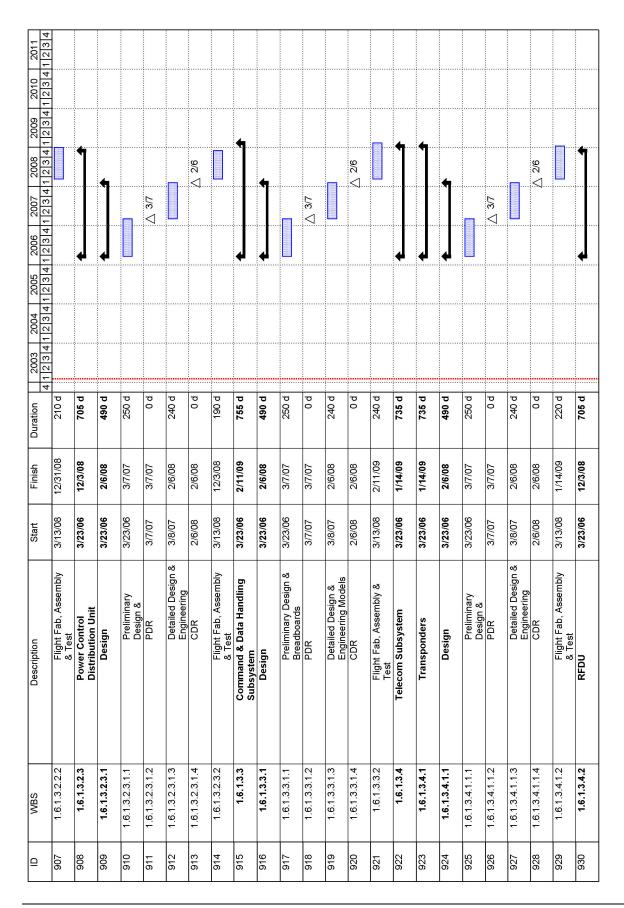
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Finish	3/20/09	60/8/9	60/8//	12/22/09	5/6/09	5/15/09	8/18/09	12/22/09	3/19/10	6/24/09	4/24/09	60/2/9	6/24/09	6/12/09	6/12/09	6/11/09	6/24/09	6/24/09	9/16/09	7/31/09	6/16/09	8/28/09	9/16/09	9/4/09
Start	1/26/09	3/23/09	5/25/09	11/17/08	11/17/08	3/23/09	7/1/09	11/11/09	2/9/09	2/9/09	2/9/09	4/27/09	60/8/9	60/8/9	6/12/09	6/12/09	6/18/09	6/24/09	60/8/9	60/8/9	6/16/09	60/8/8	8/31/09	8/31/09
Description	Optical Assy FM1 I&T	Optical Assy FM2 I&T	Optical Assy FM3 I&T	Optical Assembly Final Integration (Flight GRS)	Optical Assy EM Final I&T	Optical Assy FM1 Final I&T	Optical Assy FM2 Final I&T	Optical Assy FM3 Final I&T	Y-Tube Integration (2 work stations)	Y-Tube EM I&T	L∞	Schedule Reserve	Pack, Ship & Deliver to NASA	Pre-Ship Review Preparations	Pre-Ship Review	Ship to Observatory Contractor	Post-Delivery Inspection & Checkout	Payload Turnover For Observatory	Y-Tube FM1 I&T	T&I	FM1 PER	Schedule Reserve	Pack, Ship & Deliver to NASA	Pre-Ship Review Preparations
WBS	1.5.6.3.2	1.5.6.3.3	1.5.6.3.4	1.5.6.4	1.5.6.4.1	1.5.6.4.2	1.5.6.4.3	1.5.6.4.4	1.5.6.5	1.5.6.5.1	1.5.6.5.1.1	1.5.6.5.1.2	1.5.6.5.1.3	1.5.6.5.1.3.1	1.5.6.5.1.3.2	1.5.6.5.1.3.3	1.5.6.5.1.3.4	1.5.6.5.1.3.5	1.5.6.5.2	1.5.6.5.2.1	1.5.6.5.2.2	1.5.6.5.2.3	1.5.6.5.2.4	1.5.6.5.2.4.1
Ω	811	812	813	814	815	816	817	818	819	820	821	822	823	824	825	826	827	828	829	830	831	832	833	834

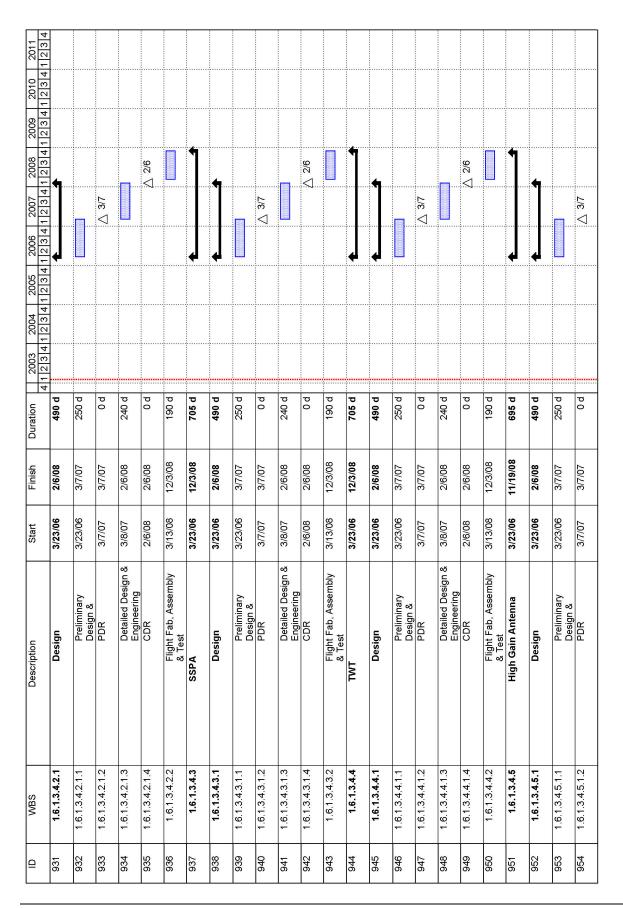
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Finish	9/4/09	60/6/6	9/16/09	9/16/09	12/2/09	10/16/09	60/8/6	11/13/09	12/2/09	11/20/09	11/20/09	11/25/09	12/2/09	12/2/09	3/19/10	2/2/10	12/31/09	3/2/10	3/19/10	3/9/10	3/9/10	3/12/10	3/19/10	3/19/10
Start	9/4/09	60/2/6	9/10/09	9/16/09	8/31/09	8/31/09	60/8/6	10/19/09	11/16/09	11/16/09	11/20/09	11/23/09	11/26/09	12/2/09	12/23/09	12/23/09	12/31/09	2/3/10	3/3/10	3/3/10	3/9/10	3/10/10	3/15/10	3/19/10
Description	Pre-Ship Review	Ship to Observatory Contractor	Post-Delivery Inspection & Checkout	Payload Turnover For Observatory	Y-Tube FM2 I&T	⊢%	PER	Schedule Reserve	Pack, Ship & Deliver to NASA	Pre-Ship Review Preparations	Pre-Ship Review	Ship to Observatory Contractor	Post-Delivery Inspection & Checkout	Payload Turnover For Observatory	Y-Tube FM3 I&T	_জ	PER	Schedule Reserve	Pack, Ship & Deliver to NASA	Pre-Ship Review Preparations	Pre-Ship Review	Ship to Observatory Contractor	Post-Delivery Inspection & Checkout	Payload Turnover For Observatory
WBS	1.5.6.5.2.4.2	1.5.6.5.2.4.3	1.5.6.5.2.4.4	1.5.6.5.2.4.5	1.5.6.5.3	1.5.6.5.3.1	1.5.6.5.3.2	1.5.6.5.3.3	1.5.6.5.3.4	1.5.6.5.3.4.1	1.5.6.5.3.4.2	1.5.6.5.3.4.3	1.5.6.5.3.4.4	1.5.6.5.3.4.5	1.5.6.5.4	1.5.6.5.4.1	1.5.6.5.4.2	1.5.6.5.4.3	1.5.6.5.4.4	1.5.6.5.4.4.1	1.5.6.5.4.4.2	1.5.6.5.4.4.3	1.5.6.5.4.4.4	1.5.6.5.4.4.5
Q	835	836	837	838	839	840	841	842	843	844	845	846	847	848	849	850	851	852	853	854	855	856	857	858

FLIGHT SYSTEM WBS 1.6

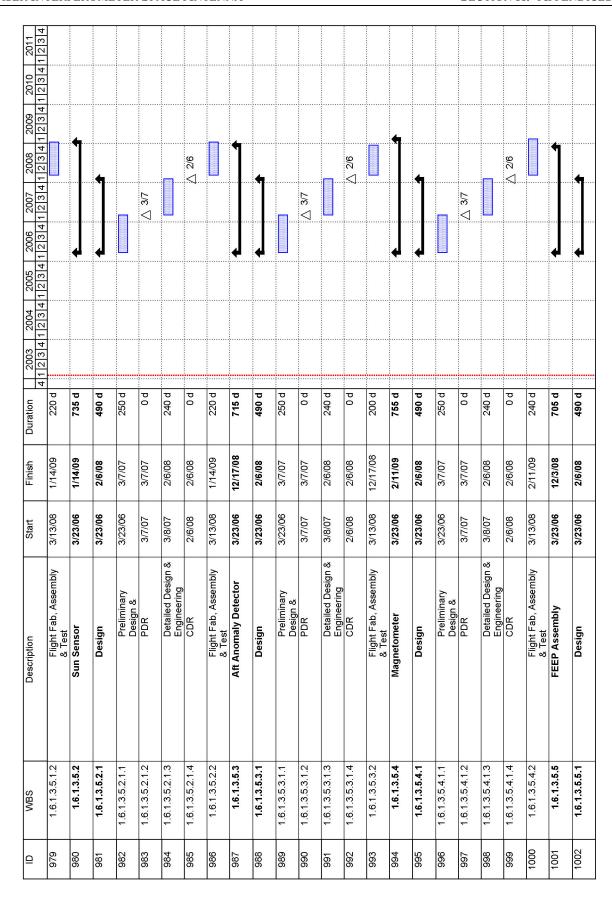
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Duration 4	2054 d	1639 d	646 d	ΡO	132 d	P O	PO	137 d	P O	1394 d	p 06	87 d	88 d	p ₀	p o	P0	PO	P O	920 d	715 d	715 d	490 d	250 d	PO
Finish	8/15/11	1/11/10	3/22/06	10/1/03	4/1/04	4/1/04	9/13/05	3/22/06	3/22/06	60/2/8	8/5/04	12/6/04	4/7/05	4/12/07	3/13/08	2/25/09	5/20/09	8/2/09	60/08/6	12/17/08	12/17/08	2/6/08	3/7/07	3/7/07
Start	10/1/03	10/1/03	10/1/03	10/1/03	10/1/03	4/1/04	9/13/05	9/13/05	3/22/06	4/2/04	4/2/04	8/6/04	12/7/04	4/12/07	3/13/08	2/25/09	5/20/09	8/2/09	3/23/06	3/23/06	3/23/06	3/23/06	3/23/06	3/7/07
Description	Flight System	Spacecraft (ESA)	ESA Acquisition Cycle	Definition Phase ITT	Definition Phase Acquisition Cycle	Definition Phase Kick-off	Implementation Phase ITT	Implementation Phase Acquisition Cycle	Implementation Phase Kick-off	Spacecraft System Engineering / Management	Preliminary Mission Architecture	Spacecraft & Technology Conc.	Mission Definition & Design	Spacecraft PDR	Spacecraft CDR	S/C Bus Flight 1 PER	S/C Bus Flight 2 PER	S/C Bus Flight 3 PER	Spacecraff Bus	Structures / Mechanisms Subsystem	Stucture & Mechanism	Design	Preliminary Design	PDR
WBS	1.6	1.6.1	1.6.1.1	1.6.1.1.1	1.6.1.1.2	1.6.1.1.3	1.6.1.1.4	1.6.1.1.5	1.6.1.1.6	1.6.1.2	1.6.1.2.1	1.6.1.2.2	1.6.1.2.3	1.6.1.2.4	1.6.1.2.5	1.6.1.2.6	1.6.1.2.7	1.6.1.2.8	1.6.1.3	1.6.1.3.1	1.6.1.3.1.1	1.6.1.3.1.1.1	1.6.1.3.1.1.1.1	1.6.1.3.1.1.1.2
□	829	860	861	862	863	864	865	998	867	898	698	870	871	872	873	874	875	876	877	878	879	880	881	882

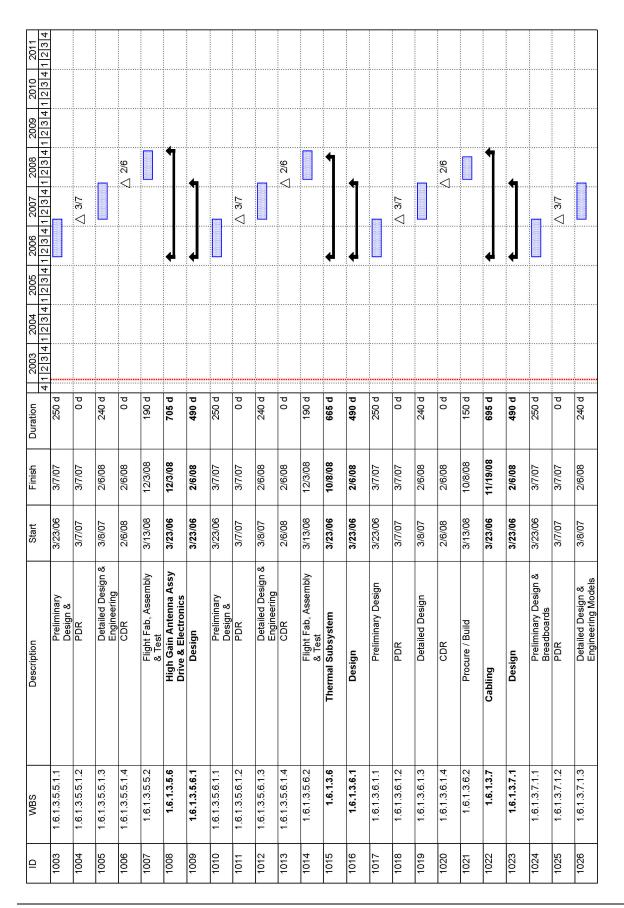




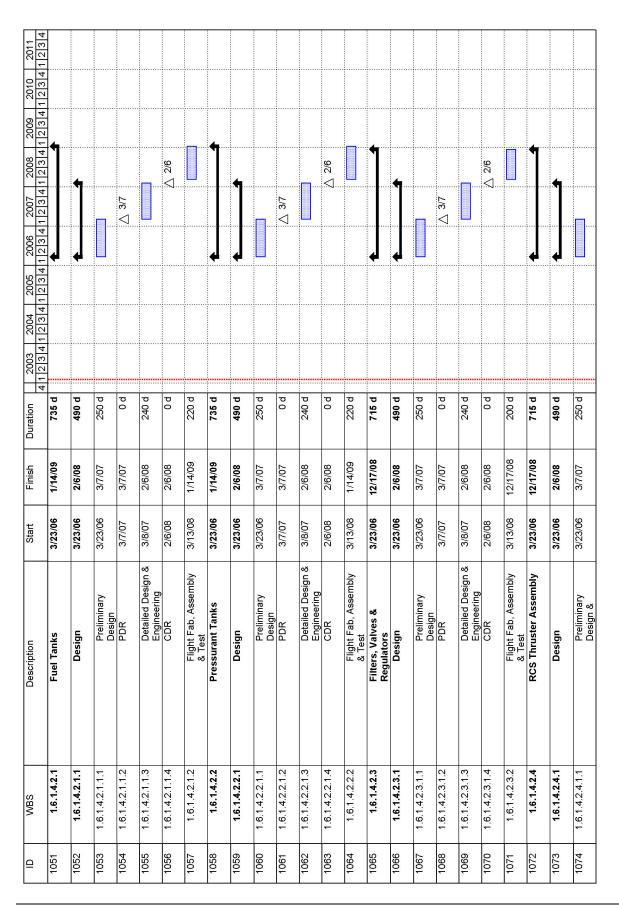


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2003 2004 1 2 3 4 1 2 3 4																								
Duration 4	240 d	p 0	180 d	p 969	490 d	250 d	p 0	240 d	p 0	180 d	p 969	490 d	250 d	p 0	240 d	p 0	180 d	755 d	735 d	490 d	250 d	p 0	240 d	p 0
Finish	2/6/08	2/6/08	11/19/08	11/19/08	2/6/08	3/7/07	3/7/07	2/6/08	2/6/08	11/19/08	11/19/08	2/6/08	3/7/07	3/7/07	2/6/08	2/6/08	11/19/08	2/11/09	1/14/09	2/6/08	3/7/07	3/7/07	2/6/08	2/6/08
Start	3/8/07	2/6/08	3/13/08	3/23/06	3/23/06	3/23/06	3/7/07	3/8/07	2/6/08	3/13/08	3/23/06	3/23/06	3/23/06	3/7/07	3/8/07	2/6/08	3/13/08	3/23/06	3/23/06	3/23/06	3/23/06	3/7/07	3/8/07	2/6/08
Description	Detailed Design & Engineering	CDR	Flight Fab, Assembly & Test	Low Gain Antenna	Design	Preliminary Design &	PDR	Detailed Design & Engineering	CDR	Flight Fab, Assembly & Test	Telecom Cabling Antenna	Design	Preliminary Design &	PDR	Detailed Design & Engineering	CDR	Flight Fab, Assembly & Test	Attitude Control Subsystem	Star Camera Assy Optical Head & Electronics	Design	Preliminary Design &	PDR	Detailed Design & Engineering	CDR
WBS	1.6.1.3.4.5.1.3	1.6.1.3.4.5.1.4	1.6.1.3.4.5.2	1.6.1.3.4.6	1.6.1.3.4.6.1	1.6.1.3.4.6.1.1	1.6.1.3.4.6.1.2	1.6.1.3.4.6.1.3	1.6.1.3.4.6.1.4	1.6.1.3.4.6.2	1.6.1.3.4.7	1.6.1.3.4.7.1	1.6.1.3.4.7.1.1	1.6.1.3.4.7.1.2	1.6.1.3.4.7.1.3	1.6.1.3.4.7.1.4	1.6.1.3.4.7.2	1.6.1.3.5	1.6.1.3.5.1	1.6.1.3.5.1.1	1.6.1.3.5.1.1.1	1.6.1.3.5.1.1.2	1.6.1.3.5.1.1.3	1.6.1.3.5.1.1.4
Ω	922	926	957	928	929	096	961	362	6963	964	3962	996	296	896	696	970	971	972	973	974	975	976	277	978





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	P O	180 d	405 d	p 06	p 09	55 d	20 d	p 006	715 d	715 d	490 d	250 d	p ₀	240 d	p o	200 d	715 d	490 d	250 d	p o	240 d	p 0	200 d	735 d
Duration			4					6		7	4	2		2				4	2		2			7
Finish	2/6/08	11/19/08	60/08/6	7/16/08	60/9/5	7/22/09	60/08/6	9/2/09	12/17/08	12/17/08	2/6/08	3/7/07	3/7/07	2/6/08	2/6/08	12/17/08	12/17/08	2/6/08	3/7/07	3/7/07	2/6/08	2/6/08	12/17/08	1/14/09
Start	2/6/08	3/13/08	3/13/08	3/13/08	2/12/09	60/2/9	7/23/09	3/23/06	3/23/06	3/23/06	3/23/06	3/23/06	3/7/07	3/8/07	2/6/08	3/13/08	3/23/06	3/23/06	3/23/06	3/7/07	3/8/07	2/6/08	3/13/08	3/23/06
Description	CDR	Flight Fab, Assembly & Test	S/C Bus I&T	S/C Bus EM I&T	S/C Bus FM1 I&T	S/C Bus FM2 I&T	S/C Bus FM3 I&T	Propulsion Module	Structure	Primary Stucture	Design	Preliminary Design &	PDR	Detailed Design & Engineering	CDR	Flight Fab, Assembly & Test	Secondary Structure	Design	Preliminary Design &	PDR	Detailed Design & Engineering	CDR	Flight Fab, Assembly & Test	Chemical Propulsion
WBS	1.6.1.3.7.1.4	1.6.1.3.7.2	1.6.1.3.8	1.6.1.3.8.1	1.6.1.3.8.2	1.6.1.3.8.3	1.6.1.3.8.4	1.6.1.4	1.6.1.4.1	1.6.1.4.1.1	1.6.1.4.1.1.1	1.6.1.4.1.1.1	1.6.1.4.1.1.2	1.6.1.4.1.1.3	1.6.1.4.1.1.4	1.6.1.4.1.1.2	1.6.1.4.1.2	1.6.1.4.1.2.1	1.6.1.4.1.2.1.1	1.6.1.4.1.2.1.2	1.6.1.4.1.2.1.3	1.6.1.4.1.2.1.4	1.6.1.4.1.2.2	1.6.1.4.2
Ω	1027	1028	1029	1030	1031	1032	1033	1034	1035	1036	1037	1038	1039	1040	1041	1042	1043	1044	1045	1046	1047	1048	1049	1050



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2003																								
Duration	P O	240 d	P O	200 d	675 d	490 d	250 d	р0	240 d	P 0	160 d	385 d	80 d	p 09	55 d	20 d	388 d	183 d	p 06	80 d	13 d	p 9	P 0	3 d
Finish	3/7/07	2/6/08	2/6/08	12/17/08	10/22/08	2/6/08	3/7/07	3/7/07	2/6/08	2/6/08	10/22/08	9/2/09	7/2/08	4/8/09	6/24/09	9/2/09	1/11/10	3/30/09	11/19/08	3/11/09	3/30/09	3/18/09	3/18/09	3/23/09
Start	3/7/07	3/8/07	2/6/08	3/13/08	3/23/06	3/23/06	3/23/06	3/7/07	3/8/07	2/6/08	3/13/08	3/13/08	3/13/08	1/15/09	4/9/09	6/22/09	7/17/08	7/17/08	7/17/08	11/20/08	3/12/09	3/12/09	3/18/09	3/19/09
Description	PDR	Detailed Design & Engineering	CDR	Flight Fab, Assembly & Test	Heaters	Design	Preliminary Design	PDR	Detailed Design & Engineering	CDR	Flight Fab, Assembly & Test	Integraton & Test	Propulsion Module EM I&T	Propulsion Module FM1	Propulsion Module FM2 I&T	Propulsion Module FM3 I&T	Spacecraft I&T (Bus & Propulsion Module)	Spacecraft EM I&T	S/C Bus-To-Propulsion Module Compatibility	Schedule Reserve	Pack, Ship & Deliver Spacecraft to	Pre-Ship Review Preparations	Pre-Ship Review	Ship to Observatory Contractor
WBS	1.6.1.4.2.4.1.2	1.6.1.4.2.4.1.3	1.6.1.4.2.4.1.4	1.6.1.4.2.4.2	1.6.1.4.2.5	1.6.1.4.2.5.1	1.6.1.4.2.5.1.1	1.6.1.4.2.5.1.2	1.6.1.4.2.5.1.3	1.6.1.4.2.5.1.4	1.6.1.4.2.5.2	1.6.1.4.3	1.6.1.4.3.1	1.6.1.4.3.2	1.6.1.4.3.3	1.6.1.4.3.4	1.6.1.5	1.6.1.5.1	1.6.1.5.1.1	1.6.1.5.1.2	1.6.1.5.1.3	1.6.1.5.1.3.1	1.6.1.5.1.3.2	1.6.1.5.1.3.3
Ω	1075	1076	1077	1078	1079	1080	1081	1082	1083	1084	1085	1086	1087	1088	1089	1090	1091	1092	1093	1094	1095	1096	1097	1098

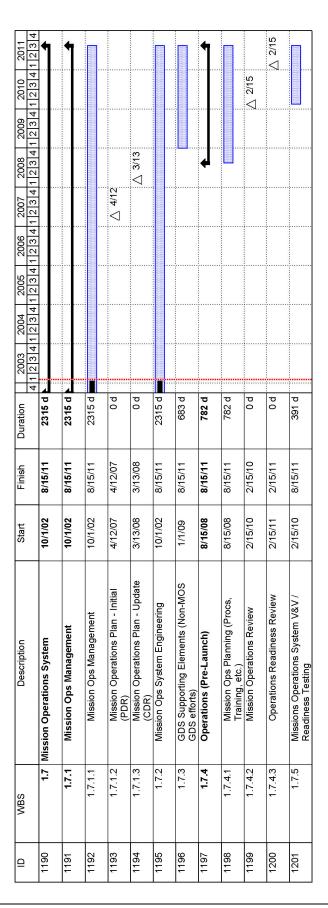
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2003																								
Duration	р	рO	p £2	40 d	20 d	13 d	P 9	ρO	3 d	5 d	P 0	p 89	35 d	20 d	13 d	5 d	ρO	34	54	P 0	P 89	30 d	20 d	13 d
Finish	60/0ɛ/ɛ	3/30/09	8/17/09	7/1/09	7/29/09	8/17/09	8/5/09	60/2/8	8/10/09	8/17/09	8/17/09	11/2/09	9/16/09	10/14/09	11/2/09	10/21/09	10/21/09	10/26/09	11/2/09	11/2/09	1/11/10	11/25/09	12/23/09	1/11/10
Start	3/24/09	3/30/09	60/2/9	60/2/9	7/2/09	7/30/09	60/08//	8/2/09	60/9/8	8/11/09	8/17/09	7/30/09	60/08//	9/17/09	10/15/09	10/15/09	10/21/09	10/22/09	10/27/09	11/2/09	10/15/09	10/15/09	11/26/09	12/24/09
Description	Post-Delivery Inspection & Checkout	Spacecraft Turnover for Observatory I&T	Spacecraft Flight 1 l&T	S/C Bus-To-Propulsion Module Compatibility	Schedule Reserve	Pack, Ship & Deliver Spacecraft to	Pre-Ship Review Preparations	Pre-Ship Review	Ship to Observatory Contractor	Post-Delivery Inspection & Checkout	Spacecraft Turnover for Observatory I&T	Spacecraft Flight 2 I&T	S/C Bus-To-Propulsion Module Compatibility	Schedule Reserve	Pack, Ship & Deliver Spacecraff to	Pre-Ship Review Preparations	Pre-Ship Review	Ship to Observatory Contractor	Post-Delivery Inspection & Checkout	Spacecraft Turnover for Observatory I&T	Spacecraft Flight 3 I&T	S/C Bus-To-Propulsion Module Compatibility	Schedule Reserve	Pack, Ship & Deliver Spacecraft to
WBS	1.6.1.5.1.3.4	1.6.1.5.1.3.5	1.6.1.5.2	1.6.1.5.2.1	1.6.1.5.2.2	1.6.1.5.2.3	1.6.1.5.2.3.1	1.6.1.5.2.3.2	1.6.1.5.2.3.3	1.6.1.5.2.3.4	1.6.1.5.2.3.5	1.6.1.5.3	1.6.1.5.3.1	1.6.1.5.3.2	1.6.1.5.3.3	1.6.1.5.3.3.1	1.6.1.5.3.3.2	1.6.1.5.3.3.3	1.6.1.5.3.3.4	1.6.1.5.3.3.5	1.6.1.5.4	1.6.1.5.4.1	1.6.1.5.4.2	1.6.1.5.4.3
Ω	1099	1100	1101	1102	1103	1104	1105	1106	1107	1108	1109	1110	1111	1112	1113	1114	1115	1116	1117	1118	1119	1120	1121	1122

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Finish	12/30/09	12/30/09	1/4/10	1/11/10	1/11/10	9/3/10	7/2/10	11/14/08	4/13/09	4/20/09	4/27/09	60/8/9	6/22/09	7/22/09	10/14/09	10/21/09	1/13/10	2/12/10	4/23/10	7/2/10	3/19/10	5/18/09	8/31/09	60/06/6
Start	12/24/09	12/30/09	12/31/09	1/5/10	1/11/10	6/2/08	6/2/08	6/2/08	3/31/09	4/14/09	4/21/09	4/28/09	60/6/9	6/22/09	7/23/09	10/15/09	10/22/09	2/1/10	2/15/10	4/26/10	12/2/08	12/2/08	8/18/09	9/17/09
Description	Pre-Ship Review Preparations	Pre-Ship Review	Ship to Observatory Contractor	Post-Delivery Inspection & Checkout		Observatory Integration & Test		Set-Up & Checkout OTS #1	Install EM Spacecraft on OTS #1	Install Payload Structural Thermal Model (STM)	Remove EM Observatory from OTS #1 & Install into Vibration	Structural Dynamics Qualification Test	Re-Install Observatory EM on OTS #1	Install Payload EM	Observatory EM Performance Testing	Remove Observatory EM from OTS-1	Observatory EM Environmental Qual Test	Re-Install Observatory EM on OTS #1	Retrofit and Test Observatory EM for Flight Spare	Spare Observatory Reserve	Observatory Flight 1 (on Observatory Test Stand 2)	Set-Up & Checkout OTS #2	Install Spacecraft Flight 1	Install Payload FM1
WBS	1.6.1.5.4.3.1	1.6.1.5.4.3.2	1.6.1.5.4.3.3	1.6.1.5.4.3.4	1.6.1.5.4.3.5	1.6.2	1.6.2.1	1.6.2.1.1	1.6.2.1.2	1.6.2.1.3	1.6.2.1.4	1.6.2.1.5	1.6.2.1.6	1.6.2.1.7	1.6.2.1.8	1.6.2.1.9	1.6.2.1.10	1.6.2.1.11	1.6.2.1.12	1.6.2.1.13	1.6.2.2	1.6.2.2.1	1.6.2.2.2	1.6.2.2.3
Ω	1123	1124	1125	1126	1127	1128	1129	1130	1131	1132	1133	1134	1135	1136	1137	1138	1139	1140	1141	1142	1143	1144	1145	1146

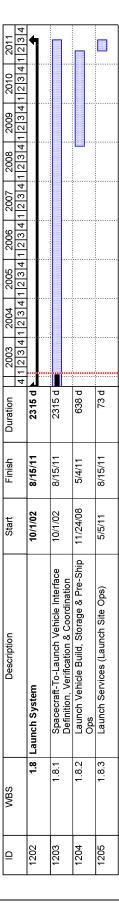
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Finish	11/18/09	10/28/09	11/20/09	1/22/10	3/19/10	3/19/10	5/28/10	11/16/09	12/16/09	1/27/10	1/6/10	1/29/10	4/2/10	5/28/10	5/28/10	9/3/10	1/25/10	4/2/10	5/7/10	4/16/10	7/9/10	9/3/10	9/3/10	12/10/10
Start	10/1/09	10/28/09	11/19/09	11/23/09	1/25/10	3/19/10	11/3/09	11/3/09	12/3/09	12/17/09	1/6/10	1/28/10	2/1/10	4/5/10	5/28/10	1/12/10	1/12/10	3/22/10	4/5/10	4/16/10	5/10/10	7/12/10	9/3/10	8/3/09
Description	Observatory Flight 1 Performance Test	Pre-Environmental Review	Remove Observatory Flight 1 from OTS-2	Observatory Flight 1 Environmental Test	Schedule Reserve	Turnover for Constellation Test	Observatory Flight 2 (on Observatory Test Stand 1)	Install Spacecraft Flight 2 on OTS#1	Install Payload FM2	Observatory Flight 2 Performance Test	Pre-Environmental Review	Remove Observatory Flight 2 from OTS-1	Observatory Flight 2 Environmental Test	Schedule Reserve	Turnover for Constellation Test	Observatory Flight 3 (on Observatory Test Stand 2)	Install Spacecraft Flight 3 on OTS#2	Install Payload FM3	Observatory Flight 3 Performance Test	Pre-Environmental Review	Observatory Flight 3 Environmental Test	Schedule Reserve	Turnover for Constellation Test	Constellation Testing
WBS	1.6.2.2.4	1.6.2.2.5	1.6.2.2.6	1.6.2.2.7	1.6.2.2.8	1.6.2.2.9	1.6.2.3	1.6.2.3.1	1.6.2.3.2	1.6.2.3.3	1.6.2.3.4	1.6.2.3.5	1.6.2.3.6	1.6.2.3.7	1.6.2.3.8	1.6.2.4	1.6.2.4.1	1.6.2.4.2	1.6.2.4.3	1.6.2.4.4	1.6.2.4.5	1.6.2.4.6	1.6.2.4.7	1.6.3
Ω	1147	1148	1149	1150	1151	1152	1153	1154	1155	1156	1157	1158	1159	1160	1161	1162	1163	1164	1165	1166	1167	1168	1169	1170

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Duration 4	100 d	30 d	30 d	30 d	40 d	103 d	34	20 d	40 d	10 d	2 d	PO	2 d	34	PO	73 d	45 d	28 d	P 0
Finish	12/18/09	4/30/10	8/13/10	10/15/10	12/10/10	5/4/11	12/15/10	2/23/11	4/20/11	5/4/11	4/27/11	4/27/11	4/29/11	5/4/11	5/4/11	8/15/11	7/6/11	8/15/11	8/15/11
Start	60/٤/8	3/22/10	7/5/10	9/6/10	10/18/10	12/13/10	12/13/10	12/16/10	2/24/11	4/21/11	4/21/11	4/27/11	4/28/11	5/2/11	5/4/11	5/5/11	5/5/11	7/7/11	8/15/11
Description	Set-Up Constellation Test Facility	Checkout With Single Observatory	Two-Observatory Test	Three-Observatory Test	Schedule Reserve	Final Integration (Pre-Launch)	Install Payload Attachment Fitting	Pre-Ship Functional Testing for Launch Campaign	Schedule Reserve	Pack, Ship & Delivery to Launch Site	Pre-Ship Review Preparations	Pre-Ship Review	Install in Shipping Container	Transport to Launch Site	Constellation Delivery to Launch Site	Launch Campaign	Constellation Launch Site Operations	Constellation / Launch Vehicle Operations	Launch
WBS	1.6.3.1	1.6.3.2	1.6.3.3	1.6.3.4	1.6.3.5	1.6.4	1.6.4.1	1.6.4.2	1.6.4.3	1.6.4.4	1.6.4.4.1	1.6.4.4.2	1.6.4.4.3	1.6.4.4.4	1.6.4.4.5	1.6.5	1.6.5.1	1.6.5.2	1.6.5.3
Ω	1171	1172	1173	1174	1175	1176	1177	1178	1179	1180	1181	1182	1183	1184	1185	1186	1187	1188	1189

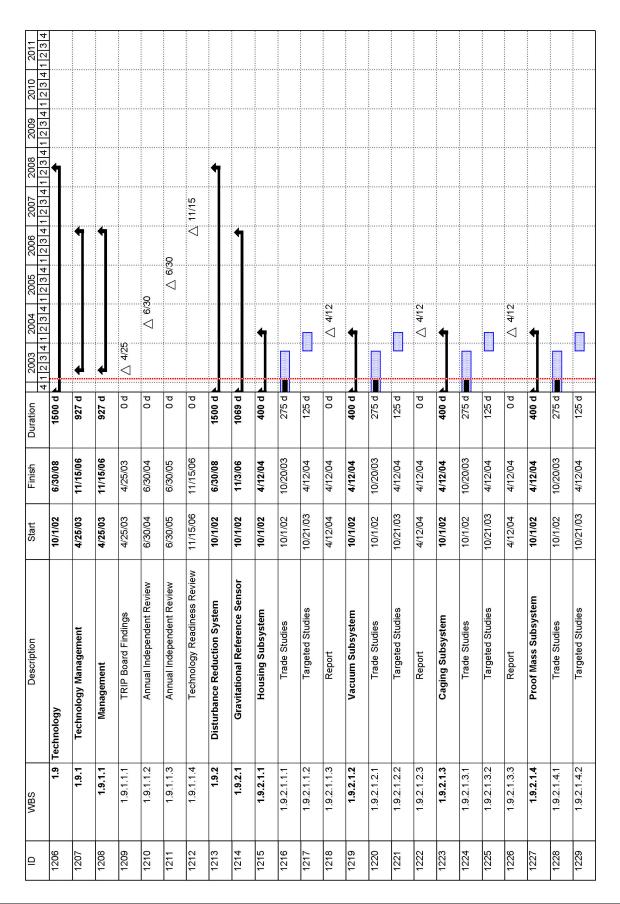
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LAUNCH SYSTEM WBS 1.8

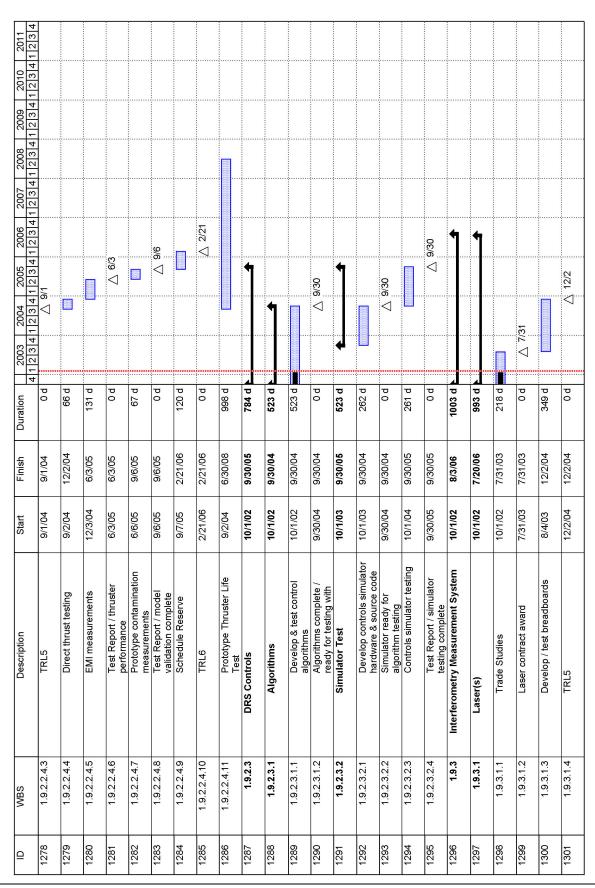


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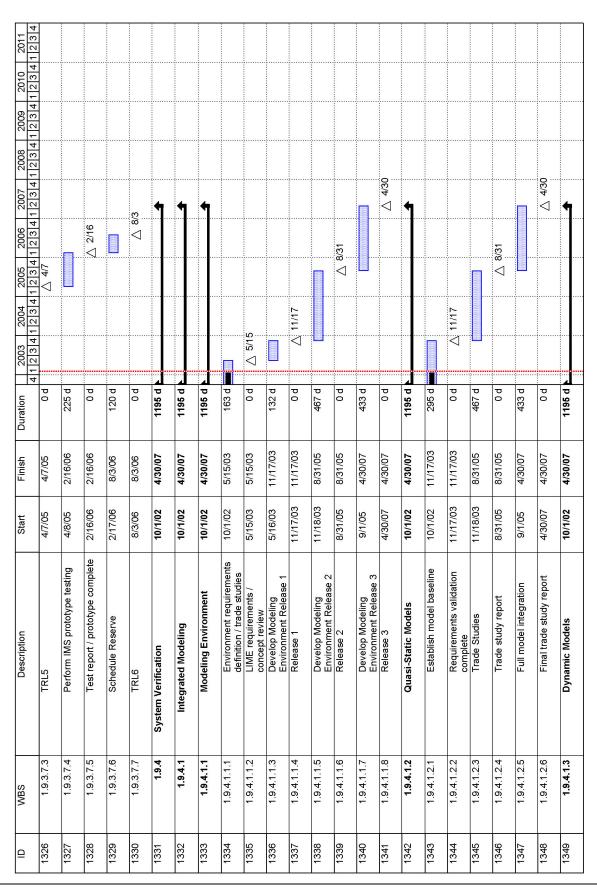


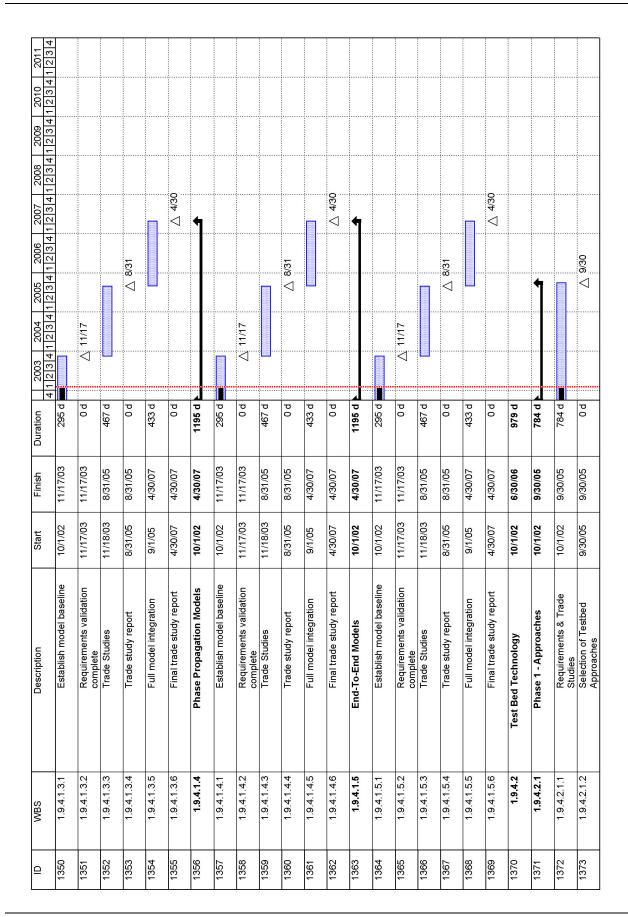
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Duration 4	p ₀	400 d	275 d	125 d	p 0	400 d	275 d	125 d	P 0	400 d	275 d	125 d	p 0	400 d	275 d	125 d	PO	p 699	150 d	168 d	P O	238 d	P O	113 d
Finish	4/12/04	4/12/04	10/20/03	4/12/04	4/12/04	4/12/04	10/20/03	4/12/04	4/12/04	4/12/04	10/20/03	4/12/04	4/12/04	4/12/04	10/20/03	4/12/04	4/12/04	11/3/06	11/8/04	6/30/05	6/30/05	90/06/5	90/06/9	11/3/06
Start	4/12/04	10/1/02	10/1/02	10/21/03	4/12/04	10/1/02	10/1/02	10/21/03	4/12/04	10/1/02	10/1/02	10/21/03	4/12/04	10/1/02	10/1/02	10/21/03	4/12/04	4/13/04	4/13/04	11/9/04	9/30/02	7/1/05	90/08/9	5/31/06
Description	Report	Proof Mass Control Subsystem	Trade Studies	Targeted Studies	Report	Sensing & Forcing Subsystem	Trade Studies	Targeted Studies	Report	Charge Control Subsystem	Trade Studies	Targeted Studies	Report	I&T Mechanisms	Trade Studies	Targeted Studies	Report	Prototype GRS (Lab Verified)	Prototype subsystem design & build	Prototype assembly & integraton	TRL5	Perform GRS prototype testing	Test report / prototype complete	Schedule Reserve
WBS	1.9.2.1.4.3	1.9.2.1.5	1.9.2.1.5.1	1.9.2.1.5.2	1.9.2.1.5.3	1.9.2.1.6	1.9.2.1.6.1	1.9.2.1.6.2	1.9.2.1.6.3	1.9.2.1.7	1.9.2.1.7.1	1.9.2.1.7.2	1.9.2.1.7.3	1.9.2.1.8	1.9.2.1.8.1	1.9.2.1.8.2	1.9.2.1.8.3	1.9.2.1.9	1.9.2.1.9.1	1.9.2.1.9.2	1.9.2.1.9.3	1.9.2.1.9.4	1.9.2.1.9.5	1.9.2.1.9.6
Ω	1230	1231	1232	1233	1234	1235	1236	1237	1238	1239	1240	1241	1242	1243	1244	1245	1246	1247	1248	1249	1250	1251	1252	1253

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Finish	11/3/06	80/08/9	9/1/04	6/17/03	6/17/03	9/18/03	9/18/03	9/1/04	2/27/04	6/2/03	6/2/03	8/4/03	8/4/03	11/27/03	2/27/04	2/27/04	2/4/04	12/3/03	12/3/03	2/4/04	2/4/04	80/08/9	9/1/04	9/1/04
Start	11/3/06	10/1/02	10/1/02	10/1/02	6/17/03	6/18/03	9/18/03	6/18/03	10/1/02	10/1/02	6/2/03	80/8/9	8/4/03	9/19/03	11/28/03	2/27/04	6/3/03	6/3/03	12/3/03	12/4/03	2/4/04	3/1/04	3/1/04	9/1/04
Description	TRL6	Thrusters	Emitter	Develop Emitter prototypes / initial testing	Prototype delivery (final configuration)	Prototype direct thrust testing	Test report / prototype ready for integration	Emitter prototype life testing	Neutralizer	Neutralizer trade studies	Neutralizer selection	Prototype build / test	Prototype delivery	Integrate Neutralizers with Emitter prototype	Direct thrust testing	Test report / prototype complete	Thruster Electronics	Develop Power Processing Unit (PPU) prototypes	PPUs ready for testing	PPU Testing	Test Report / PPUs ready Thruster prototype	Prototype Thruster	Integrate & assembly prototype thruster emitters,	Prototype thrusters ready for testing
WBS	1.9.2.1.9.7	1.9.2.2	1.9.2.2.1	1.9.2.2.1.1	1.9.2.2.1.2	1.9.2.2.1.3	1.9.2.2.1.4	1.9.2.2.1.5	1.9.2.2.2	1.9.2.2.2.1	1.9.2.2.2.2	1.9.2.2.2.3	1.9.2.2.2.4	1.9.2.2.2.5	1.9.2.2.2.6	1.9.2.2.2.7	1.9.2.2.3	1.9.2.2.3.1	1.9.2.2.3.2	1.9.2.2.3.3	1.9.2.2.3.4	1.9.2.2.4	1.9.2.2.4.1	1.9.2.2.4.2
Ω	1254	1255	1256	1257	1258	1259	1260	1261	1262	1263	1264	1265	1266	1267	1268	1269	1270	1271	1272	1273	1274	1275	1276	1277



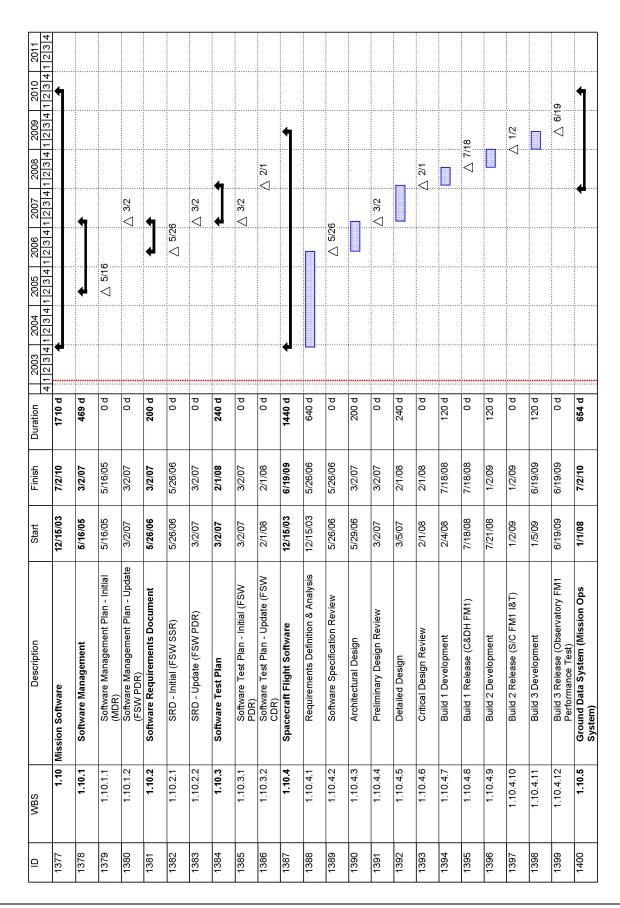
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Duration 4	305 d	120 d	p o	4 p 602	239 d	350 d	120 d	P 0	370 d ×	370 d	P O	370 d A	370 d	p o	370 d F	370 d	p o	400 d	200 d	200 d	p o	633 d	120 d	168 d
Finish	2/2/06	7/20/06	7/20/06	6/17/05	8/29/03	12/31/04	6/17/05	6/17/05	3/1/04	3/1/04	3/1/04	3/1/04	3/1/04	3/1/04	3/1/04	3/1/04	3/1/04	12/10/04	3/5/04	12/10/04	12/10/04	90/8/8	8/16/04	4/7/05
Start	12/3/04	2/3/06	7/20/06	10/1/02	10/1/02	9/1/03	1/3/05	6/17/05	10/1/02	10/1/02	3/1/04	10/1/02	10/1/02	3/1/04	10/1/02	10/1/02	3/1/04	6/2/03	6/2/03	3/8/04	12/10/04	3/2/04	3/2/04	8/17/04
Description	Develop / test prototypes	Schedule Reserve	TRL6	Laser/Clock Noise Cancellation	Develop noise cancellation algorithms	Test algorithms	Schedule Reserve	Test report / algorithms selected	Stabilization	Breadboards	Report/Select	Phase Meter	Breadboards	Report/Select	Ultra Stable Structures	Test	Report	IMS Tester	Develop	Build	Tester Ready	Prototype IMS	Prototype subsystem design & build	Prototype assembly & integraton
WBS	1.9.3.1.5	1.9.3.1.6	1.9.3.1.7	1.9.3.2	1.9.3.2.1	1.9.3.2.2	1.9.3.2.3	1.9.3.2.4	1.9.3.3	1.9.3.3.1	1.9.3.3.2	1.9.3.4	1.9.3.4.1	1.9.3.4.2	1.9.3.5	1.9.3.5.1	1.9.3.5.2	1.9.3.6	1.9.3.6.1	1.9.3.6.2	1.9.3.6.3	1.9.3.7	1.9.3.7.1	1.9.3.7.2
0	1302	1303	1304	1305	1306	1307	1308	1309	1310	1311	1312	1313	1314	1315	1316	1317	1318	1319	1320	1321	1322	1323	1324	1325

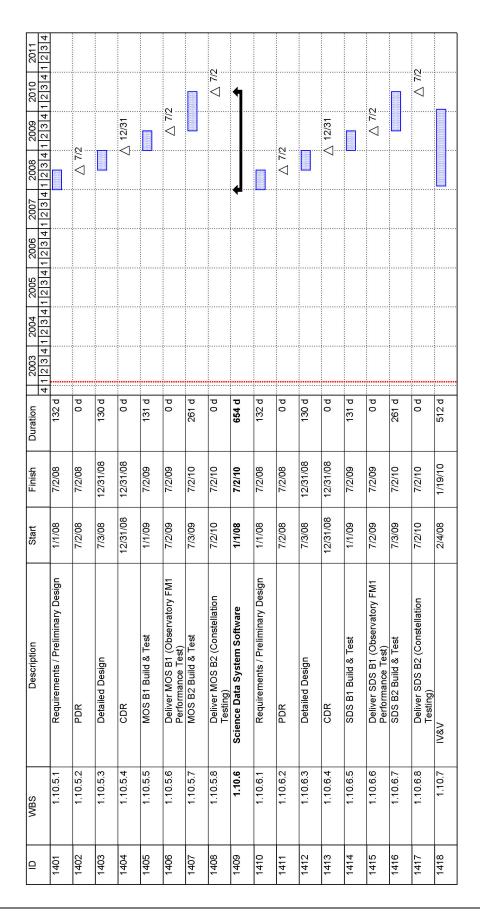




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MISSION SOFTWARE WBS 1.10





H.3 Draft International Participation Plan

H.3.1 Summary

The formulation and implementation activities of the Laser Interferometer Space Antenna (LISA) project, part of the Structure and Evolution of the Universe (SEU) Program office, will require the transfer of hardware, software and technical data to support the relationships as presented in the International Contribution Summary Matrix. This matrix establishes the basis for the eventual development of the LISA Memorandum of Understanding between NASA and the European Space Agency (ESA).

It should be noted that the LISA Program is managed out of the Goddard Space Flight Center (GSFC) for NASA, and that the LISA Project is co-jointly managed by the GSFC and the Jet Propulsion Laboratories (JPL). This arrangement will entail careful coordination between the two centers in terms of the exportation and importation of hardware, software and technical data (please refer to Addendum E for further details).

For further information on the LISA project please refer to the LISA Program Plan, the LISA Project Plan, or the following LISA web sites:

NASA http://universe.gsfc.nasa.gov/

• GSFC http://lisa.gsfc.nasa.gov/

• JPL http://lisa.jpl.nasa.gov/

• ESA http://sci.esa.int/home/lisa/

H.3.2 General Overview

H.3.2.1 International Traffic In Arms Regulations

As NASA is to plan and develop the LISA project in collaboration with ESA, the LISA development activities as well as all requirements to export hardware, software and technology will fall under the jurisdiction of the United States State Department. Most requirements to export

will fall under the State Department International Traffic in Arms Regulations (ITAR). As a general rule, any technical data regarding LISA that exceeds general purpose, marketing level information or information that has been previously placed into the public domain is subject to control under the ITAR. As a result, NASA needs to work under the authority of a Letter of Agreement (LOA) and eventually a Memorandum of Understanding (MOU) to support activities on LISA. The existence of this LOA and MOU will support the use of appropriate ITAR exemptions and support the acquisition of any State Department export licenses.

H.3.2.2 Letter of Agreement

NASA is currently operating under the framework of an LOA negotiated independently between NASA and ESA (Addendum A). The LOA between NASA and ESA became effective June 11, 2001. The agreement generally covers the formulation activities between NASA and ESA. As compared to the eventual MOU to be negotiated between NASA and ESA, the LOA is broad in scope and by its nature, absent of defined responsibilities of the individual parties. This tends to allow relatively *broad* discussions between NASA and the desired foreign party(s). LOAs generally do not go through the Circular 175 review process at the State Department before implementation.

H.3.2.3 Memorandum of Understanding

Eventually, NASA will be operating under the authority and framework of an MOU to be negotiated independently between NASA and ESA (Addendum B). The MOU will cover the implementation activities of the LISA program. Within the responsibilities section of the MOU, NASA will have to clearly and thoroughly define the actions to be performed by each party in order to conduct the LISA project, along with the benefits and rights accruing to each side. As the MOU will be used to provide authority for the eventual use of ITAR exemptions by NASA, as well as will serve as reference to industry Technical Assistance Agreement

(TAA) applications, complete and accurate definition of the responsibilities of the respective parties is critical. *Providing support (parts, assistance) to the foreign party outside the obligations as listed in the responsibilities portion of the agreement will require amendment to the original MOU or a separate agreement.*

H.3.3 Stages

H.3.3.1 Formulation

Discussions in the formulation stage tend to be broad, conceptual, generally absent of actual applied engineering and design information. The LOA(s) are broad in scope and by their nature, absent of defined responsibilities of the individual parties. This scenario generally affords NASA the ability to conduct, relatively *broad* discussions between NASA and the desired foreign party(s). As always, exchange of any NASA technical data must satisfy a "need to know" argument within the scope of the activity.

H.3.3.2 Implementation

By contrast to the formulation stage activities, exchange of technical information within the implementation stage must be more deliberate, supported by the responsibilities in the MOUs between NASA and the foreign partners and by available ITAR exemptions. Exchange of technical data will involve actual interface information and actual applied engineering information to be conducted in design reviews and technical interchange meetings. On occasion, there may be a need to partition technical meetings so that foreign parties are excluded from sessions of technical meetings where subject matter is to be discussed which exceeds what is necessary for the foreign party(s) to satisfy their responsibilities under the MOU. As always, NASA must ensure that the "need to know" argument is satisfied, within the responsibilities of the MOU.

H.3.4 LISA Industry Prime / Technical Assistance Agreements/ Non-Disclosure Agreements

H.3.4.1 Technical Assistance Agreements

GSFC anticipates that the selected prime integration contractor will choose to obtain a State Department approved TAA to support the providing of technical data and technical assistance to ESA and their eventual related entities within the LISA implementation phase. Any competing prime contractor will have to obtain a TAA to support its activities within the formulation phase. The TAA may also include all U.S. entities to be involved with the prime in directly participating in the LISA activities. In addition to possibly requesting signature on the TAA from ESA, the selected prime might also request that all foreign entities actively participating in the LISA program become signatory to the TAA. Active participation would generally mean participation in technical meetings, access to the U.S. prime contractor's facilities and participation in any portion of program integration and testing efforts.

The completion of TAA(s) has proven to be a long lead-time item on other NASA programs, often causing delays in critical activities. Delays are caused by the requirements to amend existing TAA(s) to add domestic or foreign entities, as well as reluctance by the foreign partners to sign the TAA(s) in many cases. TAA preparation time should be factored into program schedules because TAA completion will critically affect the program.

H.3.4.2 Non-Disclosure Agreements

In addition to signing the U.S. prime contractor TAA, the U.S. State Department within the TAA approval process requires all parties whom are signatory to the prime TAA to sign a non-disclosure agreement.

All additional parties who obtain ITAR sensitive technical information directly from the selected U.S. prime, but are not actively involved in the program (i.e. not involved in integration and testing, participating actively in technical meetings) may be requested to

complete a non-disclosure agreement before the data will be exported to them.

H.3.5 Technology Transfer (Written Technical Data and Software)

H.3.5.1 Technical Data

As was stated previously, NASA generally can share ITAR controlled, technical data with the foreign partners (and related entities) by authority of the LOA and MOU, utilizing exemptions to the ITAR (22 CFR 126.4 / 22 CFR 125.4 (b)(3). The technical data to be transferred must be supported by the requirements and responsibilities section in the LOA/MOU, and must not be specifically excluded by the Transfer of Technical Data and Goods section in the LOA/MOU.

H.3.5.2 NASA Provided Software

NASA provided software will generally be classified as technical data, when transferred to foreign partners and related entities. When embedded into ground support equipment or flight hardware, the software will classified for export consistent with the modification of the higher hardware assembly. As most NASA software will be treated for export as if it were other technical (written) data, there is a requirement to mark the software to limit dissemination.

If NASA needs to transfer encrypted software and/or source code, these requirements are handled as special cases, as special provisions may be required to include special licensing.

H.3.5.3 Destination Control Statement

However, as NASA is forwarding technical data under specific exemption from licensing, the receiving foreign partner (and/or related entity(s)) need to know that subject data is to be protected, in accordance with the *Transfer of Technical Data and Goods clauses of the LOA(s)/MOU(s)*. The clause that NASA needs to place (usually as a footer) on ITAR sensitive, technical data

provided to the NASA foreign partner is as follows:

These items are licensed by the United States for ultimate destination to (European Space Agency)(insert which partner applies). Diversion, retransfer, disclosure, or use contrary to applications specified in NASA agreement without prior U.S. authorization are prohibited.

If NASA does not mark the data as indicated above, the foreign partner cannot protect the data, and therefore is under no obligation to protect the NASA sensitive data. The U.S. industry competing primes have proven to be proficient at marking their data at all times.

H.3.5.4 Public Release of NASA Technical Data / Software

If there is a desire to release LISA ITAR sensitive technical data into a public forum, i.e. release through presentation and/or publication, the NASA STI process should be utilized. The originator of the requirement should follow the process contained in NPD 2220.5E/NPG 2200.2A, utilizing NASA form 1676.

Public release of NASA software could be completed utilizing the process associated with NPD 2210.1/NPG 2210.1, NASA form 1679.

H.3.6 LISA Customs/Transportation Issues (Hardware)

H.3.6.1 Temporary Exports of NASA Hardware

Under most conditions, ITAR controlled NASA hardware can be exported to a location outside of the United States without a "paper license" utilizing Government ITAR exemption 22 CFR 126.4. Generally, the following conditions **must** be satisfied:

1) The hardware must be exported to meet stated NASA responsibilities in the MOU,

2) the hardware is to remain under NASA ownership and 3) the hardware is to be returned to the United States within 4 years.

U.S. industry must have a "paper license" through the State Department to temporarily export NASA items from the United States.

As NASA can export ITAR controlled hardware to a foreign partner without a license, it is highly desirable that NASA be established as the "exporter of record". As obtaining an export license from the State Department can be a lengthy process and is not friendly to changes in shipping lists, destinations, values, etc., the Government exemption *affords* NASA the flexibility to modify shipping lists and shipment characteristics up to the time of shipment.

Provisions will need to be written into the individual NASA contracts with specific contract language that supports the notion that NASA and not the NASA contractors will export hardware to LISA foreign partners and related entities.

H.3.6.2 Permanent Exports of NASA Hardware

ITAR controlled NASA hardware can be exported to a location outside of the United States with an export license, a process initiated by utilizing State Dept. form DSP-5. Generally, the following conditions **must** be satisfied: 1) The hardware must be exported to meet stated NASA responsibilities in the MOU, 2) the hardware is to remain under NASA ownership and 3) the hardware is to remain outside of the United States for greater than 4 years.

Hardware that is to launch on a foreign launch vehicle will always require a State Department export license.

H.3.6.3 Return of NASA hardware from foreign partner facilities

As most of the hardware that will be exported to the LISA foreign collaborators is ITAR-controlled, it is critical that the foreign partner return of such hardware is coordinated through the project and the NASA GSFC Logistics Management Division.

NASA hardware is returned duty free from foreign partner and related entity locations outside of the United States.

NASA will pay freight charges and handling fees assessed in the United States for all NASA shipments to foreign locations. NASA will pay all freight charges and handling fees to return all NASA hardware items from the foreign partners and related entities locations. The LISA MOU will contain a general obligation to facilitate free customs clearance (e.g. waiver of applicable duties or taxes) for entrances to, and exits from, each side's respective country(s) for material required for LISA.

H.3.6.4 NASA Importation of ESA Hardware

According to Government ITAR exemption 22 CFR 126.4: "A license is not required for the temporary import, or temporary export, of any defense article, including technical data, by or for any agency of the U.S. Government (1) for official use by such an agency, or (2) for carrying out any foreign assistance, *cooperative project* or sales program authorized by law and subject to control by the President by other means. This exemption applies only when all aspects of a transaction are affected by a United States Government agency or when the export is covered by a United States Government Bill of Lading.

Therefore, ESA hardware, flight and non-flight, can be imported into the United States duty free utilizing the NASA Import Certification Process in accordance with 14 CFR Part 1217 (see http://www.hq.nasa.gov/office/codei/nasaec p/ecpmain.html.)

This process requires that a blanket, LISA NASA Import Certification be generated by Code *IS* in parallel with the development of the associated Memorandum of Understanding. When individual imports are anticipated, the LISA Program Manager must verify in writing that specifically identified articles to be entered on a particular date are the articles described in the LISA Importation. No additional paperwork requirements, i.e. Customs Form CF7501 should be required.

In addition to this requirement, the U.S. industry destination will most likely require that NASA generate a NASA GSFC 20-4 or equivalent DD1149 so that the ESA items can be provided as Government Furnished Equipment.

H.3.6.5 NASA Importation of LISA Items Procured Outside of the United States

Under the authority of ITAR 123.4 (a) NASA is allowed to: "temporary import (and subsequent export) without a license, for a period of up to four years, of unclassified U.S.-origin defense items (including any items manufactured abroad pursuant to U.S. Government approval) . . ."

Additionally, items procured outside of the United can be imported duty free into the United States utilizing the NASA Import Certification Process in accordance with 14 CFR Part 1217.

This process requires that an Importation Certification be prepared and endorsed by the NASA HQ or GSFC Procurement Officer. In addition, a Customs Form CF7501 (Entry Summary) will need to be generated by the GSFC Logistics Management Division, Transportation Branch

H.3.6.6 Hand-carry of NASA Hardware from the United States

The hand-carry of NASA hardware to include NASA computers, from the United States needs to be planned and executed, as if it were to be transported via a commercial carrier to a foreign location. Severe fines and penalties can be levied to the traveler if improper procedures are used to hand-carry NASA hardware from the United States, particularly items that are controlled by the ITAR. Severe fines and penalties can also result if ITAR controlled items are improperly brought back into the United States as well.

Generally it is best if this activity is performed by a civil servant. Special provisions will need to be developed if a contract employee is to hand-carry an ITAR controlled NASA item from the United States, in support of the LISA mission.

H.3.7 Foreign Visitor Processing

H.3.7.1 Short-Term Visitors

The LISA Project will arrange for the necessary visitor documentation to support short term visits by members of ESA (and related entities), using license exemptions covered by the authority of the LISA LOA(s)/MOU(s). Foreign nationals under the employ of a US contractor will also be cleared using the following process; however, the US contractor is responsible for obtaining any State Department licensing or TAAs as authority to access GSFC.

All short-term visitors to GSFC who are not permanent US residents or US citizens must be approved through NASA's Foreign National Management System (NFNMS). This process requires the completion of a Foreign Visitor Information Sheet. This profile sheet should only need to be completed once as a permanent record of one's citizenship, passport number, and other pertinent information.

In addition, the NASA sponsor of the foreign national visit needs to fill out at least two additional forms for the Goddard Space Flight Center: Information Concerning Foreign National's Visit to GSFC and Access For Foreign National to GSFC, which requires the GSFC sponsor to profile the particulars of the GSFC visit (i.e. location, sensitivity, names of visitors, if export sensitive, the legal authority for visit, etc.). In addition, the foreign nationals will require escorting by a GSFC sponsor, which requires (insert the name of the form).

These forms must be submitted by fax to the GSFC International Visits Coordinators Office (Code 100) for processing. Visits from countries of non-concern or non-proscribed countries typically require 10 business days to process. Visit requests from individuals from proscribed countries take a minimum of 30 days.

Visit requests are reviewed by the appropriate Security and Export Control activities at GSFC.

Visit requests may be forwarded to NASA HQ, such as requests from residents of proscribed countries.

H.3.7.2 Long-term Visitors (GSFC Badged Visitors)

The LISA Project will arrange for visitor documentation to support long-term visits by members of ESA (and related entities), using license exemptions covered by authority of the LISA LOAs/MOUs. Foreign nationals under the employ of a US contractor will also be cleared using the same process; however, the US contractor is responsible for obtaining any State Department licensing or TAAs as authority to support this access to GSFC.

LISA can arrange for long-term visitors to the GSFC who are not permanent US residents or US citizens to receive a GSFC picture badge after special approvals. Such approvals are currently under review. The long-term visitor must access GSFC on average three days a week to be eligible for a picture badge. As in the case of short-term access to GSFC, long-term, badged visitors must be approved through the NFNMS. Long-term, badged visitors may also be required to be approved through a National Agency Check (NAC) process or equivalent.

H.3.8 Facilities

H.3.8.1 Office Space

Tentatively, based upon the level of understanding reached in the MOU, office space can be made available based upon the availability of office space. It is assumed that office space would be required primarily for the period of Integration and Test (see Appendices C & D for milestones and expected dates).

H.3.8.2 Computer Access

According to NPG 2810.1, Chapter 4, Section 4.5.5 (with concurrence from Mr.

Henry Middelton, GSFC Information Technology (IT) Security Officer) the following information is currently the standard operating procedure in reference to computer access for Non-NASA employees/foreign visitors.

"Special Considerations for Non-NASA Employees or Contractors." If a user is not a NASA employee or contractor, the following considerations apply for security investigations:

- Remote users other than NASA employees or contractors a current Federal investigation performed by the user's parent organization is acceptable if privileged access to a NASA computer system is required. If a remote user requires privileged or limited privilege access and has not been investigated by the user's parent organization, the user's NASA sponsor must request an investigation.
- International partners international partners who seek access to U. S. Government IT resources, pursuant to an international agreement, must still be investigated. NASA Headquarters handles all investigations for representatives of foreign governments. Contact your CCS for more information."

H.3.9 Configuration Management/ Documentation Management

Need to develop a process to control access to ITAR sensitive/otherwise sensitive controlled documents. This process should be delineated in the LISA Program/Project's Configuration Management Plan(s) and referenced here.

Addendum H.3

Summary of ITAR Requirements for GSFC and JPL Guidance from GSFC's Export Control Office, Odell Young, GSFC Import/Export Specialist:

Exports shall be in compliance with applicable U.S. export laws and regulations and NASA's Export Control Program. A formal memorandum of understanding (MOU) or a letter of agreement (LOA) between NASA and the foreign parties detailing the specific roles and responsibilities of the parties will be required prior to exports.

The Department of State, administers the International Traffic in Arms Regulations (ITAR 22CFR 120-130) which governs the export of articles on the U.S. Munitions List (USML). Category XV of the USML controls spacecraft and associated equipment including related technical data and defense services as well as detailed design, development, manufacturing or production data for all spacecraft and specifically designed or modified components for all spacecraft systems.

Category XV Items under the LISA program will be exported in accordance with the below listed guidelines.

- LI. Permanent Exports: The export of hardware and technical data by NASA which will not be returned to the USA will require a U.S. Dept. of State Export License, DSP-5. If required for this Project, an estimated processing time of eight months will be included in the Project schedule.
- LI. License Exemptions: NASA is authorized to use ITAR licensing exemption 22CFR126.4(a)(2), for the temporary import, or temporary export, of defense articles or the performance of a defense service.

NASA may use ITAR licensing exemption 22CFR125.4(b)(3), for the export of technical data in furtherance of a contract requiring export of the data, and such data does not disclose the details of design, development, production, or manufacture of any defense article.

When applicable, ITAR exemptions to licensing requirements available to U.S. Government agencies will be used.

A determination must be made as to which licensing authority has responsibility for the effort under the proposal. The GSFC Export Control Office will assist Program representatives and NASA Headquarters Export Control Office in obtaining all necessary licensing. It is estimated that export license applications will be approved within six to eight months after submission.

All proposals or announcements of opportunity involving international participation should include clause 1852.225-70 Export Licenses.

Guidance from the JPL Office of Legislative and International Affairs, Stephanie Lear, Import/Export Administrator

Technical data – 125.4(b)(3) Exemption: Once an international agreement is in place, JPL may transfer interface technical data to NASA's foreign partner for use on a specific cooperative project.

Hardware – 126.4(a)(2) Exemption: JPL is not permitted to use the 126.4(a)(2) government exemption. Although, the NASA Management Office located at JPL may use this exemption to temporarily transfer ITAR controlled hardware to NASA's foreign partner without a license. Additionally, hardware exported using this exemption may not be hand carried by a JPL employee.

A State Department export license is required for all ITAR controlled hardware that will be permanently shipped out of the US. NASA will be the applicant for the license as it is their hardware.

H.4 Outline of Technical Responsibilities for International Participation

See LISA Partnership Pathfinder document following.

H-137



LASER INTERFEROMETER SPACE ANTENNA

Laser Interferometer Space Antenna (LISA)

PARTNERSHIP PATHFINDER

January 24, 2003





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1. INTRODUCTION

The Laser Interferometer Space Antenna (LISA) Mission represents a partnership between the European Space Agency (ESA) and the U.S. National Aeronautics and Space Administration (NASA) to develop a space mission capable of directly detecting gravitational waves as predicted by Einstein's General Theory of Relativity. To date this has never been accomplished although there is diverse indirect evidence of the existence of gravitational waves. The LISA mission will consist of three identical spacecraft orbiting about the Sun at one AU in a triangular formation providing a space-based laser interferometer to measure the very small dimensional changes that are caused by passing gravitational waves. These dimensional changes are on the order of one part in 10²¹ and, as such, their detection represents a profound technical challenge. The LISA Project is scientifically important in that it not only enables the direct study of strong gravitational interactions associated with black holes and probes Einstein's theory to new depths, but also provides a wholly new way of viewing the Universe outside the realm of electromagnetic radiation.

This document is developed entirely within the LISA Project to record the continuing definition of this NASA/ESA partnership to implement the LISA Mission. It will be periodically updated as the definition of the partnership evolves and it will serve to gather the technical content for the preparation of a future Memorandum of Understanding (MOU) between NASA and ESA Headquarters. This document is not intended to be an International Agreement and should not be construed as such. NASA and ESA Headquarters will subsequently negotiate a formal MOU prior to the Confirmation Review. It is, however, an agreement between the current NASA and ESA managers of the LISA Project as to their contemporary understanding of this partnership.

A Letter of Agreement is presently in effect that provides a framework for NASA and ESA to define the LISA Mission and the associated preparation of the subsequent MOU.

ESA and NASA will jointly cooperate in the development of the LISA Project with each partner seeking to contribute an approximately equal share. It is acknowledged that the technical complexity of the LISA Mission combined with the equal sharing of responsibilities will require a greater level of cooperation than has existed in earlier ESA/NASA cooperative space ventures. The LISA Mission will be implemented as a "sciencecraft," blurring the traditional distinction between the payload and the spacecraft and, consequently, technical insight must be widely shared to achieve and maintain a coherent design throughout development.

This initial version of the Partnership Pathfinder is intended to describe the key management responsibilities that constitute the LISA Project as they are presently understood and agreed upon by the current NASA and ESA managers of the LISA Project.

The LISA organizational approach is designed to capture the best talent in Europe and in the U.S. based on a NASA/ESA partnership that has naturally evolved from a decade-long collaboration between U.S. and European scientists dedicated to the definition of the LISA Mission. As a consequence of this collaboration, LISA represents a mature, well-studied mission concept as reflected in the Final Test Report published by ESA in April 2000. The scientists involved in the LISA Project are associated with many of the leading universities in the U.S. and in Europe. A number of these scientists are also intimately involved with the development of ground-based gravitational wave detectors both in the U.S. and Europe such that the valuable technical insight of these efforts represents a practical risk mitigation that is already available to the LISA Project.

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Combining this scientific expertise with the maturity of the LISA mission concept and the considerable flight experience of JPL, GSFC, and ESTEC produces a combination of scientific and engineering talent, flight experience, and laboratory/test facilities that is ideally suited to proceed with the development of the LISA Project.

Although NASA and ESA are implementing the LISA Mission as partners, clear accountability is still an essential feature of a well-planned project. Our goal is to organize the partners in a way that the decision-making processes guarantee the appropriate influence of each partner and still maintain clear individual accountability. This is accomplished through the use of Integrated Technical Teams (ITTs) that are chaired by either NASA or ESA and traditional development offices that are led by either NASA in the U.S. or ESA in Europe. In this arrangement, technical decisions flow through the ITTs and programmatic decisions flow through the development offices. Technical decisions with programmatic impacts are coordinated within the ITTs.

In principle, the partner with the predominant role chairs the ITT. For example, in formulation ESA would chair the ITTs dealing with SMART-2 while NASA would chair the ITT dealing with formulation products. Where the partners have essentially equal involvement, they will negotiate a lead. This would include such formulation activities as the development of requirements, the definition and recommended allocation of mission elements, and the coordination of technology development. The individual ITTs report to the Integrated System Team. The Integrated System Team is a "team of teams" comprised of the senior system engineers from both NASA and ESA that represent all of the mission elements involved in the LISA Mission. It is chaired by NASA and ESA has the deputy-chair. The ITTs are quite active during formulation because they are heavily involved in definition. During this process, any deadlocks within an ITT or inconsistencies between ITTs are quickly reconciled by the Mission System Engineering Office. If the Mission System Engineering Office cannot decide, the matter is referred to the Project Manager for resolution. Once the details of the NASA and ESA work packages have been defined and agreed upon, they are assigned to NASA and ESA "offices" for implementation consistent with an Interface Control Document (ICD) developed and maintained by the ITT responsible for the original definition. During implementation the ITTs monitor the progress of development, evaluate potential changes, and maintain all of the ICDs. At this time, the bulk of the work gradually shifts away from the ITTs and into the developmental offices in order to get the flight hardware and software built and tested.

All of the related functions dealing with a particular area of development are grouped into "offices" and the lead for the day-to-day conduct of each office shall be assigned to either ESA or to a specific NASA field center that will be charged to implement the strategy and decisions previously agreed upon by ESA and NASA. The development offices provide the capability to accommodate the different business practices of NASA and ESA. The entity responsible for the day-to-day conduct of an office has the lead responsibility for implementing the functions of that office based on coordinated inputs from both ESA and NASA. The implementation decisions of these individual leads shall not adversely impact either the jointly agreed upon technical interfaces or their associated budgets and schedules without the prior, specific agreement of both NASA and ESA acting through the appropriate ITT. If the other partner has substantial involvement in a particular office, then the Deputy is represented by the other partner and, for the larger and more active offices, the Deputy is co-located. It is acknowledged that the overall technical complexity of the LISA Mission requires that these offices closely collaborate through the ITTs in the execution of their assigned functions.

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The current LISA Project started with a Management Agreement between GSFC and JPL that was developed in October 2000 and dated November 1, 2000. The ESA input was based on a telephone conversation with ESTEC (John Credland) at about the same time. This Agreement focused primarily on implementation and, similarly, this first version of the Partnership Pathfinder also focuses on implementation although the organization for formulation is briefly presented.

2. PROJECT MANAGEMENT OFFICE

The Project Management Office is managed by both ESA and NASA and NASA has the overall mission management responsibility. Both Agencies are equally responsible for the overall mission success and for their respective financial resources as well as for the technical content, budgetary requirements, and the associated schedules to which the LISA Mission will be implemented. ESA and NASA collaborate to sub-allocate resources and establishing budgets, schedules, and technical requirements to all of the involved project elements.

<u>LISA Project Manager</u>: The LISA Project Manager is appointed by NASA and concurred upon by ESA. He/she is responsible for the overall management of the LISA Project. This responsibility is implemented through two Deputy Project Managers, one responsible for activities funded by ESA or its member states and the other responsible for activities funded by NASA.

ESA Deputy Project Manager: The ESA Deputy Project Manager is specifically responsible for all LISA-related activities funded by ESA or its member states, participates fully in the daily decision-making of the Project Management Office, and represents the LISA Project Manager whenever necessary.

NASA Deputy Project Manager: The NASA Deputy Project Manager is specifically responsible for all LISA-related activities funded by NASA, participates fully in the daily decision-making of the Project Management Office, and represents the LISA Project Manager whenever necessary.

ESA and NASA Project Scientists: The ESA and NASA Project Scientists are jointly responsible for ensuring that the LISA Project meets the scientific requirements established by the Mission Scientists and the LISA International Science Team. The ESA and the NASA Project Scientists work together in reviewing projected LISA technical performance in relation to the base lined scientific requirements, in ensuring that project activities remain consistent with meeting these scientific requirements, and in providing day-to-day scientific guidance to the LISA Project. The Project Scientists are responsible to fully coordinate with their respective Mission Scientists to ensure that project activities remain consistent with meeting science requirements.

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3. MISSION SCIENCE OFFICE AND THE LISA INTERNATIONAL SCIENCE TEAM

The Mission Science Office is jointly managed by the NASA Mission Scientist and the ESA Mission Scientist. Together, they lead the LISA scientific activities and are responsible for establishing and quantifying the LISA scientific requirements with the support of the LISA International Science Team

ESA and NASA Mission Scientists: The ESA and NASA Mission Scientists are responsible for the definition of the LISA scientific requirements. They perform this task in close coordination with the LISA International Science Team, of which they serve as co-chairs. They are responsible for the effective coordination of all of the U.S. and European scientific activities related to the LISA Mission. The Mission Scientists are also responsible for developing the plans for the LISA science analysis, for science outreach, and for managing and funding the LISA International Science Team. They jointly monitor the LISA designs and performance capabilities in relationship to the scientific requirements.

4. MISSION SYSTEM ENGINEERING OFFICE

The Mission System Engineering Office is managed by both ESA and NASA and NASA has the lead for the day-to-day activity. This Office is responsible for ensuring system-level coherence of all design and developmental activities of all of the mission elements. It also leads the Integrated System Team as a "team of teams" with all major project elements represented.

NASA System Engineering Manager: The NASA System Engineering Manager is responsible for the following tasks:

- Chair of the NASA/ESA Integrated System Team;
- Jointly manage the overall design of the LISA Mission consistent with the LISA Mission Requirements Document and the Integration and Verification Plans;
- Jointly manage the definition of the Mission Requirements Document and the Integration and Verification Plans;
- Jointly manage the technical coordination of all individual project elements in order to achieve overall system-level coherence;
- Jointly manage the definition and implementation of interfaces between the various system elements, and
- Jointly manage the coherence of the system budgets within the available resources.

ESA Deputy System Engineering Manager: The ESA Deputy System Engineering Manager is responsible for the following tasks:

- Deputy-Chair of the NASA/ESA Integrated System Team;
- Jointly manage the overall design of the LISA Mission consistent with the LISA Mission Requirements Document and the Integration and Verification Plans;
- Jointly manage the definition of the Mission Requirements Document and the Integration and Verification Plans;
- Jointly manage the technical coordination of all individual project elements in order to achieve overall system-level coherence;

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- Jointly manage the definition and implementation of interfaces between the various system elements, and
- Jointly manage the coherence of the system budgets within the available resources.

Integrated Technical Teams: The LISA Project uses ITTs throughout definition and development in order to effectively accommodate the interests of both partners and to achieve a sustained focus on the mission elements to be developed. This approach also minimizes the impact of the cultural differences between NASA and ESA in that it represents a management approach already familiar to both NASA and ESA. Moreover, when combined with the long-term commitments of senior scientists and engineers, this approach provides the flexibility to smoothly and efficiently transition from formulation into implementation simply by restructuring the membership of the existing ITTs. All of the ITTs coordinate their activities through the Integrated System Team and both the ITTs and the Integrated System Team report to the Mission System Engineering Office.

Integrated System Team: The most intense collaboration between partners occurs in the requirements and design definition. This occurs within the Integrated System Team that is composed of the system engineers and scientists from the involved ITTs and is chaired by the Mission System Engineering Managers from NASA and ESA. To initialize and accelerate the exchange of information between the scientists and the newly assigned engineers, the Mission System Engineering Office sponsors at least three Science / Engineering Workshops per year. Once the designs are defined to the level of specifications, they are then assigned to NASA or ESA offices for development. It is estimated that over 75% of both partner contributions will be spent on traditional contracts managed individually by the partners working to well-defined ICDs. Once development is completed, integration begins in Europe and is completed in the U.S.

The ITTs are most active during formulation because they are heavily involved in definition. Once the details of the NASA and ESA work packages have been defined and agreed upon, they are assigned to NASA and ESA offices for development to an ICD developed and maintained by the ITT responsible for the original definition. During implementation the emphasis shifts to monitoring the progress of development, evaluating potential changes, and maintaining the previously established ICDs. In implementation the bulk of the work gradually shifts away from the ITTs and into the development offices where the flight hardware and software is built and tested.

SE&I Contractor: The LISA Project will utilize a System Engineering & Integration (SE&I) Contractor to support the Integrated System Team, conduct trade-off studies, support the definition and maintenance of the ICDs for the ITTs, and ultimately implement the integration responsibilities assigned to NASA and to support the integration activities assigned to ESA. The SE&I contractor is intimately involved in assessing proposed changes to the ICDs as preparation for the subsequent I&T. The SE&I contractor supports all of the ITTs and represents the "eyes and ears" of the Mission System Engineering Office in identifying sensitive and complex issues that are not likely to be resolved within a given ITT. In such cases, the contractor proactively facilitates a solution to keep the definition process moving smoothly. The contractor will be initially available early in 2005 to support the definition process.

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5. ORGANIZATIONAL DETAILS

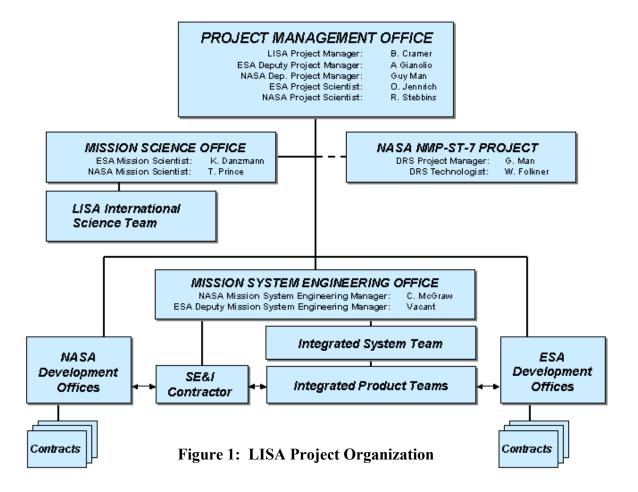


Figure 1 depicts the LISA Project organization. The same basic organizational structure serves technology development, formulation, and implementation. This is accomplished by a gradual restructuring of the ITTs to optimally serve the current needs of the Project. Similarly, the NASA and ESA development offices also evolve with the ITTs with which they are closely associated. This organizational approach also enables a smooth and efficient transition from between phases by systematically evolving the structure of the ITTs to optimally meet the anticipated needs of the next phase.

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A typical organizational structure of the ITTs and their corresponding development offices for formulation is reflected in Figure 2. The first five ITTs presently support technology development and the last two ITTs support formulation and are still in planning.

NASA DEVELOPMENT OFFICES	INTEGRATED TECHNICAL TEAMS (ITT)	ESA DEVELOPMENT OFFICES
Requirements Support	Requirements Definition/ Mission Design	Requirements Support
Mission Element Definition Support	Mission Element Definition / Allocation	Mission Element Definition Support
NASA Technology Development	Technology Coordination	ESA Technology Development
SMART-2 Support	SMART-2 Coordination	SMART-2 Development
GRS Test Package Development (ST-7/DRS)	SMART-2 Test Package Definition	GRS Test Package Development (LTP)
Operations Concept Development	Operations Concept Definition	Operations Concept Development
Formulation Product Development	Formulation Product Definition/ Coordination	Formulation Product Development

Figure 2: LISA Formulation Organization

The allocation of mission elements between NASA and ESA is about 90% complete. Several areas are still in negotiation and will be resolved later in FY03. The current status of this process is shown in Figure 3. These data are updated with each succeeding version of the Partnership Pathfinder to reflect the latest agreements relative to the allocation process. It is necessary to complete this activity as soon as possible so as to enable ESA to commence the process of soliciting European member state contributions for the payload.

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NASA		European	
GSFC	JPL	ESA	ESA MEMBER STATES
Mission Management NASA Business Management Technology Development System Engineering Integrated Modeling Payload Software Observatory Integration Final End-to-End Testing Launch Campaign Launch Vehicle On-Orbit Check-Out	Payload Management Technology Development ST-7 / DRS Science Office Operations Payload Components: Test Mass GRS Electrode Housing Phase Meter Actuator Structure Optical Benches Interferometer Optics Simulators Ground Support Equipment Payload Integration	European Business Mgmt Spacecraft Management Three Spacecraft Three Propulsion Modules Ground Support Equipment Simulators Manage Member State Contributions Technology Development System Engineering SMART-2 Mission	LTP Payload Components: GRS Front-End Electronics Charging System GRS Vacuum & Structure GRS Software Beam Acquisition Sensor Laser Stabilization Bench Lasers Telescopes Ground Support Equipment Initial Payload Integration Science Support

Figure 3: LISA Mission Element Responsibilities

Given the responsibilities for mission elements shown in Figure 3, a typical organizational structure of the ITTs and their corresponding development offices for implementation is depicted in Figure 4. The evolution to this organization begins late in formulation and is in place at the beginning of implementation.

NASA DEVELOPMENT OFFICES	INTEGRATED TECHNICAL TEAMS (ITT)	ESA DEVELOPMENT OFFICES
System Specification Support	System Specification Maintenance	System Specification Support
Standards Development/ Modification	Standards Coordination	Standards Development/ Modification
Flight S/W and GN S/W Development	Flight & Ground S/W Definition Flight S/W and GN S/W ICDs	Flight S/W Development
System/Subsystem Verification	System Verification Definition	System/Subsystem Verification
Payload Component Development	Payload Definition Payload ICDs	Payload Component Development
Spacecraft Support	Spacecraft Definition Spacecraft ICDs	Spacecraft Development (Including Prop Modules)
Overall Payload Integration Observatory Integration/Final Testing	I&T Definition I&T ICDs	Early Payload Integration Early Observatory Integration

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NASA DEVELOPMENT	INTEGRATED TECHNICAL	ESA DEVELOPMENT
OFFICES	TEAMS (ITT)	OFFICES
Launch Vehicle Development	Launch Vehicle Launch Vehicle ICDs	Launch Vehicle Support
Operations Capability Development	Operations Definition Operations ICDs	Operations Support
Science Data Analysis	Science Data Analysis	Science Data Analysis
Capability Development	Definition	Support

Figure 4: LISA Implementation Organization

6. MISSION TECHNOLOGY ACTIVITIES

ESA and NASA will establish and maintain independent but coordinated technology development efforts. NASA will establish and maintain a Technology Office managed by the NASA Mission Technologist. The NASA Mission Technologist is responsible for planning and managing the development of the advanced technologies in the US that represent the technical basis of the LISA Project.

The LISA Project Office at ESTEC, in close coordination with the relevant ESA support organizations, is responsible for planning and coordinating the development of the advanced technologies that represent the technical basis of the LISA Project in Europe. The ESA Architect is responsible for the coordination and management of the LISA Technology activities in close cooperation with the appointed Technical Officers, who are responsible for leading the individual activities. The ESA Architect and the NASA Mission Technologist are responsible for coordinating and reporting the results of their respective technology efforts through the ITT responsible for Technology Coordination.

7. LISA TECHNOLOGY FLIGHT VALIDATION

ESA and NASA agree that the early flight validation of the GRS technologies is important in enhancing our confidence of achieving the LISA Mission. To this end, both Agencies have indicated their intention to fly their respective test packages (LISA Test Package (LTP) and Disturbance Reduction System (DRS)) on a precursor mission. ESA will fly the LTP on the technology validation mission known as SMART-2. ESA has also offered NASA the opportunity to fly their DRS on the same mission and that offer has been accepted in principle. A subsequent, separate Memorandum of Understanding between ESA and NASA will formalize this cooperation.

8. CO-LOCATION OF RESIDENT REPRESENTATIVES

For each of the key positions described above, each Agency shall guarantee the availability of office accommodations and support services to representatives of the other Agency as described herein to discharge their functions and to facilitate the exchange of necessary information and data. The host Agency shall make all arrangements for access to those premises, arrange participation in any related meetings, and provide the associated documentation as required to successfully implement the functions of the key positions described herein.

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Partnership Pathfinder

9.	SIGNATU	RES
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Bryant Cramer, LISA Project Manager Alberto Gianolio, Deputy Project Manager (ESA) Guy Man, Deputy Project Manager (NASA)

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H.5 List of Abbreviations and Acronyms

A&D Architecture & Design

ACS Attitude Control Subsystem

AM Amplitude Modulated

ARCS Austrian Research Centers Seibersdorf

ATD Advanced Technology Development

AU Astronomical Unit (150 Million km)

B1 Build 1

B2 Build 2

B3 Build 3

bps bits per second

C&DH Command and Data Handling

CA Confirmation Assessment

CCD Charge Coupled Device

CCP Contamination Control Plan

CDM Cold Dark Matter

CDR Critical Design Review

CFR Code of Federal Regulations

CHAMP CHAllenging Minisatellite Payload

CPS Central Processing System

DOORS Dynamic Object Oriented Requirements System

DR&D Disposal, Re-Entry and Decommissioning

DRS Disturbance Reduction System

DSN Deep Space Network

E/O Electro-Optic

E&PO Education and Public Outreach

ELV Expendable Launch Vehicle

EM Engineering Model

EMC Electromagnetic Compatibility

EMI Electromagnetic interference

EO-1 Earth Observing-1

EOM Electro-Optic Modulator

ESA European Space Agency

ESTEC European Space Research and Technology Center

FAD Formulation Authorization Document

FEEP	Field Emission Electric Propulsion
FM	Frequency Modulated (in Section D)

FM Frequency Modulated (in Section FM Flight Model (in Section G)

FRR Flight Readiness Review

FSW Flight Software

FTR Final Technical Report

GaAs Gallium Arsenide

GDS Ground Data System

GHz gigahertz

GPS Global Positioning System

GRACE Gravity Recovery and Climate Experiment

GRC NASA Glenn Research Center

GRS Gravitational Reference Sensor

GSFC NASA Goddard Space Flight Center

GW Gravitational Wave

HGA High Gain Antenna

HW Hardware

I&T Integration and Test

ICD Interface Control Drawings / Documents

ICR Initial Confirmation Review

IMS Interferometry Measurement System

ISO International Standards Organization

IT Information Technology

ITAR International Traffic in Arms Regulations

ITT Invitation To Tender

IV&V Independent Verification and Validation

JILA Joint Institute for Laboratory Astrophysics

JPL NASA Jet Propulsion Laboratory

JWST James Webb Space Telescope

LGA Low Gain Antenna

LIGO Laser Interferometer Gravitational Wave Observatory

Li-Ion Lithium Ion

LISA Laser Interferometer Space Antenna

LIST LISA International Science Team

LOA Letter of Agreement

LRD Launch Readiness Date

LRR	Launch Readiness Review
LSR	LISA Science Requirement

LTP LISA Test Package

LZH Laser Zentrum Hannover

MAG Mission Assurance Guidelines

MAR Mission Assurance Requirements

MBH Massive Black Holes

MBW Measurement Bandwidth, 0.1 mHz - 0.1 Hz

MC Mission Concept

MCR Mission Concept Review

MDR Mission Definition Review

MHz megahertz mHz millihertz

MICM Multi-variable Instrument Cost Model

MM Minimal Mission

MOC Mission Operations Center

MOPA Master Oscillator with Power Amplifier

MOR Mission Operations Review

MOU Memorandum of Understanding
MPPT Maximum Power Point Tracking

MRR Mission Readiness Review
 MSC Mission Success Criteria
 M_o Solar Mass (2x10³⁰ kg)

NAC National Agency Check

NAR Non-Advocate Review

NASA National Aeronautics and Space Administration

Nd-YAG Neodymium Yttrium Aluminum Garnet

NEPA National Environmental Policy Act

NFNMS NASA Foreign National Management System

NIST National Institute for Standards and Technology

NMP New Millennium Program

nm Nanometer (10⁻⁹ meters)

NPD NASA Policy Directive

NPG NASA Procedure and Guideline

NPRO Non-planar Ring Oscillator

OA Optical Assembly

OAT	Observatory Architecture Team
ORR	Operational Readiness Review

PDR Preliminary Design Review

P/L Payload

PCA **Program Commitment Agreement**

PDR Preliminary Design Review PER Pre-Environmental Review

PLL Phase-Locked-Loop

РМ **Proof Mass**

picometer (10⁻¹² meters) pm

PSR Pre-Shipment Review

R1 Release 1 R2 Release 2

R3

Release 3 **RCS Reaction Control System**

RF Radio Frequency

RFI Request For Information

rms root mean square

s seconds S/C Spacecraft

SBIR Small Business Innovation Research

SCR System Concept Review

SDMP Science Data Management Plan

SDP Safety Data Package SDS Science Data System SE Systems Engineering

SE&I System Engineering and Integration SEMP System Engineering Management Plan SEU Structure and Evolution of the Universe

SIM Space Interferometry Mission

SM2 SMART-2

SMA Safety and Mission Assurance

SMART-2 Small Missions for Advanced Research in Technology

SMP Science Management Plan SMP Software Management Plan

SNR Signal-to-Noise Ratio

SRR System Requirements Review SSP System Safety Plan ST-7 Space Technology-7 ST&E Science, Technical and Economics STAR Space Triaxial Accelerometer for Research applications STP Software Test Plan STOP Integrated Structural, Thermal, and Optical analysis SuperSTAR Super Space Triaxial Accelerometer for Research applications SW Software T&C Technology & Commercialization TAA **Technical Assistance Agreement TAMA** Japanese ground based gravitational wave detector project TDI Time Delay Interferometry **TES** Tropospheric Emission Spectrometer instrument TRIP Technology Readiness and Implementation Plan **TRL** Technology Readiness Level TRR Technology Readiness Review TV Thermal Vacuum ULE Ultra-Low Expansion USO Ultra-Stable Oscillator UV Ultraviolet W Watts Work Breakdown Structure **WBS**

WD

White Dwarf

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